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A finite difference method for studying thermal deformation in three-dimensional thin films exposed to ultrashort pulsed lasers

Suyang Zhang
Louisiana Tech University

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A FINITE DIFFERENCE METHOD FOR STUDYING THERMAL
DEFORMATION IN 3D THIN FILMS EXPOSED TO
ULTRASHORT-PULSED LASERS

by

Suyang Zhang, B.S., M.S.

A Dissertation Presented in Partial Fulfillment
of the Requirements for the Degree
Doctor of Philosophy

COLLEGE OF ENGINEERING AND SCIENCE
LOUISIANA TECH UNIVERSITY

August 2008
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Exposed to Ultrashort-Pulsed Lasers

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Head of Department
Computational Analysis and Modeling
Department

Advisory Committee

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ABSTRACT

Thermal analysis related to ultrashort-pulsed lasers has been intensely studied in science and engineering communities in recent years, because the pulse duration of ultrashort-pulsed lasers is only the order of sub-picoseconds to femtoseconds, and the lasers have exclusive capabilities in limiting the undesirable spread of the thermal process zone in the heated sample. Studying the thermal deformation induced by ultrashort-pulsed lasers is essential for preventing thermal damage. For the ultrashort-pulsed laser, the thermal damage is different from that caused by the long pulsed lasers and cracks occur after heating.

This dissertation presents a new finite difference method for studying thermal deformation in 3D thin films exposed to ultrashort-pulsed lasers. The method is obtained based on the parabolic two-step model and implicit finite difference schemes on a staggered grid. It accounts for the coupling effect between lattice temperature and strain rate, as well as for the hot electron-blast effect in momentum transfer. In particular, a fourth-order compact scheme is developed for evaluating those stress derivatives in the dynamic equations of motion. The method allows us to avoid non-physical oscillation in the solution.

To test the applicability of the developed numerical scheme, we investigated the temperature rise and thermal deformation in two physical cases: (1) a 3D single-layered
thin film and (2) a 3D double-layered thin film, where the central part of the top surface was irradiated by ultrashort-pulsed lasers. Results show no non-physical oscillations in the solution. Numerical results also show the displacement and stress alterations from negative value to positive value at the center along the $z$-direction, and along $x$ and $y$-directions, indicating that the central part of the thin film expands during heating.
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# TABLE OF CONTENTS

ABSTRACT .......................................................................................................................... iii

LIST OF TABLES .................................................................................................................. viii

LIST OF FIGURES ................................................................................................................. ix

NOMENCLATURE .................................................................................................................. xiii

ACKNOWLEDGEMENTS ........................................................................................................ xvii

CHAPTER 1 INTRODUCTION .................................................................................................. 1

1.1 General Overview ............................................................................................................ 1
1.2 Research Objectives ......................................................................................................... 3
1.3 Organization of the Dissertation ..................................................................................... 4

CHAPTER 2 BACKGROUND AND PREVIOUS WORK .......................................................... 7

2.1 Microscale Heat Transfer ................................................................................................. 7
  2.1.1 Macroscopic Heat Transfer ....................................................................................... 7
  2.1.2 Wave Nature of Microscale Heat Transfer ............................................................... 9
  2.1.3 Two-Step Heat Transport Equations .................................................................... 10

2.2 Previous Work ................................................................................................................. 14
  2.2.1 Two-Dimensional Parabolic Two-Step Model ......................................................... 15
  2.2.2 Two-Dimensional Hyperbolic Two-Step Model ..................................................... 19

CHAPTER 3 3D SINGLE-LAYERED MATHEMATICAL MODEL AND FINITE DIFFERENCE SCHEME ............................................................................................................. 22

3.1 Mathematical Model ........................................................................................................ 22
  3.1.1 Governing Equations .............................................................................................. 22
  3.1.2 Initial and Boundary Conditions ............................................................................ 25

vi
LIST OF TABLES

Table 2.1 Phonon-electron coupling factor $G$, for some noble and transition metals [Qiu 1992] ................................................................. 14

Table 5.1 Thermal properties of gold [Chen 2002, Kaye 1973, Tzou 2002] .............. 54

Table 5.2 Thermophysical properties of gold and chromium [Touloukian 1970a, b Chen 2002, Kaye 1973, Tzou 2002]......................................................... 75
LIST OF FIGURES

Figure 2.1  Configuration of a metal thin film exposed to ultrashort-pulsed lasers......15

Figure 3.1  A 3D thin film with the dimension of 100μm × 100 μm × 0.1μm, irradiated by ultrashort-pulsed lasers .........................................................24

Figure 3.2  A 3D staggered grid and locations of variable .............................................27

Figure 4.1  A 3D double-layered thin film with the dimension of 100μm × 100 μm × 0.1μm, irradiated by ultrashort-pulsed lasers .............................................41

Figure 4.2  A 3D staggered grid for a thin film and locations of variables .......................45

Figure 5.1  A 3D thin film with the dimension of 100μm × 100 μm × 0.1μm, irradiated by ultrashort-pulsed lasers .........................................................53

Figure 5.2  Numerical oscillations appearing near the peaks of the curve [Wang 2007] ............................55

Figure 5.3  Change in electron temperature and displacement (w) at the center of top surface versus time for various grids (20×20×40, 20×20×80, 20×20×100) and laser fluence J of 500 J/m² ........................................56

Figure 5.4  Electron temperature profiles along z at (x_center, y_center) at different times (a) t =0.25 ps, (b) t = 0.5 ps, (c) t = 1 ps, (d) t = 10 ps and (e) t = 20 ps with a mesh of 20×20×80.........................................................57

Figure 5.5  Electron temperature profiles along z at (x_center, y_center) at different times (a) t =0.25 ps, (b) t = 0.5 ps, (c) t = 1 ps, (d) t = 10 ps and (e) t = 20 ps with a mesh of 20×20×80 with same scale.............................58

Figure 5.6  Lattice temperature profiles along z at (x_center, y_center) at different times (a) t =0.25 ps, (b) t = 0.5 ps, (c) t = 1 ps, (d) t = 10 ps and (e) t = 20 ps with a mesh of 20×20×80 and two different laser fluences J of 500 J/m² and J of 2000 J/m .........................................................59
Figure 5.7 Normal stress ($\sigma_z$) profiles along $z$ at ($x_{center}$, $y_{center}$) at different times (a) $t = 1$ ps, (b) $t = 5$ ps, (c) $t = 10$ ps, and (d) $t = 15$ ps with a mesh of $20 \times 20 \times 80$ and two different laser fluences $J$ of $500 \text{ J/m}^2$ and $2000 \text{ J/m}^2$.

Figure 5.8 Contours of electron temperature profiles in the cross section of $y = 50 \text{ \mu m}$ at different times (a) $t = 0.25$ ps, (b) $t = 0.5$ ps, (c) $t = 1$ ps, (d) $t = 10$ ps, and (e) $t = 20$ ps with a mesh of $20 \times 20 \times 80$ and laser fluence $J$ of $500 \text{ J/m}^2$.

Figure 5.9 Contours of lattice temperature profiles in the cross section of $y = 50 \text{ \mu m}$ at different times (a) $t = 0.25$ ps, (b) $t = 0.5$ ps, (c) $t = 1$ ps, (d) $t = 10$ ps, and (e) $t = 20$ ps with a mesh of $20 \times 20 \times 80$ and laser fluence $J$ of $500 \text{ J/m}^2$.

Figure 5.10 Contours of displacement ($w$) profiles in the cross section of $y = 0.5 \text{ mm}$ at different times (a) $t = 5$ ps, (b) $t = 10$ ps, (c) $t = 15$ ps, and (d) $t = 20$ ps with a mesh of $20 \times 20 \times 80$ and laser fluence $J$ of $500 \text{ J/m}^2$.

Figure 5.11 Contours of displacement ($u$) profiles in the cross section of $y = 0.5 \text{ mm}$ at different times (a) $t = 5$ ps, (b) $t = 10$ ps, (c) $t = 15$ ps, and (d) $t = 20$ ps with a mesh of $20 \times 20 \times 80$ and laser fluence $J$ of $500 \text{ J/m}^2$.

Figure 5.12 Contours of displacement ($v$) profiles in the cross section of $x = 0.5 \text{ mm}$ at different times (a) $t = 5$ ps, (b) $t = 10$ ps, (c) $t = 15$ ps, and (d) $t = 20$ ps with a mesh of $20 \times 20 \times 80$ and laser fluence $J$ of $500 \text{ J/m}^2$.

Figure 5.13 Contours of normal stress ($\sigma_x$) profiles in the cross section of $y = 50 \text{ \mu m}$ at different times (a) $t = 5$ ps, (b) $t = 10$ ps, (c) $t = 15$ ps, and (d) $t = 20$ ps with a mesh of $20 \times 20 \times 80$ and laser fluence $J$ of $500 \text{ J/m}^2$.

Figure 5.14 Contours of normal stress ($\sigma_z$) profiles in the cross section of $y = 50 \text{ \mu m}$ at different times (a) $t = 5$ ps, (b) $t = 10$ ps, (c) $t = 15$ ps, and (d) $t = 20$ ps with a mesh of $20 \times 20 \times 80$ and laser fluence $J$ of $500 \text{ J/m}^2$.

Figure 5.15 Contours of normal stress ($\sigma_y$) profiles in the cross section of $x = 50 \text{ \mu m}$ at different times (a) $t = 5$ ps, (b) $t = 10$ ps, (c) $t = 15$ ps, and (d) $t = 20$ ps with a mesh of $20 \times 20 \times 80$ and laser fluence $J$ of $500 \text{ J/m}^2$.
Figure 5.16 Contours of electron temperature profiles at the top surface of
$z = 0 \, \mu m$ at different times (a) $t = 2.25 \, ps$, (b) $t = 4.25 \, ps$,
(c) $t = 6.25 \, ps$, (d) $t = 8.25 \, ps$, (e) $t = 10.25 \, ps$, (f) $t = 12.25 \, ps$,
(g) $t = 14.25 \, ps$, and (h) $t = 16.25 \, ps$ with a mesh of $20 \times 20 \times 80$
and laser fluence $J$ of $500 \, J/m^2$ .......................................................... 70

Figure 5.17 Contours of electron temperature profiles at the top surface of
$z = 0 \, \mu m$ at different times (a) $t = 4 \, ps$, (b) $t = 6 \, ps$, (c) $t = 8 \, ps$,
(d) $t = 10 \, ps$, (e) $t = 12 \, ps$, (f) $t = 14 \, ps$, (g) $t = 16 \, ps$, and
(h) $t = 18 \, ps$ with a mesh of $20 \times 20 \times 80$ and laser fluence
$J$ of $500 \, J/m^2$ .................................................................................. 72

Figure 5.18 Contours of lattice temperature profiles at the top surface of
$z = 0 \, \mu m$ at different times (a) $t = 4 \, ps$, (b) $t = 6 \, ps$, (c) $t = 8 \, ps$,
(d) $t = 10 \, ps$, (e) $t = 12 \, ps$, (f) $t = 14 \, ps$, (g) $t = 16 \, ps$, and
(h) $t = 18 \, ps$ with a mesh of $20 \times 20 \times 80$ and laser fluence
$J$ of $500 \, J/m^2$ .................................................................................. 73

Figure 5.19 A 3D double-layered thin film with the dimension of $100 \, \mu m \times
100 \, \mu m \times 0.1 \, \mu m$, irradiated by ultrashort-pulsed lasers ................. 74

Figure 5.20 (a) Change in electron temperature and (b) displacements at the
center of top surface of thin versus time with a laser fluence
$(J)$ of $500 \, J/m^2$. The $w$ is the displacement at $(x_{\text{center}}, y_{\text{center}}, 0)$
of thin film .......................................................... 77

Figure 5.21 Electron temperature profiles along $z$ at $(x_{\text{center}}, y_{\text{center}})$ at different
times (a) $t = 0.25 \, ps$, (b) $t = 0.5 \, ps$, (c) $t = 10 \, ps$, and (d) $t = 20 \, ps$
with a mesh of $20 \times 20 \times 80$ and three different laser fluences $(J)$ of
$500 \, J/m^2$, $1000 \, J/m^2$ and $2000 \, J/m^2$ .................................................. 78

Figure 5.22 Lattice temperature profiles along $z$ at $(x_{\text{center}}, y_{\text{center}})$ at different
times (a) $t = 0.25 \, ps$, (b) $t = 0.5 \, ps$, (c) $t = 10 \, ps$, and (d) $t = 20 \, ps$
with a mesh of $20 \times 20 \times 80$ and three different laser fluences $(J)$ of
$500 \, J/m^2$, $1000 \, J/m^2$ and $2000 \, J/m^2$ .................................................. 79

Figure 5.23 Displacement $(w)$ profiles along $z$ at $(x_{\text{center}}, y_{\text{center}})$ at different
times (a) $t = 5 \, ps$, (b) $t = 10 \, ps$, (c) $t = 15 \, ps$, and (d) $t = 20 \, ps$
with a mesh of $20 \times 20 \times 80$ and three different laser fluences $(J)$ of
$500 \, J/m^2$, $1000 \, J/m^2$ and $2000 \, J/m^2$ .................................................. 81

Figure 5.24 Normal stress $(\sigma_z)$ profiles along $z$ at $(x_{\text{center}}, y_{\text{center}})$ at different
times (a) $t = 5 \, ps$, (b) $t = 10 \, ps$, (c) $t = 15 \, ps$, and (d) $t = 20 \, ps$
with a mesh of $20 \times 20 \times 80$ and three different laser fluences $(J)$ of
$500 \, J/m^2$, $1000 \, J/m^2$ and $2000 \, J/m^2$ .................................................. 83
Figure 5.25 Contours of electron temperature distributions in the cross section of \( y = 50 \text{ } \mu\text{m} \) at different times (a) \( t = 0.25 \text{ ps} \), (b) \( t = 0.5 \text{ ps} \), (c) \( t = 1 \text{ ps} \), (d) \( t = 10 \text{ ps} \), and (e) \( t = 20 \text{ ps} \) with a mesh of 20x20x80 and a laser fluence \((J)\) of 1000 J/m\(^2\).

Figure 5.26 Contours of lattice temperature distributions in the cross section of \( y = 50 \text{ } \mu\text{m} \) at different times (a) \( t = 0.25 \text{ ps} \), (b) \( t = 0.5 \text{ ps} \), (c) \( t = 1 \text{ ps} \), (d) \( t = 10 \text{ ps} \), and (e) \( t = 20 \text{ ps} \) with a mesh of 20x20x80 and a laser fluence \((J)\) of 1000 J/m\(^2\).

Figure 5.27 Contours of displacement \((w)\) distributions in the cross section of \( y = 50 \text{ } \mu\text{m} \) at different times (a) \( t = 5 \text{ ps} \), (b) \( t = 10 \text{ ps} \), (c) \( t = 15 \text{ ps} \), and (d) \( t = 20 \text{ ps} \) with a mesh of 20x20x80 and a laser fluence \((J)\) of 1000 J/m\(^2\).

Figure 5.28 Contours of displacement \((u)\) distributions in the cross section of \( y = 50 \text{ } \mu\text{m} \) at different times (a) \( t = 5 \text{ ps} \), (b) \( t = 10 \text{ ps} \), (c) \( t = 15 \text{ ps} \), and (d) \( t = 20 \text{ ps} \) with a mesh of 20x20x80 and a laser fluence \((J)\) of 1000 J/m\(^2\).

Figure 5.29 Contours of displacement \((v)\) distributions in the cross section of \( x = 50 \text{ } \mu\text{m} \) at different times (a) \( t = 5 \text{ ps} \), (b) \( t = 10 \text{ ps} \), (c) \( t = 15 \text{ ps} \), and (d) \( t = 20 \text{ ps} \) with a mesh of 20x20x80 and a laser fluence \((J)\) of 1000 J/m\(^2\).
NOMENCLATURE

$C_e$  electron heat capacity, $J/(m^3 K)$

$C_l$  lattice heat capacity, $J/(m^3 K)$

$E$  phonon/electron energy, $J$

$G$  electron-lattice coupling factor, $W/(m^3 K)$

$J$  laser fluence, $J/m^2$

$K$  bulk modulus, $Pa$

$k_e$  thermal conductivity, $W/(mK)$

$L_x$  length of micro thin film in the $x$-direction, $\mu m$

$L_y$  length of micro thin film in the $y$-direction, $\mu m$

$L_z$  length of micro thin film in the $z$-direction, $\mu m$

$m$  index for layer

$N_x$  number of grid points in the $x$-direction

$N_y$  number of grid points in the $y$-direction

$N_z$  number of grid points in the $z$-direction

$Q$  energy absorption, $W/m^2$

$R$  surface reflectivity

$r_s$  spatial profile parameter of laser,
\( S \)  
volumetric heat source, \( W/m^2 \)

\( T \)  
absolute temperature, \( K \)

\( T_e \)  
electron temperature, \( K \)

\( T_l \)  
lattice temperature, \( K \)

\( t_p \)  
laser pulse duration, \( s \)

\( u_{i,j,k}^{n} \)  
numerical solution of \( u(x_i, y_j, z_k, t_n) \)

\( u \)  
displacement in the \( x \)-direction, \( m \)

\( v \)  
displacement in the \( y \)-direction, \( m \)

\( w \)  
displacement in the \( z \)-direction, \( m \)

\( v_1 \)  
velocity component in the \( x \)-direction, \( m/s \)

\( v_2 \)  
velocity component in the \( y \)-direction, \( m/s \)

\( v_3 \)  
velocity component in the \( z \)-direction, \( m/s \)

\( x \)  
Cartesian coordinate

\( y \)  
Cartesian coordinate

\( z \)  
Cartesian coordinate

\( z_s \)  
optical penetration depth, \( m \)

\( \hat{n} \)  
unit outward normal vector on the boundary

**Greek Symbols**

\( \Delta t \)  
time increment, \( s \)

\( \Delta x \)  
rectangular grid size in the \( x \)-direction, \( m \)

\( \Delta y \)  
rectangular grid size in the \( y \)-direction, \( m \)
$\Delta z$ rectangular grid size in the $z$ - direction, $m$

$\zeta$ optical penetration depth, $m$

$\alpha_T$ thermal expansion coefficient

$\Delta_i$ finite difference operator in the $i$ - direction

$\delta_x$ central difference operator

$\delta_y$ central difference operator

$\delta_z$ central difference operator

$\Delta_x$ forward difference operator

$\Delta_{-x}$ backward difference operator

$\Delta_y$ forward difference operator

$\Delta_{-y}$ backward difference operator

$\Delta_z$ forward difference operator

$\Delta_{-z}$ backward difference operator

$\tau_e$ electron relaxation time, $ps$

$\tau_l$ lattice relaxation time, $ps$

$\varepsilon_x$ normal strain in the $x$ - direction

$\varepsilon_y$ normal strain in the $y$ - direction

$\varepsilon_z$ normal strain in the $z$ - direction

$\gamma_{xy}$ shear strain in the $xy$ - plane

$\gamma_{xz}$ shear strain in the $xz$ - plane
shear strain in the \( yz \) - plane

\[ \gamma_{yz} \]

electron-blast coefficient, \( J/(m^3K^2) \)

\[ \lambda \]

Lame’s constant, \( Pa \)

\[ \lambda \]

Lame’s constant, \( Pa \)

\[ \mu \]

density, \( kg/m^3 \)

\[ \rho \]

penetration depth \( nm \)

\[ \delta \]

Stefan-Boltzmann’s constant

\[ \sigma \]

normal stress in the \( x \) - direction

\[ \sigma_x \]

normal stress in the \( y \) - direction

\[ \sigma_y \]

normal stress in the \( z \) - direction

\[ \sigma_z \]

shear stress in the \( xy \) - plane

\[ \sigma_{xy} \]

shear stress in the \( xz \) - plane

\[ \sigma_{xz} \]

shear stress in the \( yz \) - plane

\[ \sigma_{yz} \]

Subscripts and Superscripts

0 \( \) initial value at \( t = 0 \)

e \( \) electron

i \( \) grid index in the \( x \) - direction

j \( \) grid index in the \( y \) - direction

k \( \) grid index in the \( z \) - direction

l \( \) lattice

n \( \) time level
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This dissertation owes great appreciation to many people. I wish to express my sincere gratitude to my advisor, Dr. Weizhong Dai, for his invaluable guidance and generous encouragement. It is my honor to be his student. This dissertation could not have been completed without his guidance and advice. I would like to thank Dr. Don Liu for his warmhearted help. In his PDE class, he showed me the views of mathematics so that I could realize how beautiful mathematics is. Sincere acknowledgement is also extended to Drs. Songming Hou, Andrei Paun, and Mihaela Paun for their kindness of serving as advisory committee members.

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CHAPTER 1

INTRODUCTION

1.1 General Overview

Ultrashort-pulsed lasers have been attracting great interest in the past two decades. Because the pulse duration is only the order of sub-picoseconds to femtoseconds, ultrashort-pulsed lasers have exclusive capabilities in limiting the undesirable spread of the thermal process zone in the heated sample [Tzou 2002]. This unique character has made ultrashort-pulsed lasers an ideal candidate for precise thermal processing of functional nanophase materials [Tzou 2002].

The applications of ultrashort-pulsed lasers include structural monitoring of thin metal films [Mandelis 1992, Opsal 1991], laser micromachining [Knapp 1990] and patterning [Elliot 1989], structural tailoring of microfilms [Grigoropoulos 1994], and laser synthesis and processing in thin-film deposition [Narayan 1991]. Ultrashort-pulsed lasers have been widely applied recently in many disciplines such as biology, medicine, physics, chemistry and optical technology [Liu 2000, Shirk 1998].

The three key factors are relative to the success of high energy ultrashort-pulsed lasers in real applications. First, well characterized pulse width, intensity, and
experimental techniques are required; second, we need to have reliable and accurate microscale heat transfer models; third, it is very important to prevent thermal damage.

The thermal damage caused by ultrashort-pulsed lasers is much different from that of traditional long-pulsed and continuous lasers. For the long-pulsed and continuous lasers, elevated temperatures induce the thermal damage. These high temperatures result from the continuous pumping of photon energy into the thermal process zone. Normally, the long-pulsed laser intensity drives the heat spot to the melting temperature. For the ultrashort-pulsed lasers, the thermal damage occurs after the heating pulse duration. The thin material layers are shattered without a clear signature of thermal damage by excessive temperature. The results show that a new driving force, rather than the melting temperature, brings such ultra fast damage, probably within only a few picoseconds [Tzou 2002].

The studying of ultrashort-pulsed lasers has been a popular research topic in science and engineering communities. Up to date, scientists and researchers have developed a number of models, which focus on heat transfer in the context of ultrashort-pulsed lasers. However, only a few mathematical models for studying thermal deformation induced by ultrashort-pulsed lasers have been developed [Chen 2002a, Chen 2003, Lee 2008]. Tzou and his colleagues presented a one-dimensional numerical model in a double-layered thin film [Tzou 2002], which was solved using a differential-difference approach. Chen and his colleagues considered a two-dimensional axisymmetric cylindrical thin film and proposed an explicit finite difference method by adding an artificial viscosity term to eliminate numerical oscillations [Chen 2002b]. Lee
and Tsai [Lee 2008] studied the effect of interfacial contact conductance on thermal-elastic response in a 1D two-layered material heated using Laplace transform method.

Recently, my advisor, Dr. Dai, and his colleagues developed a new numerical method for studying thermal deformation in two-dimension thin films exposed to ultrashort-pulsed lasers [Wang 2006a, Wang 2006b, Wang 2008]. This method was developed based on the parabolic two-step heat transport equations and implicit finite difference schemes on a staggered grid. It accounts for the coupling effect between lattice temperature and strain rate, as well as for the hot electron-blast effect in momentum transfer. Numerical results show that there are no numerical oscillations in the solution. Unfortunately, when applying this method to a 3D thin film case, we found that the non-physical oscillations appeared again in the normal stress in the thickness direction. This is probably because a relatively coarse grid was used in the computation. However, a finer mesh increased dramatically the computational cost.

The motivation of my dissertation is to improve our previous method and extend our research to 3D single-layered and double-layered thin film cases. Layered metal thin films are considered in the dissertation because they are widely used in engineering applications due to the fact that a single metal layer often cannot satisfy all mechanical, thermal and electronic requirements.

1.2 Research Objectives

The objective of my dissertation is to develop a finite difference method for studying thermal deformation in 3D thin films exposed to ultrashort-pulsed lasers. This method is developed based on the parabolic two-step heat transport equations and the
dynamic equations of motion. To achieve this objective, the following steps must be followed.

1. Consider a 3D thin film in Cartesian coordinates, and propose the dynamic equations of motion and energy equations as the governing equations for describing thermal deformation in the thin films induced by ultrashort-pulsed lasers.

2. Introduce three velocity components into the model, and rewrite the dynamic equations of motion to avoid numerical oscillations.

3. Design a staggered grid to avoid further numerical oscillations, and develop a finite difference scheme based on the staggered grid.

4. Develop a fourth-order compact finite difference scheme for solving stress derivatives in the dynamic equations of motion in order to eliminate the third order derivatives in the truncation error.

5. Employ the developed finite difference method to obtain electron and lattice temperatures, normal and shear stresses and strains, and displacements and velocities in a 3D single-layered metal thin film and a 3D double-layered metal thin film, respectively.

6. Test the method, and analyze the solution.

1.3 Organization of the Dissertation

Chapter 1 gives a general overview of the dissertation topic, an introduction of ultrashort-pulsed lasers, and some reviews of previous researches in relative areas. The objective of this dissertation is also proposed.
Chapter 2 reviews some background for studying thermal deformation in metal thin films exposed to ultrashort-pulsed lasers. The process of micro scale heat transfer of phonon-electron interaction model and the parabolic two-step model for micro thin films, as well as a review of previous works, are included later.

Chapter 3 proposes the mathematical model for a 3D single-layered metal thin film in Cartesian coordinates, where the film is exposed to ultrashort-pulsed lasers. Dynamic equations of motion and parabolic two-step heat conduction equations are considered to be the governing equations for describing thermal deformation in the 3D thin films induced by ultrashort-pulsed lasers. To develop a numerical method for solving the mathematical model, we first introduce three velocity components into the model, rewrite the dynamic equations of motion, and develop a staggered finite difference scheme for solving the governing equations and a numerical algorithm for obtaining temperatures, displacements, stresses, and strains.

Chapter 4 extends the previous mathematical model and numerical method to a 3D double-layered metal thin film case. Interfacial conditions are introduced between two layers. A numerical algorithm is then developed for obtaining the temperatures, displacements, stresses, and strains.

Chapter 5 gives the numerical results obtained based on the developed numerical methods. Two cases are considered in this chapter; one is a 3D single-layered thin film, and the other is a 3D double-layered thin film exposed to an ultrashort-pulsed laser with perfectly contacted interface. Three various mesh sizes were chosen in order to test the convergence of the scheme. Also, electron temperatures, lattice temperatures, displacements, stresses, and strains are calculated and analyzed in this chapter. Finally,
Chapter 6 gives the conclusions of my dissertation research and suggests future research work.
CHAPTER 2

BACKGROUND AND PREVIOUS WORK

2.1 Microscale Heat Transfer

2.1.1 Macroscopic Heat Transfer

This section is based on Chapter 2 in Wang’s dissertation [Wang 2007]. Heat is defined as energy transfer due to temperature gradients or differences in thermodynamics. At the microscale heat transfer, this view point still works. There are only three heat transfer modes, which are generally recognized as convection, conduction and radiation. In our research, heat transfer on a 3D metal thin film exposed to an ultrashort- pulsed laser occurred by radiation, whereas heat transfer across the metal thin film occurred by conduction.

An important distinction between conduction and radiation is that conduction is by molecules, since these energy carriers have a shorter mean free path, whereas radiation is by photons, which are energy carriers that have a long mean free path. These modes of transfer occur on a molecular scale.

At the microscale heat transfer, the process of heat transfer by phonon-electron interaction in metallic films and by phonon scattering in dielectric films, conductors, and
semiconductors is considered. Our research will focus our discussion on heat transfer by phonon-electron interaction in metallic films.

In classical theories of heat transfer, Fourier's law is the main phenomenological law subjected to heat conduction [Herwig 2000]. Fourier's law is a constitutive equation that depicts the way in which cause varies with effect.

Fourier's law of heat conduction,

$$\bar{q} = -k \nabla T,$$  \hspace{1cm} (2.1)

where $k$ is the thermal conductivity of the material that dictates that the heat flux vector ($\bar{q}$) and the temperature gradient ($\nabla T$) across a material volume must occur at the same time.

The energy equation derived from the first law of thermodynamics is

$$-\nabla \cdot \bar{q} = C_p \frac{\partial T}{\partial t} - Q,$$ \hspace{1cm} (2.2)

where $C_p$ is the volumetric heat capacity and $Q$ is the heat source. Substituting Equation (2.1) into Equation (2.2), we can obtain the traditional heat diffusion equation:

$$C_p \frac{\partial T}{\partial t} = \nabla \cdot (k \nabla T) + Q.$$ \hspace{1cm} (2.3)

Equation (2.3) is often referred to as a parabolic equation, and as a result, any temperature disturbance will propagate at an infinite speed.

The classical theories established are not expected to be informative enough at the microscale because they describe macroscopic behavior aggregated over many grains. Because Fourier's law does not predict finite wave speeds, the law does not accurately approximate the heat transfer in certain cases [Tang 1996, Xu 2003]. The assumption of instantaneous energy transmission fails during a short duration of initial transient, or
when the thermal propagation speed is not high, such as in low temperatures [Barron 1985]. Fourier's law breaks down at temperatures near absolute zero. They break down further when the pulsed duration becomes extremely small, even on the order of picoseconds or femtoseconds. A typical case occurs in the ultrashort-pulsed laser heating in the thermal processing of materials [Kaba 2004, Kaba 2005]. In this instance, the quasi-equilibrium assumption established in Fourier's law does not get along with other macroscopic behaviors. Specific to microscale heat transfer, Fourier's law does not accurately predict the transient temperature during microscale ($<10^{-12}$ s) laser heating of thin metal films ($<10^{-16}$ m) [Barron 2005, Barron 2006, Qiu 1993].

2.1.2 Wave Nature of Microscale Heat Transfer

Heat transfer is defined as energy transfer due to temperature differences. The principal mode of conduction heat transfer is that of vibrational energy transfer from one atom to its neighbors. Atoms in solids are constantly at very high frequencies with relatively small amplitudes. A single quantum of vibrational energy is called a phonon. These vibrations are coordinated in such a way that traveling lattice waves are produced, which propagates through the lattice at the speed of sound [Tzou 1996]. In this processing, heat transfer requires an energy carrier. In the metal thin films, electrons and phonons play the roles as the main energy carriers.

It should be pointed out that the free electron mechanism of heat transfer is much more efficient than the phonon mechanism in metal. The main reason is that phonons are more easily scattered than electrons [Touloukian 1970a, 1970b]. Another reason we should consider those electrons is that they have higher velocities than phonons. From Tzou and his colleagues' research, we know that the short mean free path of an electron
in a bulk material is normally small, say, on the order of 10 to 30 nm at room temperature, where the electron lattice is dominant. In their research, Tzou introduced that when the film thickness is on the order of the mean free path, boundary scattering becomes important. Thin films are manufactured using a number of methods and a wide variety of conditions [Tzou 1996]. When a metal thin film is heated by ultrashort-pulsed lasers, the micro structural interaction effect, such as phonon-electron or phonon scattering, makes the electron system hot, so then scattering can become significant. Thus, the microscopic energy carriers and the full range of possible scattering mechanisms should be considered in our microscale heat transfer model [Wang 2007].

2.1.3 Two-Step Heat Transport Equations

For considering the phonon-electron interaction, the conventional model should be revised to fit the heat transfer theories in microscale. For those electrons with much smaller heat capacity than metal lattice, the heating system involves excitation of the electrons and heating of the metal lattice through phonon-electron interaction in short times [Tzou 1996]. The phonon-electron interaction model was expected to exactly describe this two primary phases for energy transport. The first phase describes the deposition of energy on electrons and the second describes the transfer of the energy from the electrons to the lattice.

The first version of the phonon-electron interaction model, referred to as the two-step model, was proposed by Kaganov et al. [Kaganov 1957] and Anisimov et al. [Anisimov 1974], but it seems that they did not hit the target at that time. Tien and Qiu set up the well-known hyperbolic two-step model on a quantum mechanical and statistical basis almost 20 years later [Qiu 1993]. By setting the relaxation time of the
electron gas calculated at the Fermi surface to be zero, this hyperbolic two-step model can perfectly reduced to the parabolic two-step model, which was originally proposed by Kaganov et al. [Kaganov 1957] and Anisimov et al. [Anisimov 1974].

The parabolic two-step model can be written as:

\[ C_e(T_e) \frac{\partial T_e}{\partial t} = \nabla \cdot (kVT_e) - G(T_e - T_i), \]  
(2.4)

\[ C_l(T_i) \frac{\partial T_i}{\partial t} = G(T_e - T_i), \]  
(2.5)

Here, \( C_e(T_e) \) is the electron heat capacity, \( k \) is the thermal conductivity, \( G \) is the electron-lattice coupling factor, \( C_l(T_i) \) is the lattice heat capacity, and subscripts \( e \) and \( l \) represent the electron and metal lattice, respectively.

Equation (2.4) represents the first step, which describes the deposition of energy heating on electrons, and Equation (2.5) represents the second step, which involves the transfer of the energy from electrons to the lattice. Here, the effect of heat conduction through the metal lattice is not considered at this time.

Tzou and his colleague's research show that for an electron gas temperature lower than the Fermi temperature, of the order of \( 10^4 \) K, the electron heat capacity \( (C_e) \) is proportional to the electron temperature [Tzou 1996]. According to the above conclusion, that equation is non-linear for solving.

The electron heat capacity \( C_e \) can be obtained from Barron's research [Barron 1985]:

\[ C_e = \gamma_e T_e, \]  
(2.6)
where \( \gamma_e \) is known as the electron specific heat coefficient, and we can obtain it from experiments.

The phonon-electron coupling factor describes the energy exchange between phonons and electrons [Kaganov 1957]:

\[
G = \frac{\pi^2}{6} \frac{m_e n_e \nu_s^2}{\tau_e T_e} \quad \text{for} \quad T_e \gg T_l, \tag{2.7}
\]

where \( m_e \) is the electron mass, \( n_e \) is the number density of electrons per unit volume, and \( \nu_s \) is the speed of sound. It is obtained below as

\[
\nu_s = \frac{\sigma}{2 \pi h} (6\pi^2 n_a)^{-1} T_D, \tag{2.8}
\]

where the quantity \( h \) is Planck’s constant, \( k \) is the Boltzmann constant, \( n_a \) is the atomic number density per unit volume, and \( T_D \) represents the Debye temperature. The electron temperature (\( T_e \)) is much higher than the lattice temperature (\( T_l \)) in the early time response. Then the lattice temperature (\( T_l \)) increases, and as a result, the electron temperature (\( T_e \)) decreases due to the electron-lattice effect. The condition \( T_e \gg T_l \) in Equation (2.6) for the applicability of \( G \) is thus valid in the fast-transient process of electron-phonon dynamics. Within the limits of Wiedemann-Frenz’s law, which states that for metals at moderate temperatures (\( T_l > 0.48 T_D \)), the ratio of the thermal conductivity to the electrical conductivity is proportional to the temperature, and the constant of proportionality is independent of particular metal, the electron thermal conductivity can be expressed as [Kaganov 1957]

\[
k_e = \frac{\pi^2 n_a k^2 \tau_e T_e}{3 m_e}, \tag{2.9}
\]
or just set \( m_e \) simply below

\[
m_e = \frac{\pi^2 n_e k^2 T_e}{3 k_e}.
\]  

(2.10)

We substitute Equation (2.10) into Equation (2.7) for the electron mass yields, and then we can calculate \( G \) as

\[
G = \frac{\pi^4 (n_e v k_e)^2}{18 \sigma}.
\]  

(2.11)

This electron-lattice coupling factor is decided by the thermal conductivity \( (k) \) and the number density \( (n_e) \) of the electron gas. From Tzou's research, the electron-lattice coupling factor does not show a strong dependence upon temperature and is not affected by relaxation time [Tzou 1996].

From Equation (2.10) we can see that the number density \( (n_e) \) of the electron gas is a key quantity for estimating the value of \( G \). Qiu and his colleagues’ assumed one free electron per atom for noble metals and employed the \( s \)-band approximation for the valence electrons in transition metals [Qiu 1992]. Thus, the value for the number density of the electron gas is chosen as a fraction of the valence electrons. The phonon-electron coupling factor can be calculated. These experimentally obtained values are listed in Table 2.1 for comparison. In this dissertation, Equation (2.4) is used for calculating the unknown electron-gas temperature \( (T_e) \), and Equation (2.5) is used for calculating the unknown metal-lattice temperature \( (T_i) \).
Table 2.1 Phonon-electron coupling factor $G$, for some noble and transition metals [Qiu 1992]

<table>
<thead>
<tr>
<th>Metal</th>
<th>Calculated, $\times 10^{16} \text{ W/m}^3\text{K}$</th>
<th>Measured, $\times 10^{16} \text{ W/m}^3\text{K}$</th>
</tr>
</thead>
<tbody>
<tr>
<td>Cu</td>
<td>14</td>
<td>$4.8 \pm 0.7$ [Brorson 1990]</td>
</tr>
<tr>
<td></td>
<td></td>
<td>10 [Elsayed-Ali 1987]</td>
</tr>
<tr>
<td>Ag</td>
<td>3.1</td>
<td>2.8 [Groeneveld 1990]</td>
</tr>
<tr>
<td>Au</td>
<td>2.6</td>
<td>$2.8 \pm 0.5$ [Brorson 1990]</td>
</tr>
<tr>
<td>Cr</td>
<td>$45 \ (n_e/n_\alpha = 0.5)$</td>
<td>$42 \pm 5$ [Brorson 1990]</td>
</tr>
<tr>
<td>W</td>
<td>$27 \ (n_e/n_\alpha = 1.0)$</td>
<td>$26 \pm 3$ [Brorson 1990]</td>
</tr>
<tr>
<td>V</td>
<td>$648 \ (n_e/n_\alpha = 2.0)$</td>
<td>$523 \pm 37$ [Brorson 1990]</td>
</tr>
<tr>
<td>Nb</td>
<td>$138 \ (n_e/n_\alpha = 2.0)$</td>
<td>$387 \pm 36$ [Brorson 1990]</td>
</tr>
<tr>
<td>Pb</td>
<td>62</td>
<td>$12.4 \pm 1.4$ [Brorson 1990]</td>
</tr>
<tr>
<td>Ti</td>
<td>$202 \ (n_e/n_\alpha = 1.0)$</td>
<td>$185 \pm 16$ [Brorson 1990]</td>
</tr>
</tbody>
</table>

2.2 Previous Work

two-dimensional axisymmetric cylindrical thin film and proposed an explicit finite difference method by adding an artificial viscosity term to eliminate numerical oscillations [Chen 2002, Chen 2003]. Lee and Tsai studied the effect of interfacial contact conductance on thermal-elastic response in a 1D two-layered material heated using Laplace transform method [Lee 2008].

2.2.1 Two-Dimensional Parabolic Two-Step Model

In 2006, Dai and his colleagues [Wang 2006a] developed a finite difference method for studying thermal deformation in a two-dimension single-layered thin film exposed to ultrashort-pulsed lasers based on the parabolic two-step heat transport equations, as shown in Figure 2.1.

![Figure 2.1 Configuration of a metal thin film exposed to ultrashort-pulsed lasers.](image)

The governing equations for studying thermal deformation can be expressed as:

\[
\rho \frac{\partial^2 u}{\partial t^2} = \frac{\partial \sigma_x}{\partial x} + \frac{\partial \sigma_y}{\partial y} + 2\Lambda T_e \frac{\partial T_e}{\partial x},
\]
(2.12)

\[
\rho \frac{\partial^2 v}{\partial t^2} = \frac{\partial \sigma_x}{\partial x} + \frac{\partial \sigma_y}{\partial y} + 2\Lambda T_e \frac{\partial T_e}{\partial y},
\]
(2.13)

where

\[
\sigma_x = \lambda (\varepsilon_x + \varepsilon_y) + 2\mu \varepsilon_x - (3\lambda + 2\mu)\alpha_\tau (T_i - T_0),
\]
(2.14)

\[
\sigma_y = \lambda (\varepsilon_x + \varepsilon_y) + 2\mu \varepsilon_y - (3\lambda + 2\mu)\alpha_\tau (T_i - T_0),
\]
(2.15)

\[
\sigma_{xy} = \mu \varepsilon_{xy},
\]
(2.16)

\[
\varepsilon_x = \frac{\partial u}{\partial x}, \varepsilon_y = \frac{\partial u}{\partial y}, \varepsilon_{xy} = \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x},
\]
(2.17)

\[
\lambda = K - \frac{2}{3} \mu.
\]
(2.18)

(2) Energy Equations [Chen 2002a, Qiu 1992, Tzou 2002,]

\[
C_e(T_e) \frac{\partial T_e}{\partial t} = \frac{\partial}{\partial x} \left[ k_e(T_e, T_i) \frac{\partial T_e}{\partial x} \right] + \frac{\partial}{\partial y} \left[ k_e(T_e, T_i) \frac{\partial T_e}{\partial y} \right] - G(T_e - T_i) + Q,
\]
(2.19)

\[
C_i \frac{\partial T_i}{\partial t} = G(T_e - T_i) - (3\lambda + 2\mu)\alpha_\tau T_0 \frac{\partial}{\partial t} (\varepsilon_x + \varepsilon_y),
\]
(2.20)

where the heat source is given by

\[
Q = 0.94 J \frac{1 - R}{t_p x_s} \exp \left[ -\frac{x}{x_s} - \left( \frac{y}{y_s} \right)^2 - 2.77 \left( \frac{t - 2t_p}{t_p} \right)^2 \right].
\]
(2.21)

Here, \( C_e(T_e) = C_{e0} \left( \frac{T_e}{T_0} \right) \), and \( k_e(T_e, T_i) = k_0 \left( \frac{T_e}{T_i} \right) \). Equation (2.19) and Equation (2.20)

are often referred to as parabolic two-step heat transport equations.
The boundary conditions are assumed to be

\[ \sigma_x = 0, \sigma_{xy} = 0, \text{ at } x = 0, L_x, \]  
\[ (2.22) \]

\[ \sigma_y = 0, \sigma_{xy} = 0, \text{ at } y = 0, L_y, \]  
\[ (2.23) \]

\[ \frac{\partial T_e}{\partial n} = 0, \frac{\partial T_i}{\partial n} = 0. \]  
\[ (2.24) \]

The initial conditions are assumed to be

\[ T_e = T_i = T_0, u = v = 0, u_t = v_t = 0, \text{ at } t = 0. \]  
\[ (2.25) \]

Layered metal thin films were also considered [Wang 2006b] because they are widely used in engineering applications because a single metal layer often cannot satisfy all mechanical, thermal, and electronic requirements [Lor 2000].

The governing equations are written as follows:

(1) Dynamic Equations of Motion [Brorson 1987, Chen 2002a, Tzou 2002]

\[ \rho^{(m)} \frac{\partial^2 u^{(m)}}{\partial t^2} = \frac{\partial \sigma_x^{(m)}}{\partial x} + \frac{\partial \sigma_{xy}^{(m)}}{\partial y} + 2\Lambda^{(m)} \tau^{(m)} \frac{\partial T_e^{(m)}}{\partial x}, \]  
\[ (2.26) \]

\[ \rho^{(m)} \frac{\partial^2 v^{(m)}}{\partial t^2} = \frac{\partial \sigma_y^{(m)}}{\partial x} + \frac{\partial \sigma_{xy}^{(m)}}{\partial y} + 2\Lambda^{(m)} \tau^{(m)} \frac{\partial T_e^{(m)}}{\partial y}, \]  
\[ (2.27) \]

where

\[ \sigma_x^{(m)} = \lambda^{(m)} (\varepsilon_x^{(m)} + \varepsilon_y^{(m)}) + 2\mu^{(m)} \varepsilon_x^{(m)} - (3\lambda^{(m)} + 2\mu^{(m)}) \alpha_x^{(m)} (T_e^{(m)} - T_0), \]  
\[ (2.28) \]

\[ \sigma_y^{(m)} = \lambda^{(m)} (\varepsilon_x^{(m)} + \varepsilon_y^{(m)}) + 2\mu^{(m)} \varepsilon_y^{(m)} - (3\lambda^{(m)} + 2\mu^{(m)}) \alpha_y^{(m)} (T_e^{(m)} - T_0), \]  
\[ (2.29) \]

\[ \varepsilon_x^{(m)} = \frac{\partial u^{(m)}}{\partial x}, \quad \varepsilon_y^{(m)} = \frac{\partial v^{(m)}}{\partial y}, \quad \varepsilon_{xy}^{(m)} = \frac{\partial u^{(m)}}{\partial y} + \frac{\partial v^{(m)}}{\partial x}. \]  
\[ (2.30) \]

Here, \( m = 1, 2 \) denotes layer 1 and 2, respectively.

(2) Energy Equations [Chen 2002a, Qiu 1992, Tzou 2002]
\[(C_e(T_e))^{(m)} \frac{\partial T^{(m)}_e}{\partial t} = \frac{\partial}{\partial x}\left[(k_e(T_e,T_i))^{(m)} \frac{\partial T^{(m)}_e}{\partial x}\right] + \frac{\partial}{\partial y}\left[(k_e(T_e,T_i))^{(m)} \frac{\partial T^{(m)}_e}{\partial y}\right] - G^{(m)} (T^{(m)}_e - T^{(m)}_i) + Q, \quad (2.31)\]

\[C_l^{(m)} \frac{\partial T^{(m)}_l}{\partial t} = G^{(m)} (T^{(m)}_e - T^{(m)}_l) - (3 \lambda^{(m)} + 2 \mu^{(m)}) \alpha^{(m)} T_0 \frac{\partial}{\partial t} (\varepsilon^{(m)}_x + \varepsilon^{(m)}_y), \quad (2.32)\]

where the heat source is given by

\[Q = 0.94J \frac{1 - R}{t_p x_s} \exp\left[-\frac{x}{x_s} - \left(\frac{y}{y_s}\right)^2 - 2.77 \left(\frac{t - 2t_p}{t_p}\right)^2\right]. \quad (2.33)\]

The boundary conditions are

\[\sigma_x^{(1)} = 0, \sigma_{xy}^{(1)} = 0, \text{ at } x = 0, \text{ and } \sigma_x^{(2)} = 0, \sigma_{xy}^{(2)} = 0, \text{ at } x = L_x, \quad (2.34)\]

\[\sigma_y^{(1)} = 0, \sigma_{xy}^{(1)} = 0, \text{ at } y = 0, \text{ and } \sigma_y^{(2)} = 0, \sigma_{xy}^{(2)} = 0, \text{ at } y = L_y, \quad (2.35)\]

\[\frac{\partial T^{(m)}_e}{\partial n} = 0, \frac{\partial T^{(m)}_l}{\partial n} = 0. \quad (2.36)\]

The initial conditions are assumed to be

\[T^{(m)}_e = T^{(m)}_l = T_0, \quad u^{(m)} = v^{(m)} = 0, \quad u^{(m)} = v^{(m)} = 0, \text{ at } t = 0. \quad (2.37)\]

The interfacial conditions are assumed to be, at \(x = L_x/2\),

\[u^{(1)} = u^{(2)}, v^{(1)} = v^{(2)}, \quad (2.38)\]

\[\sigma_x^{(1)} = \sigma_x^{(2)}, \sigma_{xy}^{(1)} = \sigma_{xy}^{(2)}, \quad (2.39)\]

\[T^{(1)}_e = T^{(2)}_e, \quad k^{(1)}_e \frac{\partial T^{(1)}_e}{\partial x} = k^{(2)}_e \frac{\partial T^{(2)}_e}{\partial x}. \quad (2.40)\]
2.2.2 Two-Dimensional Hyperbolic Two-Step Model

Recently, my advisor, Dr. Dai, and his colleagues have further developed a numerical method based on hyperbolic two-step model for studying thermal deformation in 2D single-layered thin films exposed to ultrashort-pulsed lasers [Niu 2008a, 2008b]. The governing equations for studying thermal deformation in the thin film can be expressed as:

(1) Dynamic Equations of Motion [Brorson 1987, Chen 2002a, Tzou 2002]

\[
\rho \frac{\partial^2 u}{\partial t^2} = \frac{\partial \sigma_x}{\partial x} + \frac{\partial \sigma_{xy}}{\partial y} + 2\Lambda T_e \frac{\partial T_e}{\partial x}, \tag{2.41}
\]

\[
\rho \frac{\partial^2 v}{\partial t^2} = \frac{\partial \sigma_y}{\partial x} + \frac{\partial \sigma_{xy}}{\partial y} + 2\Lambda T_e \frac{\partial T_e}{\partial y}, \tag{2.42}
\]

where

\[
\sigma_x = \lambda (\varepsilon_x + \varepsilon_y) + 2\mu \varepsilon_x - (3\lambda + 2\mu)\alpha_x (T_e - T_0), \tag{2.43}
\]

\[
\sigma_y = \lambda (\varepsilon_x + \varepsilon_y) + 2\mu \varepsilon_y - (3\lambda + 2\mu)\alpha_y (T_e - T_0), \tag{2.44}
\]

\[
\sigma_{xy} = \mu \varepsilon_{xy}, \tag{2.45}
\]

\[
\varepsilon_x = \frac{\partial u}{\partial x}, \varepsilon_y = \frac{\partial u}{\partial y}, \varepsilon_{xy} = \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x}, \tag{2.46}
\]

\[
\lambda = \frac{K}{3} - \frac{2}{3} \mu. \tag{2.47}
\]

(2) Energy Equations [Chen 2002a, Qiu 1992, Tzou 2002]

\[
C_e \frac{\partial T_e}{\partial t} = -\frac{\partial q_e^x}{\partial x} - \frac{\partial q_e^y}{\partial y} - G(T_e - T_i) + Q, \tag{2.48}
\]

\[
\tau_e \frac{\partial q_e^x}{\partial t} + q_e^x = -k_e \frac{\partial T_e}{\partial x}, \tag{2.49}
\]
\[ \tau_e \frac{\partial q_e^y}{\partial t} + q_e^y = -k_e \frac{\partial T_e}{\partial y}, \quad (2.50) \]

\[ C_i \frac{\partial T_i}{\partial t} = -\frac{\partial q_i^x}{\partial x} - \frac{\partial q_i^y}{\partial y} + G(T_e - T_i) - (3\lambda + 2\mu)\alpha_i T_0 \frac{\partial}{\partial t}(\varepsilon_x + \varepsilon_y), \quad (2.51) \]

\[ \tau_i \frac{\partial q_i^x}{\partial t} + q_i^x = -k_i \frac{\partial T_i}{\partial x}, \quad (2.52) \]

\[ \tau_i \frac{\partial q_i^y}{\partial t} + q_i^y = -k_i \frac{\partial T_i}{\partial y}. \quad (2.53) \]

These energy equations are referred to as hyperbolic two-step heat transport equations. The boundary conditions are assumed to be

\[ \sigma_x = 0, \sigma_y = 0, \text{ at } x = 0, L_x, \quad (2.54) \]

\[ \sigma_y = 0, \sigma_{xy} = 0, \text{ at } y = 0, L_y, \quad (2.55) \]

\[ \frac{\partial T_e}{\partial n} = 0, \frac{\partial T_i}{\partial n} = 0. \quad (2.56) \]

The initial conditions are assumed to be

\[ T_e = T_i = T_0, u = v = 0, u_r = v_r = 0, \text{ at } t = 0. \quad (2.57) \]

If \( \tau_e, \tau_i \) and \( k_e \) are zeros, the hyperbolic two-step model can be reduced to the parabolic two-step model.

Numerical results show that the method in Wang [Wang 2006a, 2006b] and Niu [Niu 2008a, 2008b] allow us to avoid non-physical oscillations in the solution. Unfortunately, when applying this method to a 3D thin film case, we found that the non-physical oscillations appeared again in the normal stress in the thickness direction. This result probably occurred because we used a relatively coarse grid in the computation. However a finer grid increased dramatically the computational cost. Hence, the
motivation of my dissertation research is to improve their previous methods and extend the research to 3D single-layered and double-layered thin film cases.
CHAPTER 3

3D SINGLE-LAYERED MATHEMATICAL MODEL
AND FINITE DIFFERENCE SCHEME

In this chapter, we consider a 3D thin film exposed to ultrashort-pulsed lasers. We will purpose a mathematical model for studying thermal deformation in the 3D thin film and then develop a numerical method for studying the temperatures, displacements, stresses, and strains based on the mathematical model.

3.1 Mathematical Model

3.1.1 Governing Equations

We consider a 3D single-layered thin film in Cartesian coordinates, which is exposed to ultrashort-pulsed lasers, as shown in Figure 3.1. The governing equations for studying thermal deformation in the thin film induced by ultrashort-pulsed lasers can be expressed as follows:


\[ \rho \frac{\partial^2 u}{\partial t^2} = \frac{\partial \sigma_x}{\partial x} + \frac{\partial \sigma_{xy}}{\partial y} + \frac{\partial \sigma_{xz}}{\partial z} + 2\lambda T_z \frac{\partial T_z}{\partial x}, \]  

(3.1)
\[
\frac{\partial^2 v}{\partial t^2} = \frac{\partial \sigma_{xy}}{\partial x} + \frac{\partial \sigma_y}{\partial y} + \frac{\partial \sigma_{yz}}{\partial z} + 2\Lambda T_e \frac{\partial T_e}{\partial y},
\]
(3.2)

\[
\frac{\partial^2 w}{\partial t^2} = \frac{\partial \sigma_{xz}}{\partial x} + \frac{\partial \sigma_{yx}}{\partial y} + \frac{\partial \sigma_{zz}}{\partial z} + 2\Lambda T_e \frac{\partial T_e}{\partial z},
\]
(3.3)

\[
\sigma_x = \lambda (\varepsilon_x + \varepsilon_y + \varepsilon_z) + 2\mu \varepsilon_x - (3\lambda + 2\mu)\alpha_x(T_e - T_0),
\]
(3.4a)

\[
\sigma_y = \lambda (\varepsilon_x + \varepsilon_y + \varepsilon_z) + 2\mu \varepsilon_y - (3\lambda + 2\mu)\alpha_y(T_e - T_0),
\]
(3.4b)

\[
\sigma_z = \lambda (\varepsilon_x + \varepsilon_y + \varepsilon_z) + 2\mu \varepsilon_z - (3\lambda + 2\mu)\alpha_z(T_e - T_0),
\]
(3.4c)

and

\[
\sigma_{xy} = \mu \gamma_{xy}, \sigma_{xz} = \mu \gamma_{xz}, \sigma_{yz} = \mu \gamma_{yz},
\]
(3.4d)

\[
\varepsilon_x = \frac{\partial u}{\partial x}, \varepsilon_y = \frac{\partial v}{\partial y}, \varepsilon_z = \frac{\partial w}{\partial z},
\]
(3.4e)

\[
\gamma_{xy} = \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x}, \gamma_{xz} = \frac{\partial u}{\partial z} + \frac{\partial w}{\partial x}, \gamma_{yz} = \frac{\partial v}{\partial z} + \frac{\partial w}{\partial y}.
\]
(3.4f)

Here, \(w\) is the displacement in the thickness direction (z-direction) and \(u, v\) are the displacement in the \(x, y\) directions, respectively; \(\varepsilon_x, \varepsilon_y, \varepsilon_z\) are the normal strains in the \(x, y\) and \(z\) directions, respectively; \(\gamma_{xy}\) is the shear strain in the \(xy\) - plane; \(\gamma_{xz}\) is the shear strain in the \(xz\) - plane; \(\gamma_{yz}\) is the shear strain in the \(yz\) - plane; \(\sigma_x, \sigma_y, \sigma_z\) are the normal stresses in the \(x, y\) and \(z\) directions, respectively; \(\sigma_{xy}\) is the shear stress in the \(xy\) - plane; \(\sigma_{xz}\) is the shear strain in the \(xz\) - plane; \(\sigma_{yz}\) is the shear strain in the \(yz\) - plane; \(T_e\) and \(T_l\) are electron and lattice temperatures, respectively; \(T_0\) is the initial temperature; \(\rho\) is the density; \(\Lambda\) is the electron-blast coefficient; \(\lambda = K - \frac{2}{3} \mu\) [Reismann 1980] is
Lame constant; $K$ is the bulk modulus; $\mu$ is the shear modulus; and $\alpha_\gamma$ is the thermal expansion coefficient.

Figure 3.1. A 3D thin film with the dimension of $100\mu m \times 100 \mu m \times 0.1 \mu m$, irradiated by ultrashort-pulsed lasers.


$$C_e(T_e) \frac{\partial T_e}{\partial t} = \frac{\partial}{\partial x} \left[ k_e(T_e,T_i) \frac{\partial T_e}{\partial x} \right] + \frac{\partial}{\partial y} \left[ k_e(T_e,T_i) \frac{\partial T_e}{\partial y} \right] + \frac{\partial}{\partial z} \left[ k_e(T_e,T_i) \frac{\partial T_e}{\partial z} \right] - G(T_e - T_i) + Q,$$

$$C_l \frac{\partial T_l}{\partial t} = G(T_e - T_i) + (3\lambda + 2\mu)\alpha_\gamma T_0 \frac{\partial}{\partial t}(\varepsilon_x + \varepsilon_y + \varepsilon_z),$$

where the heat source is considered to be a Gaussian distribution and is given by [Tzou 1997]:
\[ Q(x, y, z, t) = 0.94J \frac{1 - R}{\tau_e \delta} \exp \left[ -\frac{z}{\delta} - \frac{(x - x_0)^2 + (y - y_0)^2}{r_s^2} - 2.77\left(\frac{t - 2t_p}{t_p}\right)^2 \right], \] (3.7)

Here, \( C_e(T_e) = C_{e0} \left( \frac{T_e}{T_0} \right) \) is the electron heat capacity, \( k_e(T_e, T_i) = k_0 \left( \frac{T_e}{T_i} \right) \) is the thermal conductivity, \( G \) is the electron-lattice coupling factor, \( C_l \) is the lattice heat capacity, \( Q \) is the energy absorption rate, \( J \) is the laser fluence, \( R \) is the surface reflectivity, and \( t_p \) is the laser pulse duration, \( \delta \) is the optical penetration depth, \( r_s \) is the spatial profile parameter. In addition, 0.94 and 2.77 in Eq. (3.7) are given in [Qiu 1992, Tzou 1997, Chen 2002].

3.1.2 Initial and Boundary Conditions

The boundary conditions are assumed to be stress free and thermally insulated [Bruno 1997, Chen 2002, Swartz 1989]:

\[ \sigma_x = 0, \quad \sigma_y = 0, \quad \sigma_z = 0, \text{ at } x = 0, L_x, \] (3.8a)

\[ \sigma_y = 0, \quad \sigma_{xy} = 0, \quad \sigma_{yz} = 0, \text{ at } y = 0, L_y. \] (3.8b)

\[ \sigma_z = 0, \quad \sigma_{xz} = 0, \quad \sigma_{yz} = 0, \text{ at } z = 0, L_z. \] (3.8c)

\[ \frac{\partial T_e}{\partial \vec{n}} = 0, \quad \frac{\partial T_i}{\partial \vec{n}} = 0. \] (3.9)

where \( \vec{n} \) is the unit outward normal vector on the boundary. It should be pointed out that insulated boundaries are imposed due to the assumption that no heat is lost from the film surfaces in the short time response [Tzou 1997].
The initial conditions are assumed to be

\[ T_e = T_i = T_0, \quad u = v = w = 0, \quad u_i = v_i = w_i = 0, \quad \text{at} \ t = 0. \]  

(3.10)

Solving the above mathematical model analytically could be very difficult because of nonlinearity and three dimensions. Hence, we will seek a numerical method for solving the mathematical model.

### 3.2 Finite Difference Scheme

#### 3.2.1 Finite Difference Scheme

Using a similar argument as that in [Wang 2006a, Wang 2006b, Wang 2007], we first introduce three velocity components \( v_1, v_2, \) and \( v_3 \) into the model and re-write the dynamic equations of motion, Equations (3.1)-(3.4), as follows:

\[
\begin{align*}
\frac{\partial v_1}{\partial t} &= \frac{\partial u}{\partial t}, \quad \frac{\partial v_2}{\partial t} = \frac{\partial v}{\partial t}, \quad \frac{\partial v_3}{\partial t} = \frac{\partial w}{\partial t}, \\
\rho \frac{\partial v_1}{\partial t} &= \frac{\partial \sigma_x}{\partial x} + \frac{\partial \sigma_{xy}}{\partial y} + \frac{\partial \sigma_{xz}}{\partial z} + \Lambda \frac{\partial T_e^2}{\partial x}, \\
\rho \frac{\partial v_2}{\partial t} &= \frac{\partial \sigma_{xy}}{\partial x} + \frac{\partial \sigma_y}{\partial y} + \frac{\partial \sigma_{yz}}{\partial z} + \Lambda \frac{\partial T_e^2}{\partial y}, \\
\rho \frac{\partial v_3}{\partial t} &= \frac{\partial \sigma_{xz}}{\partial x} + \frac{\partial \sigma_{yz}}{\partial y} + \frac{\partial \sigma_z}{\partial z} + \Lambda \frac{\partial T_e^2}{\partial z},
\end{align*}
\]

(3.11)

\[
\begin{align*}
\frac{\partial \epsilon_x}{\partial t} &= \frac{\partial v_1}{\partial x}, \quad \frac{\partial \epsilon_y}{\partial t} = \frac{\partial v_2}{\partial y}, \quad \frac{\partial \epsilon_z}{\partial t} = \frac{\partial v_3}{\partial z}, \\
\frac{\partial \gamma_{xy}}{\partial t} &= \frac{\partial v_2}{\partial x} + \frac{\partial v_1}{\partial y}, \quad \frac{\partial \gamma_{xz}}{\partial t} = \frac{\partial v_3}{\partial x} + \frac{\partial v_1}{\partial z}, \quad \frac{\partial \gamma_{yz}}{\partial t} = \frac{\partial v_2}{\partial y} + \frac{\partial v_3}{\partial y}.
\end{align*}
\]

(3.15a)

(3.15b)

To develop a finite difference scheme, we then design a staggered grid as shown in Figure 3.2,
where \( v_1 \) is placed at \((x_{\frac{i-1}{2}}, y_j, z_k)\), \( v_2 \) is placed at \((x_i, y_{\frac{j-1}{2}}, z_k)\), \( v_3 \) is placed at \((x_{\frac{i+1}{2}}, y_j, z_k)\), \( \gamma_{xy} \) and \( \sigma_{xy} \) are placed at \((x_{\frac{i-1}{2}}, y_{\frac{j-1}{2}}, z_k)\), \( \gamma_{yz} \) and \( \sigma_{yz} \) are placed at \((x_i, y_{\frac{j+1}{2}}, z_{\frac{k-1}{2}})\), \( \gamma_{zx} \) and \( \sigma_{zx} \) are placed at \((x_{\frac{i+1}{2}}, y_j, z_{\frac{k+1}{2}})\), while \( \varepsilon_x, \varepsilon_y, \varepsilon_z, \sigma_x, \sigma_y, \sigma_z, T_e \) and \( T_i \) are placed at \((x_i, y_j, z_k)\). Here, \( i \), \( j \) and \( k \) are indices with \( 1 \leq i \leq N_x + 1 \), \( 1 \leq j \leq N_y + 1 \), and \( 1 \leq k \leq N_z + 1 \), such that \( N_x \Delta x = L_x \), \( N_y \Delta y = L_y \), and \( N_z \Delta z = L_z \), where \( \Delta x, \Delta y \) and \( \Delta z \) are spatial step sizes. We denote \((v_1)''_{i+\frac{1}{2}, j, k}\), \((v_2)''_{i, j+\frac{1}{2}, k}\), and \((v_3)''_{i, j, k+\frac{1}{2}}\) as
numerical approximations of $v_i((i+\frac{1}{2})\Delta x, j\Delta y, k\Delta z, n\Delta t)$, $v_j(i\Delta x, (j+\frac{1}{2})\Delta y, k\Delta z, n\Delta t)$, and $v_k(i\Delta x, j\Delta y, (k+\frac{1}{2})\Delta z, n\Delta t)$, respectively, where $\Delta t$ is the time increment. Similar notations are used for other variables. Furthermore, we introduce the finite difference operators, $\Delta_x$ and $\delta_x$, as follows:

$$\Delta_x u^n_{i,j,k} = u^n_{i+1,j,k} - u^n_{i-1,j,k},$$

$$\delta_x u^n_{i,j,k} = u^n_{i+\frac{1}{2},j,k} - u^n_{i-\frac{1}{2},j,k}.$$ 

$\delta_x$ and $\delta_z$ are defined similar to $\delta_x$.

It should be pointed out that the staggered-grid method is often employed in computational fluid dynamics to prevent the solution from oscillations [Wang 2006a]. For example, if $v_1$ and $\varepsilon_x$ in Equation (3.15a) are placed at the same location, employing a central finite difference scheme may produce a velocity component $v_1$, a wave solution implying oscillation.

To avoid non-physical oscillations in the solution, we further employ a fourth-order compact finite difference scheme for obtaining stress derivatives, $\frac{\partial \sigma_x}{\partial x}$, $\frac{\partial \sigma_{sx}}{\partial y}$, $\frac{\partial \sigma_{sz}}{\partial z}$ and etc. in Equations (3.12)- (3.14).

For example, we calculate $\frac{\partial \sigma_x}{\partial x}$ as follows:

$$a \frac{\partial \sigma_x(i-1)}{\partial x} + b \frac{\partial \sigma_x(i)}{\partial x} + a \frac{\partial \sigma_x(i+1)}{\partial x} = \frac{\sigma_x(i+1/2) - \sigma_x(i-1/2)}{\Delta x},$$

$$2 + \frac{1}{2} \leq i \leq N_x - \frac{1}{2}, \quad (3.16)$$
where $a$ and $b$ are unknown constants. Here, we omit indices $j$, $k$, and $n$ for simplicity.

Using the Taylor series expansion, we obtain

$$\sigma_x(i+1/2) = \sigma_x(i) + \frac{\Delta x}{2} \frac{\partial \sigma_x(i)}{\partial x} + \frac{\Delta x^2}{2!} \frac{\partial^2 \sigma_x(i)}{\partial x^2} + \frac{\Delta x^3}{3!} \frac{\partial^3 \sigma_x(i)}{\partial x^3} + \frac{\Delta x^4}{4!} \frac{\partial^4 \sigma_x(i)}{\partial x^4} + O(\Delta x^5),$$

(3.17a)

$$\sigma_x(i-1/2) = \sigma_x(i) - \frac{\Delta x}{2} \frac{\partial \sigma_x(i)}{\partial x} + \frac{\Delta x^2}{2!} \frac{\partial^2 \sigma_x(i)}{\partial x^2} - \frac{\Delta x^3}{3!} \frac{\partial^3 \sigma_x(i)}{\partial x^3} + \frac{\Delta x^4}{4!} \frac{\partial^4 \sigma_x(i)}{\partial x^4} + O(\Delta x^5),$$

(3.17b)

$$\frac{\partial \sigma_x(i+1)}{\partial x} = \frac{\partial \sigma_x(i)}{\partial x} + \frac{\Delta x}{2} \frac{\partial^2 \sigma_x(i)}{\partial x^2} + \frac{\Delta x^2}{2!} \frac{\partial^3 \sigma_x(i)}{\partial x^3} + \frac{\Delta x^3}{3!} \frac{\partial^4 \sigma_x(i)}{\partial x^4} + O(\Delta x^4),$$

(3.17c)

$$\frac{\partial \sigma_x(i-1)}{\partial x} = \frac{\partial \sigma_x(i)}{\partial x} - \frac{\Delta x}{2} \frac{\partial^2 \sigma_x(i)}{\partial x^2} + \frac{\Delta x^2}{2!} \frac{\partial^3 \sigma_x(i)}{\partial x^3} - \frac{\Delta x^3}{3!} \frac{\partial^4 \sigma_x(i)}{\partial x^4} + O(\Delta x^4).$$

(3.17d)

Substituting the above equations into Equation (3.16) and comparing the corresponding terms, we obtain

$$2a + b = 1, \quad a = \frac{1}{24}, \quad b = \frac{11}{12},$$

(3.18)

with truncation error of $O(\Delta x^4)$. It should be pointed out that the dissipative term $\frac{\partial^3 \sigma_x(i)}{\partial x^3}$ has been eliminated from the truncation error. Hence, $\frac{\partial \sigma_x(i)}{\partial x}$ can be obtained by solving the following tridiagonal system

$$\frac{1}{24} \frac{\partial \sigma_x(i-1)}{\partial x} + \frac{11}{12} \frac{\partial \sigma_x(i)}{\partial x} + \frac{1}{24} \frac{\partial \sigma_x(i+1)}{\partial x} = \frac{\sigma_x(i+\frac{1}{2}) - \sigma_x(i-\frac{1}{2})}{\Delta x},$$

(3.19)

$$2 + \frac{1}{2} \leq i \leq N_x - \frac{1}{2},$$

$$\frac{\partial \sigma_x(i-\frac{1}{2})}{\partial x} = \frac{\sigma_x(2) - \sigma_x(1)}{\Delta x}, \quad \frac{\partial \sigma_x(N_x + \frac{1}{2})}{\partial x} = \frac{\sigma_x(N_x + 1) - \sigma_x(N_x)}{\Delta x}.$$

(3.20)
Using a similar argument, we can evaluate other stress derivatives in Equations (3.12)-(3.14). Hence, the implicit finite difference schemes for solving Equations (3.12)-(3.14) can be written as follows:

\[
\rho \frac{1}{\Delta t} \Delta_{\text{eff}}(v_i)_{i,j,k}^{n+1} = \frac{\partial (\sigma_x^{n+1})_{i+\frac{1}{2},j,k}}{\partial x} + \frac{\partial (\sigma_y^{n+1})_{i+\frac{1}{2},j,k}}{\partial y} + \frac{\partial (\sigma_z^{n+1})_{i+\frac{1}{2},j,k}}{\partial z} + \Lambda \frac{1}{\Delta t} \delta_x (T_e^2)^{n+1}_{i+\frac{1}{2},j,k},
\]

(3.21)

\[
\rho \frac{1}{\Delta t} \Delta_{\text{eff}}(v_y)_{i,j+\frac{1}{2},k}^{n+1} = \frac{\partial (\sigma_y^{n+1})_{i,j+\frac{1}{2},k+1}}{\partial y} + \frac{\partial (\sigma_x^{n+1})_{i,j+\frac{1}{2},k+1}}{\partial x} + \frac{\partial (\sigma_z^{n+1})_{i,j+\frac{1}{2},k+1}}{\partial z} + \Lambda \frac{1}{\Delta t} \delta_y (T_e^2)^{n+1}_{i,j+\frac{1}{2},k},
\]

(3.22)

\[
\rho \frac{1}{\Delta t} \Delta_{\text{eff}}(v_z)_{i,j+\frac{1}{2},k+1}^{n+1} = \frac{\partial (\sigma_z^{n+1})_{i,j,k+\frac{1}{2}}}{\partial z} + \frac{\partial (\sigma_x^{n+1})_{i,j,k+\frac{1}{2}}}{\partial x} + \frac{\partial (\sigma_y^{n+1})_{i,j,k+\frac{1}{2}}}{\partial y} + \Lambda \frac{1}{\Delta t} \delta_z (T_e^2)^{n+1}_{i,j,k+\frac{1}{2}}.
\]

(3.23)

On the other hand, the finite difference schemes for the rest of the governing equations can be seen as generalizations of the schemes described in [Wang 2006a] to the 3D case. We summarize these generalizations below:

\[
\frac{1}{\Delta t} \Delta_{\text{eff}}(\varepsilon_x)_{i,j,k}^{n+1} = \frac{1}{\Delta x} \delta_x (\varepsilon_x)_{i,j,k}^{n+1},
\]

(3.24a)

\[
\frac{1}{\Delta t} \Delta_{\text{eff}}(\varepsilon_y)_{i,j,k}^{n+1} = \frac{1}{\Delta y} \delta_y (\varepsilon_y)_{i,j,k}^{n+1},
\]

(3.24b)

\[
\frac{1}{\Delta t} \Delta_{\text{eff}}(\varepsilon_z)_{i,j,k}^{n+1} = \frac{1}{\Delta z} \delta_z (\varepsilon_z)_{i,j,k}^{n+1},
\]

(3.24c)

\[
\frac{1}{\Delta t} \Delta_{\text{eff}}(\gamma_{xy})_{i\pm\frac{1}{2},j\pm\frac{1}{2},k\pm\frac{1}{2}}^{n+1} = \frac{1}{\Delta y} \delta_x (\gamma_{xy})_{i\pm\frac{1}{2},j\pm\frac{1}{2},k\pm\frac{1}{2}}^{n+1} + \frac{1}{\Delta x} \delta_y (\gamma_{xy})_{i\pm\frac{1}{2},j\pm\frac{1}{2},k\pm\frac{1}{2}}^{n+1},
\]

(3.25a)

\[
\frac{1}{\Delta t} \Delta_{\text{eff}}(\gamma_{yz})_{i\pm\frac{1}{2},j\pm\frac{1}{2},k\pm\frac{1}{2}}^{n+1} = \frac{1}{\Delta z} \delta_x (\gamma_{yz})_{i\pm\frac{1}{2},j\pm\frac{1}{2},k\pm\frac{1}{2}}^{n+1} + \frac{1}{\Delta x} \delta_z (\gamma_{yz})_{i\pm\frac{1}{2},j\pm\frac{1}{2},k\pm\frac{1}{2}}^{n+1},
\]

(3.25b)

\[
\frac{1}{\Delta t} \Delta_{\text{eff}}(\gamma_{zx})_{i\pm\frac{1}{2},j\pm\frac{1}{2},k\pm\frac{1}{2}}^{n+1} = \frac{1}{\Delta z} \delta_y (\gamma_{zx})_{i\pm\frac{1}{2},j\pm\frac{1}{2},k\pm\frac{1}{2}}^{n+1} + \frac{1}{\Delta y} \delta_z (\gamma_{zx})_{i\pm\frac{1}{2},j\pm\frac{1}{2},k\pm\frac{1}{2}}^{n+1},
\]

(3.25c)
\( (\sigma_{x})_{i,j,k}^{n+1} = \lambda [(\varepsilon_{x})_{i,j,k}^{n+1} + (\varepsilon_{y})_{i,j,k}^{n+1} + (\varepsilon_{z})_{i,j,k}^{n+1}] + 2\mu (\varepsilon_{x})_{i,j,k}^{n+1} - (3\lambda + 2\mu)\alpha_T [(T_e)_{i,j,k}^{n+1} - T_0] \),  
(3.26a) 

\( (\sigma_{y})_{i,j,k}^{n+1} = \lambda [(\varepsilon_{x})_{i,j,k}^{n+1} + (\varepsilon_{y})_{i,j,k}^{n+1} + (\varepsilon_{z})_{i,j,k}^{n+1}] + 2\mu (\varepsilon_{y})_{i,j,k}^{n+1} - (3\lambda + 2\mu)\alpha_T [(T_e)_{i,j,k}^{n+1} - T_0] \),  
(3.26b) 

\( (\sigma_{z})_{i,j,k}^{n+1} = \lambda [(\varepsilon_{x})_{i,j,k}^{n+1} + (\varepsilon_{y})_{i,j,k}^{n+1} + (\varepsilon_{z})_{i,j,k}^{n+1}] + 2\mu (\varepsilon_{z})_{i,j,k}^{n+1} - (3\lambda + 2\mu)\alpha_T [(T_e)_{i,j,k}^{n+1} - T_0] \),  
(3.26c) 

\( (\sigma_{xy})_{i+\frac{1}{2},j,k}^{n+1} = \mu (\gamma_{xy})_{i+\frac{1}{2}}^{n+1} \),  
(3.27a) 

\( (\sigma_{xz})_{i+\frac{1}{2},j,k}^{n+1} = \mu (\gamma_{xz})_{i+\frac{1}{2},j,k}^{n+1} \),  
(3.27b) 

\( (\sigma_{yz})_{i+\frac{1}{2},j,k}^{n+1} = \mu (\gamma_{yz})_{i+\frac{1}{2},j,k}^{n+1} \),  
(3.27c) 

\[ \frac{1}{\Delta t} \Delta_{x} u_{i+\frac{1}{2},j,k}^{n+1} = (v_1)_{i+\frac{1}{2}}^{n+1} \],  
(3.28a) 

\[ \frac{1}{\Delta t} \Delta_{y} v_{i+\frac{1}{2},j,k}^{n+1} = (v_2)_{i+\frac{1}{2}}^{n+1} \],  
(3.28b) 

\[ \frac{1}{\Delta t} \Delta_{z} w_{i+\frac{1}{2},j,k}^{n+1} = (v_3)_{i+\frac{1}{2}}^{n+1} \],  
(3.28c) 

\[ C_{e0} \frac{(T_e)_{i,j,k}^{n+1} + (T_e)_{i,j,k}^{n}}{2T_0} \cdot \frac{1}{\Delta t} \Delta_{x} (T_e)_{i+\frac{1}{2},j,k}^{n+1} \]

\[ = \frac{1}{2\Delta x^2} ((k_e)_{i+\frac{1}{2},j,k}^{n+1} \delta_x (T_e)_{i+\frac{1}{2},j,k}^{n+1} - (k_e)_{i-\frac{1}{2},j,k}^{n+1} \delta_x (T_e)_{i-\frac{1}{2},j,k}^{n+1}) \]
\[ + \frac{1}{2\Delta y^2} ((k_e)_{i,j+\frac{1}{2},k}^{n+1} \delta_y (T_e)_{i,j+\frac{1}{2},k}^{n+1} - (k_e)_{i,j-\frac{1}{2},k}^{n+1} \delta_y (T_e)_{i,j-\frac{1}{2},k}^{n+1}) \]
\[ + \frac{1}{2\Delta z^2} ((k_e)_{i,j,k+\frac{1}{2}}^{n+1} \delta_z (T_e)_{i,j,k+\frac{1}{2}}^{n+1} - (k_e)_{i,j,k-\frac{1}{2}}^{n+1} \delta_z (T_e)_{i,j,k-\frac{1}{2}}^{n+1}) \]  
(3.29)
To complete the formulation of our numerical method, we now turn our attention to the approximation of boundary and initial conditions:

\[
\begin{align*}
\sigma_x^n_{i,j,k} &= (\sigma_x)_{N_x+1,i,j,k} = 0, \quad 1 \leq j \leq N_y + 1, 1 \leq k \leq N_z + 1, \\
\sigma_y^n_{i+1/2,j,k} &= (\sigma_y)_{N_x+1/2,j,k} = 0, \quad 1 \leq j \leq N_y, 1 \leq k \leq N_z, \\
\sigma_z^n_{i,j,k+1/2} &= (\sigma_z)_{N_x,i,j,k+1/2} = 0, \quad 1 \leq j \leq N_y, 1 \leq k \leq N_z, \\
\sigma_{yz}^n_{i+1/2,j+1/2} &= (\sigma_{yz})_{N_x+1/2,i+1/2,j} = 0, \quad 1 \leq i \leq N_x + 1, 1 \leq k \leq N_z + 1, \\
\sigma_x^n_{i+1/2,j,k} &= (\sigma_x)_{N_x+1/2,i+1/2,j} = 0, \quad 1 \leq i \leq N_x, 1 \leq k \leq N_z, \\
\sigma_y^n_{i,j+1/2,k} &= (\sigma_y)_{N_y,i,j+1/2} = 0, \quad 1 \leq i \leq N_x, 1 \leq k \leq N_z, \\
\sigma_z^n_{i,j+1/2} &= (\sigma_z)_{i+1/2,j,N_z+1} = 0, \quad 1 \leq j \leq N_y + 1, \\
\sigma_{xy}^n_{i+1/2,j} &= (\sigma_{xy})_{i+1/2,j,N_y+1} = 0, \quad 1 \leq i \leq N_x, 1 \leq j \leq N_y.
\end{align*}
\]
\[(\sigma_{xy})_{i+\frac{1}{2},j+\frac{1}{2}}^n = (\sigma_{xy})_{i+\frac{1}{2},j+\frac{1}{2}}^{n+1} = 0, \quad 1 \leq i \leq N_x, \quad 1 \leq j \leq N_y, \quad (3.33c)\]

\[(\sigma_{yx})_{i+\frac{1}{2},j+\frac{1}{2}}^n = (\sigma_{yx})_{i+\frac{1}{2},j+\frac{1}{2}}^{n+1} = 0, \quad 1 \leq i \leq N_x, \quad 1 \leq j \leq N_y, \quad (3.33d)\]

\[(T_e)_{i,j,k}^n = (T_e)_{i+1,j+1,k}^n, \quad (T_e)_{N_x+1,j,k}^n = (T_e)_{i,j,k}^n \quad (3.34a)\]

\[(T_e)_{i,j,1}^n = (T_e)_{i,j+1,2}^n, \quad (T_e)_{i,j,N_y+1,k}^n = (T_e)_{i,j,k}^n \quad (3.34b)\]

\[(T_e)_{i,j,1}^n = (T_e)_{i,j+1,2}^n, \quad (T_e)_{i,j,N_y+1,k}^n = (T_e)_{i,j,k}^n \quad (3.34c)\]

\[(T_i)_{i,j,k}^n = (T_i)_{i,j+1,2}^n, \quad (T_i)_{i+1,j,k+1}^n = (T_i)_{i,j,k}^n \quad (3.35a)\]

\[(T_i)_{i,j,k}^n = (T_i)_{i,j+1,2}^n, \quad (T_i)_{i,j,N_y+1,k}^n = (T_i)_{i,j,k}^n \quad (3.35b)\]

\[(T_i)_{i,j,1}^n = (T_i)_{i,j+1,2}^n, \quad (T_i)_{i,j,N_y+1,k}^n = (T_i)_{i,j,k}^n \quad (3.35c)\]

where \(1 \leq i \leq N_x + 1, \quad 1 \leq j \leq N_y + 1, \quad 1 \leq k \leq N_z + 1\), for any time level \(n\). The initial conditions are approximated as

\[u_{i,j,k}^0 = v_{i,j,k}^0 = w_{i,j,k}^0 = 0, \quad (3.36a)\]

\[(v_1)_{i,j,k}^0 = (v_2)_{i,j,k}^0 = (v_3)_{i,j,k}^0 = 0, \quad (3.36b)\]

\[(T_e)_{i,j,k}^0 = (T_i)_{i,j,k}^0 = T_0, \quad (3.36c)\]

\[(\epsilon_x)_{i,j,k}^0 = (\epsilon_y)_{i,j,k}^0 = (\epsilon_z)_{i,j,k}^0 = 0, \quad (3.36d)\]

\[(\sigma_{xx})_{i,j,k}^0 = (\sigma_{yy})_{i,j,k}^0 = (\sigma_{zz})_{i,j,k}^0 = 0, \quad (3.36e)\]

\[(\sigma_{xy})_{i,j,k}^0 = (\sigma_{yx})_{i,j,k}^0 = (\sigma_{yz})_{i,j,k}^0 = 0, \quad (3.36f)\]

\[(\sigma_{xz})_{i,j,k}^0 = (\sigma_{yz})_{i,j,k}^0 = (\sigma_{zz})_{i,j,k}^0 = 0, \quad (3.36g)\]
\[(\sigma_{yx})_{t,j,k}^0 = (\gamma_{yx})_{t,j,k}^0 = 0.\]  

(3.36h)

where \(1 \leq i \leq N_x + 1, 1 \leq j \leq N_y + 1, 1 \leq k \leq N_z + 1\), for any time level \(n\).

### 3.2.2 General Algorithm

It also should be pointed out that Equations (3.21)-(3.23) are nonlinear since the terms \(\delta_z((T_e)_{i,j,k}^{n+1})^2\), \(\delta_y((T_e)_{i,j,k}^{n+1})^2\) and \(\delta_z((T_e)_{i,j,k}^{n+1})^2\) are nonlinear. Also, it can be seen that Equation (3.28) is nonlinear. Therefore, the above scheme must be solved iteratively. An iterative method for solving the above scheme at time level \(n + 1\) is developed as follows:

1. Set the initial values for \((\varepsilon_x)^{n+1}\), \((\varepsilon_y)^{n+1}\), \((\varepsilon_z)^{n+1}\), \((\gamma_{xy})^{n+1}\), \((\gamma_{xz})^{n+1}\) and \((\gamma_{yx})^{n+1}\), solve iteratively Equations (3.33) and (3.34) coupled with the interfacial conditions, Equations (3.28)-(3.29), for \((T_e)^{n+1}\) and \((T_l)^{n+1}\).

2. Solve for \((\sigma_x)^{n+1}\), \((\sigma_y)^{n+1}\), \((\sigma_z)^{n+1}\), \((\sigma_{xy})^{n+1}\), \((\sigma_{xz})^{n+1}\) and \((\sigma_{yz})^{n+1}\) using Equations (3.26)-(3.27).

3. Solve for the derivatives of \((\sigma_x)^{n+1}\), \((\sigma_y)^{n+1}\), \((\sigma_z)^{n+1}\), \((\sigma_{xy})^{n+1}\), \((\sigma_{xz})^{n+1}\) and \((\sigma_{yz})^{n+1}\) using Equations (3.19)-(3.20) or similar equations.

4. Solve for \((v_1)^{n+1}\), \((v_2)^{n+1}\) and \((v_3)^{n+1}\) using Equations (21)-(23)

5. Update \((\varepsilon_x)^{n+1}\), \((\varepsilon_y)^{n+1}\), \((\varepsilon_z)^{n+1}\), \((\gamma_{xy})^{n+1}\), \((\gamma_{xz})^{n+1}\) and \((\gamma_{yx})^{n+1}\), using Equations (3.24)-(3.25).

Given the required accuracy \(\xi_1\) (for temperature) and \(\xi_2\) (for strain), repeat the above steps until a convergent solution is obtained based on the following criteria.
\[ (T_e)_{i,j,k}^{n+1(new)} - (T_e)_{i,j,k}^{n+1(old)} \leq \xi_1, \]
\[ |(\varepsilon_x)_{i,j,k}^{n+1(new)} - (\varepsilon_x)_{i,j,k}^{n+1(old)}| \leq \xi_2, \]
\[ |(\varepsilon_y)_{i,j,k}^{n+1(new)} - (\varepsilon_y)_{i,j,k}^{n+1(old)}| \leq \xi_2, \]
\[ |(\varepsilon_z)_{i,j,k}^{n+1(new)} - (\varepsilon_z)_{i,j,k}^{n+1(old)}| \leq \xi_2, \]
\[ |(\gamma_{xy})_{i,j,k}^{n+1(new)} - (\gamma_{xy})_{i,j,k}^{n+1(old)}| \leq \xi_2, \]
\[ |(\gamma_{xz})_{i,j,k}^{n+1(new)} - (\gamma_{xz})_{i,j,k}^{n+1(old)}| \leq \xi_2, \]
\[ |(\gamma_{yz})_{i,j,k}^{n+1(new)} - (\gamma_{yz})_{i,j,k}^{n+1(old)}| \leq \xi_2. \]

3.2.3 **Algorithm for Calculating Electron Lattice Temperature**

It should be pointed out that \((T_e)^{n+1}\) and \((T_l)^{n+1}\) are solved based on Jacobi iteration. To this end, we solve \((T_e)^{n+1}_{i,j,k}\) from Eq. (3.30) and then substitute it into Eq. (3.29) so that we obtain an equation only containing \((T_e)^{n+1}_{i,j,k}\). Thus, Jacobi iteration is employed to obtain \((T_e)^{n+1}_{i,j,k}\). Once \((T_e)^{n+1}_{i,j,k}\) is obtained, \((T_l)^{n+1}_{i,j,k}\) can be obtained the same way. Below are the equations for calculating electron temperature and lattice temperature.

We can obtain the lattice temperature from Equation (3.38):

\[
(T_l)_{i,j,k}^{n+1} = \frac{d}{(1+d)}(T_e)_{i,j,k}^{n+1} + \frac{d}{(1+d)}((T_e)_{i,j,k}^n - (T_l)_{i,j,k}^n) + \frac{1}{(1+d)}(T_l)_{i,j,k}^n

- \frac{ee}{(1+d)} \left[ (\varepsilon_x)_{i,j,k}^{n+1} + (\varepsilon_y)_{i,j,k}^{n+1} + (\varepsilon_z)_{i,j,k}^{n+1} - ((\varepsilon_x)_{i,j,k}^n + (\varepsilon_y)_{i,j,k}^n + (\varepsilon_z)_{i,j,k}^n) \right] .
\]

(3.38)
where

\[ d = \frac{G \cdot \Delta t}{2C_i}, \quad (3.39) \]

\[ ee = \frac{(3\lambda + 2\mu)\alpha_T \cdot T_0}{C_i} \quad (3.40) \]

Then we can easily calculate the electron temperature from Equation (3.41).

\[
(T_e)_{i,j}^{n+1} = \frac{1}{(a + b_1 + b_2 + b_3 + b_4 + b_5 + b_6 + \frac{G\Delta t}{2} - \frac{G\Delta t}{2} \cdot \frac{d}{1+d})}
\times (b_1 (T_e)_{i+1,j,k}^{n+1} + b_2 (T_e)_{i-1,j,k}^{n+1} + b_3 (T_e)_{i,j+1,k}^{n+1} + b_4 (T_e)_{i,j-1,k}^{n+1})
+ b_5 (T_e)_{i,j,k+1}^{n+1} + b_6 (T_e)_{i,j,k-1}^{n+1}
+ c_1 ((T_e)_{i+1,j,k}^{n} - (T_e)_{i,j,k}^{n}) - ((T_e)_{i,j,k}^{n} - (T_e)_{i-1,j,k}^{n})
+ c_3 ((T_e)_{i,j+1,k}^{n} - (T_e)_{i,j,k}^{n}) - c_4 ((T_e)_{i,j,k}^{n} - (T_e)_{i,j-1,k}^{n})
+ c_5 ((T_e)_{i,j,k+1}^{n} - (T_e)_{i,j,k}^{n}) - c_6 ((T_e)_{i,j,k}^{n} - (T_e)_{i,j,k-1}^{n})
+ \frac{G\Delta t}{2} \left( \frac{d}{1+d} \right) ((T_e)_{i,j,k}^{n} - (T_e)_{i,j,k}^{n}) + \frac{G\Delta t}{2} \left( \frac{d}{1+d} \right) ((T_e)_{i,j,k}^{n} - (T_e)_{i,j,k}^{n})
\times \left[ ((e_x)_{i,j,k}^{n+1} + (e_y)_{i,j,k}^{n+1} + (e_z)_{i,j,k}^{n+1}) - ((e_x)_{i,j,k}^{n} + (e_y)_{i,j,k}^{n} + (e_z)_{i,j,k}^{n}) \right]
(3.41)
\]

where the electron heat capacity \( C_e(T_e) \) can be obtained below [Zhang 2008a, 2008b]:

\[
C_e(T_e) = C_{e0} \left( \frac{T_e}{T_0} \right) = C_{e0} \left( \frac{T_e^{n+1} + T_e^n}{2T_0} \right), \quad (3.42)
\]
the thermal conductivity \( k_e(T_e, T_l) \) is calculated based on

\[
k_e(T_e, T_l) = k_0 \frac{T_e}{T_l},
\]

(3.43)

and constants \( a, b_i \) (i=1, ..., 6) and \( c_i \) (i=1, ..., 6) are given as follows:

\[
a = C_0 \frac{(T_e)_{i,j,l}^{n+1} + (T_e)_{i,j,k}^{n}}{2T_0},
\]

(3.44)

\[
b_1 = \frac{\frac{(T_e)_{i+,j,l}^{n+1}}{2 \Delta x^2}}{\Delta t} = k_0 \frac{\frac{(T_e)_{i+,j,l}^{n+1} + (T_e)_{i+,j,k}^{n+1}}{2}}{\Delta t},
\]

(3.45)

\[
b_2 = \frac{\frac{(T_e)_{i-,j,l}^{n+1}}{2 \Delta x^2}}{\Delta t} = k_0 \frac{\frac{(T_e)_{i-,j,l}^{n+1} + (T_e)_{i-,j,k}^{n+1}}{2}}{\Delta t},
\]

(3.46)

\[
b_3 = \frac{\frac{(T_e)_{i,j+,l}^{n+1}}{2 \Delta y^2}}{\Delta t} = k_0 \frac{\frac{(T_e)_{i,j+,l}^{n+1} + (T_e)_{i,j+,k}^{n+1}}{2}}{\Delta t},
\]

(3.47)

\[
b_4 = \frac{\frac{(T_e)_{i,j-,l}^{n+1}}{2 \Delta y^2}}{\Delta t} = k_0 \frac{\frac{(T_e)_{i,j-,l}^{n+1} + (T_e)_{i,j-,k}^{n+1}}{2}}{\Delta t},
\]

(3.48)

\[
b_5 = \frac{\frac{(T_e)_{i,j,k+,l}^{n+1}}{2 \Delta z^2}}{\Delta t} = k_0 \frac{\frac{(T_e)_{i,j,k+,l}^{n+1} + (T_e)_{i,j,k+,k}^{n+1}}{2}}{\Delta t},
\]

(3.49)

\[
b_6 = \frac{\frac{(T_e)_{i,j,k-,l}^{n+1}}{2 \Delta z^2}}{\Delta t} = k_0 \frac{\frac{(T_e)_{i,j,k-,l}^{n+1} + (T_e)_{i,j,k-,k}^{n+1}}{2}}{\Delta t},
\]

(3.50)

\[
c_1 = \frac{\frac{(T_e)_{i+,j,1}^{n+1} + (T_e)_{i+,j,k}^{n}}{2 \Delta x^2}}{\Delta t} = k_0 \frac{\frac{(T_e)_{i+,j,1}^{n+1} + (T_e)_{i+,j,k}^{n}}{2}}{\Delta t},
\]

(3.51)
\[
\begin{align*}
\text{c}_2 &= \frac{(k_e)^{n_{i-\frac{1}{2},j,k}}}{2\Delta x^2} \cdot \Delta t = \frac{k_0}{2} \frac{(T_e)_{i,j,1} + (T_e)_{i-1,j,1}}{(T_l)_{i,j,1} + (T_l)_{i-1,j,1}} \cdot \Delta t \quad \text{(3.52)} \\
\text{c}_3 &= \frac{(k_e)^{n_{i,j+\frac{1}{2},k}}}{2\Delta y^2} \cdot \Delta t = \frac{k_0}{2} \frac{(T_e)_{i,j,1} + (T_e)_{i,j-1,1}}{(T_l)_{i,j,1} + (T_l)_{i,j-1,1}} \cdot \Delta t \quad \text{(3.53)} \\
\text{c}_4 &= \frac{(k_e)^{n_{i,j-\frac{1}{2},k}}}{2\Delta y^2} \cdot \Delta t = \frac{k_0}{2} \frac{(T_e)_{i,j,1} + (T_e)_{i,j-1,1}}{(T_l)_{i,j,1} + (T_l)_{i,j-1,1}} \cdot \Delta t \quad \text{(3.54)} \\
\text{c}_5 &= \frac{(k_e)^{n_{i,j,k+\frac{1}{2}}}}{2\Delta z^2} \cdot \Delta t = \frac{k_0}{2} \frac{(T_e)_{i,j,1} + (T_e)_{i,j,k}}{(T_l)_{i,j,1} + (T_l)_{i,j,k}} \cdot \Delta t \quad \text{(3.55)} \\
\text{c}_6 &= \frac{(k_e)^{n_{i,j,k-\frac{1}{2}}}}{2\Delta z^2} \cdot \Delta t = \frac{k_0}{2} \frac{(T_e)_{i,j,1} + (T_e)_{i,j,k-1}}{(T_l)_{i,j,1} + (T_l)_{i,j,k-1}} \cdot \Delta t \quad \text{(3.56)}
\end{align*}
\]
CHAPTER 4

3D DOUBLE-LAYERED MATHEMATICAL MODEL
AND FINITE DIFFERENCE SCHEMES

In this chapter, we consider a 3D double-layered thin film exposed to ultrashort-pulsed lasers. We will purpose a mathematical model for studying thermal deformation in the 3D thin film and then develop a numerical method for studying the temperatures, displacements, stresses and strains based on the mathematical model.

4.1 Mathematical Model

4.1.1 Governing Equations

We consider a 3D double-layered thin film in Cartesian coordinates, which is exposed to an ultrashort-pulsed laser, as shown in Figure 4.1. The governing equations for studying thermal deformation in the thin film induced by ultrashort-pulsed lasers can be expressed as follows:


\[
\rho^{(m)} \frac{\partial^2 u^{(m)}}{\partial t^2} = \frac{\partial \sigma_x^{(m)}}{\partial x} + \frac{\partial \sigma_y^{(m)}}{\partial y} + \frac{\partial \sigma_z^{(m)}}{\partial z} + 2 \Lambda^{(m)} T_e^{(m)} \frac{\partial T_e^{(m)}}{\partial x},
\]  

(4.1)
\[ \rho^{(m)} \frac{\partial^2 v^{(m)}}{\partial t^2} = \frac{\partial \sigma^{(m)}_{xy}}{\partial x} + \frac{\partial \sigma^{(m)}_{yz}}{\partial y} + \frac{\partial \sigma^{(m)}_{xz}}{\partial z} + 2 \Lambda^{(m)} \gamma^{(m)}_\epsilon \frac{\partial T^{(m)}_\epsilon}{\partial y}, \]  
\[ \rho^{(m)} \frac{\partial^2 w^{(m)}}{\partial t^2} = \frac{\partial \sigma^{(m)}_{xy}}{\partial x} + \frac{\partial \sigma^{(m)}_{yz}}{\partial y} + \frac{\partial \sigma^{(m)}_{xz}}{\partial z} + 2 \Lambda^{(m)} T^{(m)}_e \frac{\partial T^{(m)}_e}{\partial y}, \]  
\[ \sigma^{(m)}_x = \lambda^{(m)} (\varepsilon^{(m)}_x + \varepsilon^{(m)}_y + \varepsilon^{(m)}_z) + 2 \mu^{(m)} \epsilon^{(m)}_x - (3 \lambda^{(m)} + 2 \mu^{(m)}) \sigma^{(m)}_y (T^{(m)}_i - T_0), \]  
\[ \sigma^{(m)}_y = \lambda^{(m)} (\varepsilon^{(m)}_x + \varepsilon^{(m)}_y + \varepsilon^{(m)}_z) + 2 \mu^{(m)} \epsilon^{(m)}_y - (3 \lambda^{(m)} + 2 \mu^{(m)}) \sigma^{(m)}_z (T^{(m)}_i - T_0), \]  
\[ \sigma^{(m)}_z = \lambda^{(m)} (\varepsilon^{(m)}_x + \varepsilon^{(m)}_y + \varepsilon^{(m)}_z) + 2 \mu^{(m)} \epsilon^{(m)}_z - (3 \lambda^{(m)} + 2 \mu^{(m)}) \sigma^{(m)}_y (T^{(m)}_i - T_0), \]  
\[ \sigma^{(m)}_{xy} = \mu^{(m)} \gamma^{(m)}_{xy}, \quad \sigma^{(m)}_{xz} = \mu^{(m)} \gamma^{(m)}_{xz}, \quad \sigma^{(m)}_{yz} = \mu^{(m)} \gamma^{(m)}_{yz}, \]  
\[ \gamma^{(m)}_{xy} = \frac{\partial u^{(m)}}{\partial y} + \frac{\partial v^{(m)}}{\partial x}, \quad \gamma^{(m)}_{xz} = \frac{\partial u^{(m)}}{\partial z} + \frac{\partial w^{(m)}}{\partial x}, \quad \gamma^{(m)}_{yz} = \frac{\partial v^{(m)}}{\partial z} + \frac{\partial w^{(m)}}{\partial y}. \]  

Here, \( m = 1, 2 \) denotes layer 1 and layer 2, respectively; \( u^{(m)}, v^{(m)}, \) and \( w^{(m)} \) are the displacements in the \( x, y, z \) directions, respectively; \( \varepsilon^{(m)}_x, \varepsilon^{(m)}_y, \) and \( \varepsilon^{(m)}_z \) are the normal strains in the \( x, y, \) and \( z \) directions, respectively; \( \gamma^{(m)}_{xy} \) is the shear strain in the \( xy \) - plane, \( \gamma^{(m)}_{xz} \) is the shear strain in the \( xz \) - plane, \( \gamma^{(m)}_{yz} \) is the shear strain in the \( yz \) - plane; \( \sigma^{(m)}_x, \sigma^{(m)}_y, \) and \( \sigma^{(m)}_z \) are the normal stresses in the \( x, y, \) and \( z \) directions, respectively; \( \sigma^{(m)}_{xy} \) is the shear stress in the \( xy \) - plane, \( \sigma^{(m)}_{xz} \) is the shear stress in the \( xz \) - plane, and \( \sigma^{(m)}_{yz} \) is the shear stress in the \( yz \) - plane; \( T^{(m)}_e \) and \( T^{(m)}_i \) are electron and lattice temperatures, respectively; \( T_0 \) is the initial temperature; \( \rho^{(m)} \) is the density; \( \Lambda^{(m)} \) is the
electron-blast coefficient; \( \lambda^{(m)} = K^{(m)} - \frac{2}{3} \mu^{(m)} \) [Reismann 1980]) and \( \mu^{(m)} \) are Lame's coefficients; and \( \alpha_{T}^{(m)} \) is the thermal expansion coefficient.

![Ultrashort-pulsed lasers](image)

Figure 4.1 A 3D double-layered thin film with the dimension of 100\( \mu \text{m} \times 100 \mu \text{m} \times 0.1 \mu \text{m} \), irradiated by ultrashort-pulsed lasers.


\[
C_e^{(m)}(T_e) \frac{\partial T_e^{(m)}}{\partial t} = \frac{\partial}{\partial x} (k_e^{(m)}(T_e, T_i) \frac{\partial T_e^{(m)}}{\partial x}) + \frac{\partial}{\partial y} (k_e^{(m)}(T_e, T_i) \frac{\partial T_e^{(m)}}{\partial y}) + \frac{\partial}{\partial z} (k_e^{(m)}(T_e, T_i) \frac{\partial T_e^{(m)}}{\partial z}) - G^{(m)}(T_e^{(m)} - T_i^{(m)}) + Q, \tag{4.5}
\]

\[
C_i^{(m)} \frac{\partial T_i^{(m)}}{\partial t} = G^{(m)}(T_e^{(m)} - T_i^{(m)}) - (3\lambda^{(m)} + 2\mu^{(m)})\alpha_{T}^{(m)} T_0 \frac{\partial}{\partial t} (\varepsilon_x^{(m)} + \varepsilon_y^{(m)} + \varepsilon_z^{(m)}), \tag{4.6}
\]

where the heat source introduced by [Chen 2002] is extended for a Gaussian laser beam focusing at \((x_0, y_0)\) on the top surface as

\[
Q(x, y, z, t) = 0.94J \frac{1 - R}{t_p z_s} \exp\left(-\frac{z}{z_s} - \frac{(x - x_0)^2}{r_s^2} - \frac{(y - y_0)^2}{r_s^2}\right) - 2.77\left(\frac{t - 2t_p}{t_p}\right)^2. \tag{4.7}
\]
Here, $C_e^{(m)}(T_e) = C_e^{(m)} \frac{T_e^{(m)}}{T_0}$ is the electron heat capacity, $k_e^{(m)}(T_e, T_l) = k_e^{(m)} \frac{T_e^{(m)}}{T_l^{(m)}}$ is the thermal conductivity, $G^{(m)}$ is the electron-lattice coupling factor, $C_l^{(m)}$ is the lattice heat capacity, $Q$ is the energy absorption rate, $J$ is the laser fluence, $R$ is the surface reflectivity, $t_p$ is the laser pulse duration, $z_s$ is the optical penetration depth, and $r_s$ is the spatial profile parameter. In addition, 0.94 and 2.77 in Equation (4.7) are given in [Qiu 1992, Tzou 1997]. Equations (4.5) and (4.6) are often referred to as parabolic two-step heat transport equations [Qiu 1994]. It should be pointed out that the term 

$$(3\lambda^{(m)} + 2\mu^{(m)}) \alpha T_0 \frac{\partial}{\partial t} (c^{(m)}_x + c^{(m)}_y + c^{(m)}_z)$$

is added in Eq. (4.6) to consider the coupling effect between lattice temperature and strain rate. However, from our experience the strain rate effect is insignificant.

### 4.1.2 Initial, Boundary and Interfacial Conditions

The boundary conditions are assumed to be stress free [Chen 2002, Tzou 2002, Lor 1999] and no heat is lost from the surface in the short time response [Tzou 1997]:

\begin{align*}
\sigma_x^{(m)} &= 0, \quad \sigma_y^{(m)} = 0, \quad \sigma_z^{(m)} = 0, \quad \text{at } x = 0, L_x, \\
\sigma_x^{(m)} &= 0, \quad \sigma_y^{(m)} = 0, \quad \sigma_z^{(m)} = 0, \quad \text{at } y = 0, L_y, \\
\sigma_x^{(m)} &= 0, \quad \sigma_y^{(m)} = 0, \quad \sigma_z^{(m)} = 0, \quad \text{at } z = 0, L_z, \\
\frac{\partial T_e^{(m)}}{\partial \tilde{n}} &= 0, \quad \frac{\partial T_i^{(m)}}{\partial \tilde{n}} = 0.
\end{align*}

(4.8a, 4.8b, 4.8c, 4.8d)

where $\tilde{n}$ is the unit outward normal vector on the boundary.
The interfacial conditions are assumed to be perfect thermal contact at \( z = \frac{L_z}{2} \) (the continuity of temperature and heat flux across the interface),

\[
\begin{align*}
    u^{(1)} &= u^{(2)}, \quad v^{(1)} = v^{(2)}, \quad w^{(1)} = w^{(2)}, \\
    \sigma_z^{(1)} &= \sigma_z^{(2)}, \quad \sigma_x^{(1)} = \sigma_x^{(2)}, \quad \sigma_y^{(1)} = \sigma_y^{(2)}, \\
    T_e^{(1)} &= T_e^{(2)}, \quad k_e^{(1)} \frac{\partial T_e^{(1)}}{\partial z} = k_e^{(2)} \frac{\partial T_e^{(2)}}{\partial z}.
\end{align*}
\]

(4.9a) (4.9b) (4.9c)

The initial conditions at \( t = 0 \) are assumed to be

\[
\begin{align*}
    T_e^{(m)} &= T_l^{(m)} = T_0, \quad u^{(m)} = v^{(m)} = w^{(m)} = 0, \\
    u_t^{(m)} &= v_t^{(m)} = w_t^{(m)} = 0.
\end{align*}
\]

(4.10)

It should be pointed out that the laser beam is applied on the top surface \( (z = 0) \) at \( t = 0 \), and the peak intensity occurs when \( t = 2t_p \). Solving the above mathematical model analytically could be very difficult because of nonlinearity and three dimensions. Hence, we seek a numerical method for solving the mathematical model.

### 4.2 Finite Difference Scheme

#### 4.2.1 Finite Difference Scheme

Following the approach in [Wang 2006a, Wang 2006b, Wang 2008, Zhang 2008b], we first introduce three velocity components \( v_1, v_2, \) and \( v_3 \) into the model, and then re-write the dynamic equations of motion, Equations (4.1)-(4.4), as follows:

\[
\begin{align*}
    v_1^{(m)} &= \frac{\partial u^{(m)}}{\partial t}, \quad v_2^{(m)} = \frac{\partial v^{(m)}}{\partial t}, \quad v_3^{(m)} = \frac{\partial w^{(m)}}{\partial t}, \\
    \rho^{(m)} \frac{\partial v_1^{(m)}}{\partial t} &= \frac{\partial \sigma_x^{(m)}}{\partial x} + \frac{\partial \sigma_{xy}^{(m)}}{\partial y} + \frac{\partial \sigma_{xz}^{(m)}}{\partial z} + 2\Lambda^{(m)} T_e^{(m)} \frac{\partial T_e^{(m)}}{\partial x},
\end{align*}
\]

(4.11) (4.12)
\[
\rho^{(m)} \frac{\partial v_2^{(m)}}{\partial t} = \frac{\partial \sigma_{xy}^{(m)}}{\partial x} + \frac{\partial \sigma_{yz}^{(m)}}{\partial y} + \frac{\partial \sigma_{zx}^{(m)}}{\partial z} + 2\Lambda^{(m)} T_e^{(m)} \frac{\partial T_e^{(m)}}{\partial y},
\]

\[
\rho^{(m)} \frac{\partial v_3^{(m)}}{\partial t} = \frac{\partial \sigma_{yx}^{(m)}}{\partial x} + \frac{\partial \sigma_{yz}^{(m)}}{\partial y} + \frac{\partial \sigma_{zx}^{(m)}}{\partial z} + 2\Lambda^{(m)} T_e^{(m)} \frac{\partial T_e^{(m)}}{\partial y},
\]

\[
\frac{\partial \epsilon_x^{(m)}}{\partial t} = \frac{\partial v_1^{(m)}}{\partial x} + \frac{\partial v_2^{(m)}}{\partial y} + \frac{\partial v_3^{(m)}}{\partial z},
\]

\[
\frac{\partial \gamma_{xy}^{(m)}}{\partial t} = \frac{\partial v_1^{(m)}}{\partial x} + \frac{\partial v_2^{(m)}}{\partial y} + \frac{\partial v_3^{(m)}}{\partial z},
\]

To develop a finite difference scheme, we then design a staggered grid as shown in Figure 4.2, where \( v_1^{(m)} \) is placed at \( (x_{i+\frac{1}{2}}, y_j, z_k) \), \( v_2^{(m)} \) is placed at \( (x_i, y_{j+\frac{1}{2}}, z_k) \), \( v_3^{(m)} \) is placed at \( (x_i, y_j, z_{k+\frac{1}{2}}) \), \( \gamma_{xy}^{(m)} \) and \( \sigma_{xy}^{(m)} \) are placed at \( (x_{i+\frac{1}{2}}, y_j, z_k) \), \( \gamma_{yz}^{(m)} \) and \( \sigma_{yz}^{(m)} \) are placed at \( (x_i, y_{j+\frac{1}{2}}, z_k) \), \( \epsilon_x^{(m)} \), \( \epsilon_y^{(m)} \), \( \epsilon_z^{(m)} \), \( \sigma_x^{(m)} \), \( \sigma_y^{(m)} \), \( \sigma_z^{(m)} \), \( T_e^{(m)} \) and \( T_l^{(m)} \) are at \( (x_i, y_j, z_k) \). Here, \( i, j, \) and \( k \) are indices with \( 1 \leq i \leq N_x + 1 \), \( 1 \leq j \leq N_y + 1 \), and \( 1 \leq k \leq N_z + 1 \), such that \( N_x \Delta x = L_x \), \( N_y \Delta y = L_y \) and \( N_z \Delta z = L_z \), where \( \Delta x \), \( \Delta y \) and \( \Delta z \) are spatial step sizes. We denote \( (v_1^{(m)})_{i+\frac{1}{2},j,k}^{n} \), \( (v_2^{(m)})_{i,j+\frac{1}{2},k}^{n} \), and \( (v_3^{(m)})_{i,j,k+\frac{1}{2}}^{n} \) as numerical approximations of \( v_1^{(m)}(i+\frac{1}{2})\Delta x, j\Delta y, k\Delta z, n\Delta t \), \( v_2^{(m)}(i\Delta x, j+\frac{1}{2})\Delta y, k\Delta z, n\Delta t \) and \( v_3^{(m)}(i\Delta x, j, k+\frac{1}{2})\Delta z, n\Delta t \), respectively, where \( \Delta t \) is the time increment. Similar notations are used for other variables.
Figure 4.2 A 3D staggered grid for a thin film and locations of variables.
Furthermore, we introduce the finite difference operators, $\Delta_x$ and $\delta_z$, as follows:

$$\Delta_x u_{i,j,k}^n = u_{i+1,j,k}^n - u_{i,j,k+1}^n,$$

$$\delta_z u_{i,j,k}^n = u_{i,j,k}^{n+1} - u_{i-1,j,k}^{n+1}.$$

Finally, $\delta_y$ and $\delta_z$ are defined similar to $\delta_x$.

To avoid non-physical oscillations in the solution, we further follow the approach in [Wang 2006a, Wang 2000b, Wang 2007, Zhang 2008a] and employ a fourth-order compact finite difference scheme for obtaining stress derivatives, $\frac{\partial \sigma_x}{\partial x}$, $\frac{\partial \sigma_y}{\partial y}$, $\frac{\partial \sigma_z}{\partial z}$ and etc. in Equations (4.12)-(4.14). For example, $\frac{\partial \sigma_x}{\partial x}$ can be obtained by solving the following tridiagonal linear system (indices $j$ and $k$ are omitted).

$$\frac{1}{24} \frac{\partial (\sigma_x^{(m)})_{i+1}}{\partial x} + \frac{11}{12} \frac{\partial (\sigma_x^{(m)})_{i}}{\partial x} + \frac{1}{24} \frac{\partial (\sigma_x^{(m)})_{i+2}}{\partial x} = \frac{(\sigma_x^{(m)})_{i+\frac{1}{2}} - (\sigma_x^{(m)})_{i-\frac{1}{2}}}{\Delta x},$$

where

$$\frac{\partial (\sigma_x^{(m)})_i}{\partial x} = \frac{(\sigma_x^{(m)})_{i+1} - (\sigma_x^{(m)})_{i-1}}{2 \Delta x}.$$

As such, the implicit finite difference schemes for solving Equations (4.11)-(4.14) coupled with Equations (4.4a)-(4.4f) can be written as follows:

$$\rho^{(m)} \frac{1}{\Delta t} \Delta_x (v_{ij,k}^{(m)})_{i+\frac{1}{2},j,k} = \frac{\partial (\sigma_x^{(m)})_{i+\frac{1}{2},j,k}}{\partial x} + \frac{\partial (\sigma_y^{(m)})_{i+\frac{1}{2},j,k}}{\partial y} + \frac{\partial (\sigma_z^{(m)})_{i+\frac{1}{2},j,k}}{\partial z}$$

$$+ \Lambda^{(m)} \frac{1}{\Delta x} \delta_x ((T_e^{(m)})_{i+\frac{1}{2},j,k}^{(m)})^2,$$
\[
\rho^{(m)} \frac{1}{\Delta t} \Delta_x (v^{(m)}_2)_{i,j,k+\frac{1}{2}}^{n+1} = \frac{\partial (\sigma^{(m)}_{xy})_{i,j+\frac{1}{2},k+\frac{1}{2}}^{n+1}}{\partial y} + \frac{\partial (\sigma^{(m)}_{xz})_{i,j,k+\frac{1}{2}}^{n+1}}{\partial x} + \frac{\partial (\sigma^{(m)}_{yz})_{i,j+\frac{1}{2},k+\frac{1}{2}}^{n+1}}{\partial z} + \Lambda^{(m)} \frac{1}{\Delta y} \delta_y ((T^{(m)}_e)_{i,j+\frac{1}{2},k+\frac{1}{2}}^{n+1})^2,
\]

\[
\rho^{(m)} \frac{1}{\Delta t} \Delta_x (v^{(m)}_3)_{i,j,k+\frac{1}{2}}^{n+1} = \frac{\partial (\sigma^{(m)}_{yz})_{i,j+\frac{1}{2},k+\frac{1}{2}}^{n+1}}{\partial y} + \frac{\partial (\sigma^{(m)}_{xz})_{i,j+\frac{1}{2},k+\frac{1}{2}}^{n+1}}{\partial x} + \frac{\partial (\sigma^{(m)}_{xy})_{i,j+\frac{1}{2},k+\frac{1}{2}}^{n+1}}{\partial z} + \Lambda^{(m)} \frac{1}{\Delta z} \delta_z ((T^{(m)}_e)_{i,j,k+\frac{1}{2}}^{n+1})^2,
\]

\[
\frac{1}{\Delta t} \Delta_x (\varepsilon^{(m)}_x)_{i,j,k} = \frac{1}{\Delta x} \delta_x (v^{(m)}_1)_{i,j,k},
\]

\[
\frac{1}{\Delta t} \Delta_x (\varepsilon^{(m)}_y)_{i,j,k} = \frac{1}{\Delta y} \delta_y (v^{(m)}_2)_{i,j,k},
\]

\[
\frac{1}{\Delta t} \Delta_x (\varepsilon^{(m)}_z)_{i,j,k} = \frac{1}{\Delta z} \delta_z (v^{(m)}_3)_{i,j,k},
\]

\[
\frac{1}{\Delta t} \Delta_y (v^{(m)}_{xy})_{i+\frac{1}{2},j+\frac{1}{2},k} = \frac{1}{\Delta y} \delta_y (v^{(m)}_1)_{i+\frac{1}{2},j+\frac{1}{2},k} + \frac{1}{\Delta x} \delta_x (v^{(m)}_2)_{i+\frac{1}{2},j+\frac{1}{2},k},
\]

\[
\frac{1}{\Delta t} \Delta_y (v^{(m)}_{yx})_{i+\frac{1}{2},j+\frac{1}{2},k+\frac{1}{2}} = \frac{1}{\Delta z} \delta_z (v^{(m)}_3)_{i+\frac{1}{2},j+\frac{1}{2},k+\frac{1}{2}} + \frac{1}{\Delta x} \delta_x (v^{(m)}_1)_{i+\frac{1}{2},j+\frac{1}{2},k+\frac{1}{2}},
\]

\[
\frac{1}{\Delta t} \Delta_y (v^{(m)}_{yz})_{i,j+\frac{1}{2},k} = \frac{1}{\Delta z} \delta_z (v^{(m)}_3)_{i,j+\frac{1}{2},k} + \frac{1}{\Delta y} \delta_y (v^{(m)}_2)_{i,j+\frac{1}{2},k},
\]

\[
(\sigma^{(m)}_{x})_{i,j,k} = \lambda^{(m)} [(\varepsilon^{(m)}_{x})_{i,j,k} + (\varepsilon^{(m)}_{y})_{i,j,k} + (\varepsilon^{(m)}_{z})_{i,j,k}] + 2 \mu^{(m)} (\varepsilon^{(m)}_{y})_{i,j,k} - T_0,
\]

\[
(\sigma^{(m)}_{y})_{i,j,k} = \lambda^{(m)} [(\varepsilon^{(m)}_{x})_{i,j,k} + (\varepsilon^{(m)}_{y})_{i,j,k} + (\varepsilon^{(m)}_{z})_{i,j,k}] + 2 \mu^{(m)} (\varepsilon^{(m)}_{x})_{i,j,k} - T_0,
\]

\[
(\sigma^{(m)}_{z})_{i,j,k} = \lambda^{(m)} [(\varepsilon^{(m)}_{x})_{i,j,k} + (\varepsilon^{(m)}_{y})_{i,j,k} + (\varepsilon^{(m)}_{z})_{i,j,k}] + 2 \mu^{(m)} (\varepsilon^{(m)}_{z})_{i,j,k} - T_0.
\]
\[
\begin{align*}
\frac{1}{1 - u(\omega, L)^2} u' \frac{\partial}{\partial x} u(\omega, L)^2 + \frac{1}{1 - u(\omega, L)^2} u' \frac{\partial}{\partial x} u(\omega, L)^2 = \frac{1}{1 - u(\omega, L)^2} u' \frac{\partial}{\partial x} u(\omega, L)^2 + \frac{1}{1 - u(\omega, L)^2} u' \frac{\partial}{\partial x} u(\omega, L)^2
\end{align*}
\]

where

\[
\begin{align*}
(1)\quad \frac{1}{1 - u(\omega, L)^2} u' \frac{\partial}{\partial x} u(\omega, L)^2 + \frac{1}{1 - u(\omega, L)^2} u' \frac{\partial}{\partial x} u(\omega, L)^2 = \frac{1}{1 - u(\omega, L)^2} u' \frac{\partial}{\partial x} u(\omega, L)^2 + \frac{1}{1 - u(\omega, L)^2} u' \frac{\partial}{\partial x} u(\omega, L)^2
\end{align*}
\]

the Crank-Nicholson finite difference method [Smith 1998] is used. On the other hand, the energy equations (4.4)-(4.6) are solved using

\[
\begin{align*}
(4.26)\quad & \frac{1}{1 - u(\omega, L)^2} u' \frac{\partial}{\partial x} u(\omega, L)^2 + \frac{1}{1 - u(\omega, L)^2} u' \frac{\partial}{\partial x} u(\omega, L)^2 = \frac{1}{1 - u(\omega, L)^2} u' \frac{\partial}{\partial x} u(\omega, L)^2 + \frac{1}{1 - u(\omega, L)^2} u' \frac{\partial}{\partial x} u(\omega, L)^2
\end{align*}
\]

\[
\begin{align*}
(4.25)\quad & \frac{1}{1 - u(\omega, L)^2} u' \frac{\partial}{\partial x} u(\omega, L)^2 + \frac{1}{1 - u(\omega, L)^2} u' \frac{\partial}{\partial x} u(\omega, L)^2 = \frac{1}{1 - u(\omega, L)^2} u' \frac{\partial}{\partial x} u(\omega, L)^2 + \frac{1}{1 - u(\omega, L)^2} u' \frac{\partial}{\partial x} u(\omega, L)^2
\end{align*}
\]

\[
\begin{align*}
(4.4)\quad & \frac{1}{1 - u(\omega, L)^2} u' \frac{\partial}{\partial x} u(\omega, L)^2 + \frac{1}{1 - u(\omega, L)^2} u' \frac{\partial}{\partial x} u(\omega, L)^2 = \frac{1}{1 - u(\omega, L)^2} u' \frac{\partial}{\partial x} u(\omega, L)^2 + \frac{1}{1 - u(\omega, L)^2} u' \frac{\partial}{\partial x} u(\omega, L)^2
\end{align*}
\]

\[
\begin{align*}
(4.24)\quad & \frac{1}{1 - u(\omega, L)^2} u' \frac{\partial}{\partial x} u(\omega, L)^2 + \frac{1}{1 - u(\omega, L)^2} u' \frac{\partial}{\partial x} u(\omega, L)^2 = \frac{1}{1 - u(\omega, L)^2} u' \frac{\partial}{\partial x} u(\omega, L)^2 + \frac{1}{1 - u(\omega, L)^2} u' \frac{\partial}{\partial x} u(\omega, L)^2
\end{align*}
\]

\[
\begin{align*}
(4.23)\quad & \frac{1}{1 - u(\omega, L)^2} u' \frac{\partial}{\partial x} u(\omega, L)^2 + \frac{1}{1 - u(\omega, L)^2} u' \frac{\partial}{\partial x} u(\omega, L)^2 = \frac{1}{1 - u(\omega, L)^2} u' \frac{\partial}{\partial x} u(\omega, L)^2 + \frac{1}{1 - u(\omega, L)^2} u' \frac{\partial}{\partial x} u(\omega, L)^2
\end{align*}
\]
\[-G^{(m)} \left( \frac{(T_e^{(m)})_{i,j,k}^{n+1}}{2} + \frac{(T_e^{(m)})_{i,j,k}^{n}}{2} - \frac{(T_f^{(m)})_{i,j,k}^{n+1}}{2} - \frac{(T_f^{(m)})_{i,j,k}^{n}}{2} \right) + Q_{i,j,k}^{n+1}, \]

\[
C^{(m)}_f \frac{1}{\Delta t} \Delta (T_f^{(m)})_{i,j,k}^{n+1} = G^{(m)} \left( \frac{(T_e^{(m)})_{i,j,k}^{n+1}}{2} + \frac{(T_e^{(m)})_{i,j,k}^{n}}{2} - \frac{(T_f^{(m)})_{i,j,k}^{n+1}}{2} - \frac{(T_f^{(m)})_{i,j,k}^{n}}{2} \right) \]

\[-\left(3\lambda^{(m)} + 2\mu^{(m)}\right) \frac{\alpha_f^{(m)}}{\Delta t} \Delta_e \left( \sigma_x^{(m)} \right)_{i,j,k}^{n+1} + \Delta_e \left( \sigma_y^{(m)} \right)_{i,j,k}^{n+1} + \Delta_e \left( \sigma_z^{(m)} \right)_{i,j,k}^{n+1}. \]

The interfacial conditions for the velocity components \(v_1^{(m)}\), \(v_2^{(m)}\), and \(v_3^{(m)}\) are obtained based on Equation (9a):

\[
(v_1^{(1)})_{i,j,N+\frac{1}{2}}^{n+1} = (v_1^{(2)})_{i,j,N+\frac{1}{2}}^{n+1}, \quad (4.28a)
\]

\[
(v_2^{(1)})_{i,j,N+\frac{1}{2}}^{n+1} = (v_2^{(2)})_{i,j,N+\frac{1}{2}}^{n+1}, \quad (4.28b)
\]

\[
(v_3^{(1)})_{i,j,N+\frac{1}{2}}^{n+1} = (v_3^{(2)})_{i,j,N+\frac{1}{2}}^{n+1}, \quad (4.28c)
\]

and from Equations (4.9b) and (4.9c)

\[
(\sigma_z^{(1)})_{i,j,N+\frac{1}{2}}^{n+1} = (\sigma_z^{(2)})_{i,j,N+\frac{1}{2}}^{n+1}, \quad (4.28d)
\]

\[
(\sigma_{xz}^{(1)})_{i,j,N+\frac{1}{2}}^{n+1} = (\sigma_{xz}^{(2)})_{i,j,N+\frac{1}{2}}^{n+1}, \quad (4.28e)
\]

\[
(\sigma_{xz}^{(1)})_{i,j,N+\frac{1}{2}}^{n+1} = (\sigma_{xz}^{(2)})_{i,j,N+\frac{1}{2}}^{n+1}, \quad (4.28f)
\]

\[
\left( k_e^{(1)} \right)_{i,j,N+\frac{1}{2}}^{n+1} = \frac{(T_e^{(1)})_{i,j,N+\frac{1}{2}}^{n+1} - (T_e^{(1)})_{i,j,N}^{n+1}}{\Delta z} = \left( k_e^{(2)} \right)_{i,j,N+\frac{1}{2}}^{n+1} \frac{(T_e^{(2)})_{i,j,N+\frac{1}{2}}^{n+1} - (T_e^{(2)})_{i,j,N}^{n+1}}{\Delta z}, \quad (4.28g)
\]

\[
(T_e^{(1)})_{i,j,N+\frac{1}{2}}^{n+1} = (T_e^{(2)})_{i,j,N+\frac{1}{2}}^{n+1}. \quad (4.28h)
\]
4.2.2 General Algorithm

It should be pointed out that Equations (4.18)-(4.20) are nonlinear since the terms $\delta_x \left( (T_{e_{(m)}}^{(n+1)})_{i,j,k} \right)^2$, $\delta_y \left( (T_{e_{(m)}}^{(n+1)})_{i,j,k} \right)^2$, and $\delta_z \left( (T_{e_{(m)}}^{(n+1)})_{i,j,k} \right)^2$ are nonlinear. Also, it can be seen that Equation (4.26) is nonlinear. Therefore, the above scheme must be solved iteratively. An iterative method for solving the above scheme at time level $n + 1$ is developed as follows:

1. Set the initial values for $(\varepsilon_x^{(m)})^{n+1}$, $(\varepsilon_y^{(m)})^{n+1}$, $(\varepsilon_z^{(m)})^{n+1}$, $(\gamma_{xy}^{(m)})^{n+1}$, $(\gamma_{xz}^{(m)})^{n+1}$, and $(\gamma_{yz}^{(m)})^{n+1}$. Solve iteratively Equations (4.26) and (4.27) coupled with the interfacial conditions, Equations (4.28c)-(4.28f), for $(T_{e_{(m)}}^{(n+1)})$ and $(T_{i}^{(n+1)})$.

2. Solve for $(\sigma_x^{(m)})^{n+1}$, $(\sigma_y^{(m)})^{n+1}$, $(\sigma_z^{(m)})^{n+1}$, $(\sigma_{xy}^{(m)})^{n+1}$, $(\sigma_{xz}^{(m)})^{n+1}$, and $(\sigma_{yz}^{(m)})^{n+1}$ using Equations (4.23)-(4.24).

3. Solve for the derivatives of $(\sigma_x^{(m)})^{n+1}$, $(\sigma_y^{(m)})^{n+1}$, $(\sigma_z^{(m)})^{n+1}$, $(\sigma_{xy}^{(m)})^{n+1}$, $(\sigma_{xz}^{(m)})^{n+1}$, and $(\sigma_{yz}^{(m)})^{n+1}$ using Equations (4.16)-(4.17) or similar equations.

4. Solve for $(v_x^{(m)})^{n+1}$, $(v_y^{(m)})^{n+1}$, and $(v_z^{(m)})^{n+1}$ using Equations (4.18)-(4.20).

5. Update $(\varepsilon_x^{(m)})^{n+1}$, $(\varepsilon_y^{(m)})^{n+1}$, $(\varepsilon_z^{(m)})^{n+1}$, $(\gamma_{xy}^{(m)})^{n+1}$, $(\gamma_{xz}^{(m)})^{n+1}$, and $(\gamma_{yz}^{(m)})^{n+1}$, using Equations (4.21)-(4.22).

Given the required accuracy $\xi_1$ (for temperature) and $\xi_2$ (for strain), repeat the above steps until a convergent solution is obtained based on the following criteria.

$$
\left| (T_{e_{(m)}}^{(n+1)})_{i,j,k} - (T_{e_{(m)}}^{(n+1)})_{i,j,k}^{(old)} \right| \leq \xi_1,
$$

$$
\left| (\varepsilon_x^{(m)})^{n+1}_{i,j,k} - (\varepsilon_x^{(m)})^{n+1}_{i,j,k}^{(old)} \right| \leq \xi_2, \left| (\varepsilon_y^{(m)})^{n+1}_{i,j,k} - (\varepsilon_y^{(m)})^{n+1}_{i,j,k}^{(old)} \right| \leq \xi_2,
$$
Using a similar argument, a numerical method can be obtained for studying thermal deformation in a double-layered micro sphere induced by an ultrashort-pulsed laser.
CHAPTER 5

NUMERICAL EXAMPLES

In this chapter, we will discuss two example cases and analyze the results to test the developed numerical scheme.

5.1 Three-Dimension Single-Layered Case

5.1.1 Example Description

To test the applicability of the developed numerical scheme, we investigated the temperature rise and thermal deformation in a 3D single-layered thin film with the dimensions 100 \( \mu \text{m} \times 100 \mu \text{m} \times 0.1 \mu \text{m} \), as shown in Figure 5.1.

Three meshes of 20\( \times \)20\( \times \)40, 20\( \times \)20\( \times \)80, 20\( \times \)20\( \times \)100 were chosen in order to test the convergence of the scheme. The time increment was chosen to be 0.005 ps and \( T_0 \) was set to be 300 K. Two different values of laser fluences (\( J = 500 \, \text{J/m}^2 \), 2000 \( \text{J/m}^2 \)) were chosen to study the hot electron blast force. The convergence criteria were chosen to be \( \xi_1 = 10^{-8} \) for temperature and \( \xi_2 = 10^{-16} \) for deformation. The thermophysical properties for gold are listed in Table 5.1 [Chen 2002, Kaye 1973, Tzou 2002].
Figure 5.1 A 3D thin film with the dimension of 100μm x 100 μm x 0.1μm, irradiated by ultrashort-pulsed lasers.

5.1.2 Results and Analysis

The first scenario is that the laser was focused on the center of the film surface. Figure 5.3a shows the change in electron temperature ($\Delta T_e/(\Delta T_e)_{\text{max}}$) at the center ($x_{\text{center}} = 50 \text{ μm}$, $y_{\text{center}} = 50 \text{ μm}$, and $z = 0 \text{ μm}$) with laser fluences $J = 500 \text{ J/ m}^2$. The maximum temperature rise of $T_e$ (i.e., $(\Delta T_e)_{\text{max}}$) is about 3755.7 K, which is close to around 3800 K obtained in [Qiu 1994]. Figure 5.3b shows the displacement ($w$) at the center ($x_{\text{center}}, y_{\text{center}}, 0$) versus time. It can be seen from both figures that mesh size has no significant effect on the solution and hence the solution is convergent.
Table 5.1 Thermal properties of gold [Chen 2002, Kaye 1973, Tzou 2002]

<table>
<thead>
<tr>
<th>Properties</th>
<th>Unit</th>
<th>Gold</th>
</tr>
</thead>
<tbody>
<tr>
<td>$\rho$</td>
<td>kg/m$^3$</td>
<td>19300</td>
</tr>
<tr>
<td>$\Lambda$</td>
<td>J/(m$^2$K$^2$)</td>
<td>70</td>
</tr>
<tr>
<td>$\lambda$</td>
<td>Pa</td>
<td>199.0×10$^9$</td>
</tr>
<tr>
<td>$\mu$</td>
<td>Pa</td>
<td>27.0×10$^9$</td>
</tr>
<tr>
<td>$\alpha_T$</td>
<td>1/K</td>
<td>14.2×10$^{-6}$</td>
</tr>
<tr>
<td>$C_{e0}$</td>
<td>J/(m$^2$K)</td>
<td>2.1×10$^4$</td>
</tr>
<tr>
<td>$C_l$</td>
<td>J/(m$^2$K)</td>
<td>2.5×10$^6$</td>
</tr>
<tr>
<td>$G$</td>
<td>W/(m$^2$K)</td>
<td>2.6×10$^6$</td>
</tr>
<tr>
<td>$k_e$</td>
<td>W/(mK)</td>
<td>315</td>
</tr>
<tr>
<td>$R$</td>
<td></td>
<td>0.93</td>
</tr>
<tr>
<td>$t_p$</td>
<td>s</td>
<td>0.1×10$^{-12}$</td>
</tr>
<tr>
<td>$z_s, \zeta$</td>
<td>m</td>
<td>15.3×10$^{-9}$</td>
</tr>
<tr>
<td>$r_s$</td>
<td>m</td>
<td>1.0×10$^{-6}$</td>
</tr>
<tr>
<td>$J$</td>
<td>J/m$^2$</td>
<td>500, 1000, 2000</td>
</tr>
</tbody>
</table>

Figure 5.4 shows electron temperature along $z$ direction at $(x_{\text{center}}, y_{\text{center}})$ with two different laser fluences ($J = 500 \text{ J/m}^2$ and $2000 \text{ J/m}^2$) at different times (a) $t = 0.25$ ps, (b) $t = 0.5$ ps, (c) $t = 1$ ps, (d) $t = 10$ ps, and (e) $t = 20$ ps, respectively. A same scale plot is displayed in Figure 5.5. Figure 5.6 shows lattice temperature along $z$ direction at $(x_{\text{center}}, y_{\text{center}})$ with two different laser fluences ($J = 500 \text{ J/m}^2$ and $2000 \text{ J/m}^2$) at different times (a) $t = 0.25$ ps, (b) $t = 0.5$ ps, (c) $t = 1$ ps, (d) $t = 10$ ps, and (e) $t = 20$ ps, respectively. It can be seen from those figures that the electron temperature rises to its maximum at the beginning and then decreases, while the lattice temperature rises gradually with time.

Figure 5.7 shows normal stress $\sigma z$ along $z$ at $(x_{\text{center}}, y_{\text{center}})$ at different times (a) $t = 1$ ps, (b) $t = 5$ ps, (c) $t = 10$ ps, and (d) $t = 15$ ps with a mesh of $20\times20\times80$ and two
different laser fluences ($J = 500 \text{ J/m}^2$ and $2000 \text{ J/m}^2$). Usually, numerical oscillations appear near the peaks of the curve, as shown in Figure 5.2 (see Figure 5.3 in [Wang 2007]). Figure 5.7 (particularly, Figure 5.7b-Figure 5.7d) indicates that the normal stress $\sigma_z$ does not show non-physical oscillations near the peak of the curve. Figures 5.8-5.15 were plotted based on the results obtained in a mesh of $40\times40\times100$ with a laser fluence of $J = 500 \text{ J/m}^2$.

![Graph showing numerical oscillations near the peaks of the curve](image)

**Figure 5.2** Numerical oscillations appearing near the peaks of the curve [Wang 2007].

Figures 5.8 and 5.9 show contours of electron temperature profile and lattice temperature profile in the cross section of $y = y_{\text{center}}$ at different times (a) $t = 0.25 \text{ ps}$, (b) $t = 0.5 \text{ ps}$, (c) $t = 1 \text{ ps}$, (d) $t = 10 \text{ ps}$, and (e) $t = 20 \text{ ps}$, respectively. It can be seen from both figures that the heat is mainly transferred along the $z$ direction. This result illustrates the fact that the femtosecond lasers are an ideal candidate for precise thermal processing of functional nanophase materials.
Figure 5.3 Change in electron temperature and displacement ($w$) at the center of top surface versus time for various grids ($20 \times 20 \times 40$, $20 \times 20 \times 80$, $20 \times 20 \times 100$) and laser fluence $J$ of 500 $J/m^2$. 
Figure 5.4 Electron temperature profiles along $z$ at $(x_{\text{center}}, y_{\text{center}})$ at different times (a) $t = 0.25$ ps, (b) $t = 0.5$ ps, (c) $t = 1$ ps, (d) $t = 10$ ps and (e) $t = 20$ ps with a mesh of $20 \times 20 \times 80$. 
Figure 5.5 Electron temperature profiles along $z$ at $(x_{\text{center}}, y_{\text{center}})$ at different times (a) $t = 0.25$ ps, (b) $t = 0.5$ ps, (c) $t = 1$ ps, (d) $t = 10$ ps and (e) $t = 20$ ps with a mesh of $20 \times 20 \times 80$ with same scale.
Figure 5.6 Lattice temperature profiles along $z$ at ($x_{\text{center}}, y_{\text{center}}$) at different times (a) $t = 0.25$ ps, (b) $t = 0.5$ ps, (c) $t = 1$ ps, (d) $t = 10$ ps and (e) $t = 20$ ps with a mesh of $20 \times 20 \times 80$ and two different laser fluences $J$ of 500 J/m$^2$ and $J$ of 2000 J/m$^2$. 
Figure 5.7 Normal stress ($\sigma_z$) profiles along $z$ at ($x_{center}, y_{center}$) at different times (a) $t = 1$ ps, (b) $t = 5$ ps, (c) $t = 10$ ps, and (d) $t = 15$ ps with a mesh of $20 \times 20 \times 80$ and two different laser fluences $J$ of 500 J/m$^2$ and 2000 J/m$^2$. 
Figure 5.8 Contours of electron temperature profiles in the cross section of y = 50 µm at different times (a) t = 0.25 ps, (b) t = 0.5 ps, (c) t = 1 ps, (d) t = 10 ps, and (e) t = 20 ps with a mesh of 20 x 20 x 80 and laser fluence J of 500 J/m².
Figure 5.9: Contours of lattice temperature profiles in the cross section of $y = 50 \, \mu m$ at different times (a) $t = 0.25$ ps, (b) $t = 0.5$ ps, (c) $t = 1$ ps, (d) $t = 10$ ps, and (e) $t = 20$ ps with a mesh of $20 \times 20 \times 80$ and laser fluence $J$ of $500 \, J/m^2$. 
Figures 5.10-5.15 show contours of displacements \((u, v, w)\) and normal stresses \(\sigma_x, \sigma_y, \sigma_z\) in the cross section of \(y = y_{\text{center}}\) at different times (a) \(t = 5\) ps, (b) \(t = 10\) ps, (c) \(t = 15\) ps, and (d) \(t = 20\) ps, respectively.

Figure 5.10 Contours of displacement \((w)\) profiles in the cross section of \(y = 0.5\) mm at different times (a) \(t = 5\) ps, (b) \(t = 10\) ps, (c) \(t = 15\) ps, and (d) \(t = 20\) ps with a mesh of \(20 \times 20 \times 80\) and laser fluence \(J\) of \(500\) J/m\(^2\).
Figures 5.10-5.12 indicate that the central part of the film is expanding because of displacement changes from negative to positive along the center line in the $z$ direction, and along $x$ and $y$ directions, respectively.

Figure 5.11 Contours of displacement ($u$) profiles in the cross section of $y = 0.5$ mm at different times (a) $t = 5$ ps, (b) $t = 10$ ps, (c) $t = 15$ ps, and (d) $t = 20$ ps with a mesh of $20 \times 20 \times 80$ and laser fluence $J$ of $500$ J/m$^2$. 
Figure 5.12 Contours of displacement ($v$) profiles in the cross section of $x = 0.5$ mm at different times (a) $t = 5$ ps, (b) $t = 10$ ps, (c) $t = 15$ ps, and (d) $t = 20$ ps with a mesh of $20 \times 20 \times 80$ and laser fluence $J$ of $500 \text{ J/m}^2$. 
Similar stress alterations can be observed from Figures 5.13-5.15. Numerical results show the displacement and stress alterations at the center along the z direction, and along x and y directions, which reveal that the central part of thin film expands.

Figure 5.13 Contours of normal stress ($\sigma_z$) profiles in the cross section of $y = 50 \mu m$ at different times (a) $t = 5 ps$, (b) $t = 10 ps$, (c) $t = 15 ps$, and (d) $t = 20 ps$ with a mesh of $20 \times 20 \times 80$ and laser fluence $J$ of $500 \text{ J/m}^2$. 
Figure 5.14 Contours of normal stress ($\sigma_z$) profiles in the cross section of $y = 0.5$ mm at different times (a) $t = 5$ ps, (b) $t = 10$ ps, (c) $t = 15$ ps, and (d) $t = 20$ ps with a mesh of $20 \times 20 \times 80$ and laser fluence $J$ of 500 J/m$^2$. 
Figure 5.15 Contours of normal stress ($\sigma_y$) profiles in the cross section of $x = 0.5$ mm at different times (a) $t = 5$ ps, (b) $t = 10$ ps, (c) $t = 15$ ps, and (d) $t = 20$ ps with a mesh of $20 \times 20 \times 80$ and laser fluence $J$ of $500$ J/m$^2$. 
The second scenario is to consider that the laser irradiates circulatively around the center of the top surface. Figure 5.16 shows contours of electron temperature profiles at the top surface of $z = 0 \mu m$ with a mesh of $20 \times 20 \times 80$ and laser fluence $J$ of 500 J/m$^2$ at different times (a) $t = 2.25$ ps, (b) $t = 4.25$ ps, (c) $t = 6.25$ ps, (d) $t = 8.25$ ps, (e) $t = 10.25$ ps, (f) $t = 12.25$ ps, (g) $t = 14.25$ ps, and (h) $t = 16.25$ ps, respectively.

As the laser irradiates circulatively around the center of the top surface, it can be seen from those figures that the electron temperature rises to its maximum in every turn at $t = 2.25$ ps, $t = 4.25$ ps, $t = 6.25$ ps, $t = 8.25$ ps, $t = 10.25$ ps, $t = 12.25$ ps, $t = 14.25$ ps, and $t = 16.25$ ps, respectively. The maximum temperature of $T_e$ is about 3845.29 K. The peak of the electron temperature also makes a circle around the center in the surface within 18 ps.
Figure 5.16 Contours of electron temperature profiles at the top surface of $z = 0$ μm at different times (a) $t = 2.25$ ps, (b) $t = 4.25$ ps, (c) $t = 6.25$ ps, (d) $t = 8.25$ ps, (e) $t = 10.25$ ps, (f) $t = 12.25$ ps, (g) $t = 14.25$ ps, and (h) $t = 16.25$ ps with a mesh of $20 \times 20 \times 80$ and laser fluence $J$ of 500 J/m$^2$. 
Figures 5.17 and 5.18 show contours of electron temperature and lattice temperature profiles at the top surface of z = 0 μm at different times (a) t = 4 ps, (b) t = 6 ps, (c) t = 8 ps, (d) t = 10 ps, (e) t = 12 ps, (f) t = 14 ps, (g) t = 16 ps, and (h) t = 18 ps with a mesh of 20×20×80 and a laser fluence J of 500 J/m². Results show that the electron temperature rises to its maximum at the beginning and then decreases within every 2 ps while the lattice temperature rises gradually with time.
Figure 5.17 Contours of electron temperature profiles at the top surface of $z = 0 \, \mu m$ at different times (a) $t = 4 \, ps$, (b) $t = 6 \, ps$, (c) $t = 8 \, ps$, (d) $t = 10 \, ps$, (e) $t = 12 \, ps$, (f) $t = 14 \, ps$, (g) $t = 16 \, ps$, and (h) $t = 18 \, ps$ with a mesh of $20 \times 20 \times 80$ and laser fluence $J$ of $500 \, J/m^2$. 
Figure 5.18 Contours of lattice temperature profiles at the top surface of z = 0 μm at different times (a) t = 4 ps, (b) t = 6 ps, (c) t = 8 ps, (d) t = 10 ps, (e) t = 12 ps, (f) t = 14 ps, (g) t = 16 ps, and (h) t = 18 ps with a mesh of 20x20x80 and laser fluence J of 500 J/m².
5.2 Three-Dimension Double-Layered Case

5.2.1 Example Description

To test the applicability of the developed numerical scheme for double-layered thin film case, we investigated the temperature rises and thermal deformations in a 3D double-layered thin film consisting of a gold layer on a chromium padding layer with dimensions $100\mu m \times 100\mu m \times 0.1\mu m$, as shown in Figure 5.19. The thermophysical properties for gold and chromium are listed in Table 5.2 [Touloukian 1970a, b, Chen 2002, Kaye 1973, Tzou 2002].

Ultrashort-pulsed lasers

Figure 5.19 A 3D double-layered thin film with the dimension of $100 \mu m \times 100 \mu m \times 0.1 \mu m$, irradiated by ultrashort-pulsed lasers.
Table 5.2 Thermophysical properties of gold and chromium [Touloukian 1970a, b, Chen 2002, Kaye 1973, Tzou 2002]

<table>
<thead>
<tr>
<th>Properties</th>
<th>Unit</th>
<th>Gold</th>
<th>Chromium</th>
<th>Others</th>
</tr>
</thead>
<tbody>
<tr>
<td>$\rho$</td>
<td>kg/m$^3$</td>
<td>19300</td>
<td>7190</td>
<td></td>
</tr>
<tr>
<td>$\Lambda$</td>
<td>J/(m$^3$K$^2$)</td>
<td>70</td>
<td>1933</td>
<td></td>
</tr>
<tr>
<td>$\lambda$</td>
<td>Pa</td>
<td>199.0×10$^9$</td>
<td>83.3×10$^9$</td>
<td></td>
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<tr>
<td>$\mu$</td>
<td>Pa</td>
<td>27.0×10$^9$</td>
<td>115.0×10$^9$</td>
<td></td>
</tr>
<tr>
<td>$\alpha_x$</td>
<td>1/K</td>
<td>14.2×10$^{-6}$</td>
<td>4.9×10$^{-6}$</td>
<td></td>
</tr>
<tr>
<td>$C_{so}$</td>
<td>J/(m$^3$K)</td>
<td>2.1×10$^4$</td>
<td>5.8×10$^4$</td>
<td></td>
</tr>
<tr>
<td>$C_i$</td>
<td>J/(m$^3$K)</td>
<td>2.5×10$^6$</td>
<td>3.3×10$^6$</td>
<td></td>
</tr>
<tr>
<td>$G$</td>
<td>W/(m$^3$K)</td>
<td>2.6×10$^{16}$</td>
<td>42×10$^{16}$</td>
<td></td>
</tr>
<tr>
<td>$k_e$</td>
<td>W/(mK)</td>
<td>315</td>
<td>94</td>
<td></td>
</tr>
<tr>
<td>$R$</td>
<td></td>
<td></td>
<td>0.93</td>
<td></td>
</tr>
<tr>
<td>$t_p$</td>
<td>s</td>
<td></td>
<td>0.1×10$^{-12}$</td>
<td></td>
</tr>
<tr>
<td>$z, \zeta$</td>
<td>m</td>
<td></td>
<td>15.3×10$^{-9}$</td>
<td></td>
</tr>
<tr>
<td>$r_s$</td>
<td>m</td>
<td></td>
<td>1.0×10$^{-6}$</td>
<td></td>
</tr>
<tr>
<td>$J$</td>
<td>J/m$^2$</td>
<td></td>
<td>500, 1000, 2000</td>
<td></td>
</tr>
</tbody>
</table>

5.2.2 Results and Analysis

We assumed that the laser was focused on the center of the top surface of the thin film. Three different values of laser fluences ($J = 500$ J/m$^2$, 1000 J/m$^2$ and 2000 J/m$^2$) were chosen to study the hot electron blast force. Three meshes of $20 \times 20 \times 60$, $20 \times 20 \times 80$, $20 \times 20 \times 100$, for each layer in $(x,y,z)$ for the thin film were chosen in order to test the convergence of the scheme. The time increment was chosen to be 0.005 ps and $T_0$ was set to be 300 K. The convergence criteria were chosen to be $\xi_1 = 10^{-8}$ for temperature and $\xi_1 = 10^{-16}$ for deformation. We investigated the temperature rises and thermal deformations in a 3D double-layered thin film consisting of a gold layer on a chromium padding layer.
Figure 5.20a shows the changes in electron temperature \((\Delta T_e/(\Delta T_e)_{max})\) at the center \((x_{center} = 50 \mu m, y_{center} = 50 \mu m\) and \(z = 0 \mu m\)) of the thin film with a laser fluence of \(J = 500 \text{ J/m}^2\). The maximum temperature rise of \(T_e\) (i.e., \((\Delta T_e)_{max}\)) is about 3765 K, which is close to the 3727 K obtained by Tzou et al. [Tzou 2002]. This is probably because of the different geometries. It can be seen from Figure 5.20a that the mesh size 20×20×60, 20×20×80, 20×20×100 had no significant effect on the solution and hence the solution is convergent.

From Figure 5.20b, the negative value of displacement \((w)\) indicates that the thin film at the center \((x_{center}, y_{center}, 0)\) is expanding along the negative \(z\) direction. It also can be seen from Figure 5.20b that the mesh size has no significant effect on the solution; hence, the solution is convergent.

Figures 5.21 shows electron temperature of the double-layered thin film along \(z\) direction at \((x_{center}, y_{center})\) with three different laser fluences \((J = 500 \text{ J/m}^2, 1000 \text{ J/m}^2\) and \(2000 \text{ J/m}^2\)) at different times \((a) t = 0.25 \text{ ps}, (b) t = 0.5 \text{ ps}, (c) t = 10 \text{ ps}, \) and \((d) t = 20 \text{ ps},\) respectively. Similarly, Figures 5.22 shows lattice temperature of the double-layered thin film along \(z\) direction at \((x_{center}, y_{center})\) with three different laser fluences \((J = 500 \text{ J/m}^2, 1000 \text{ J/m}^2\) and \(2000 \text{ J/m}^2\)) at different times \((a) t = 0.25 \text{ ps}, (b) t = 0.5 \text{ ps}, (c) t = 10 \text{ ps},\) and \((d) t = 20 \text{ ps},\) respectively. It can be seen from Figure 5.21 that the electron temperature is in maximum at \(t = 0.25 \text{ ps},\) but then it decays, with time and it is almost uniform at \(t = 20 \text{ ps}\) along the thickness direction.
Figure 5.20 (a) Change in electron temperature and (b) displacements at the center of top surface of thin versus time with a laser fluence \((J)\) of 500 J/m\(^2\). The \(w\) is the displacement at \((x_{\text{center}}, y_{\text{center}}, 0)\) of thin film.
Figure 5.21. Electron temperature profiles along $z$ at $(x_{\text{center}}, y_{\text{center}})$ at different times (a) $t = 0.25$ ps, (b) $t = 0.5$ ps, (c) $t = 10$ ps, and (d) $t = 20$ ps with a mesh of $20 \times 20 \times 80$ and three different laser fluences ($J$) of 500 J/m$^2$, 1000 J/m$^2$ and 2000 J/m$^2$. 
Figure 5.22 Lattice temperature profiles along $z$ at $(x_{\text{center}}, y_{\text{center}})$ at different times (a) $t = 0.25$ ps, (b) $t = 0.5$ ps, (c) $t = 10$ ps, and (d) $t = 20$ ps with a mesh of $20 \times 20 \times 80$ and three different laser fluences ($J$) of $500$ J/m$^2$, $1000$ J/m$^2$ and $2000$ J/m$^2$. 
On the other hand, Figure 5.22 shows that the lattice temperature increases gradually with time in both gold and chromium layers, due to constant heating of acoustic phonons by electrons. Since the heat is transferred from the gold layer to the chromium layer and the conductivity of chromium is smaller than that of gold, the lattice temperature increases drastically across the interface. A clear discontinuity of the temperature gradient at the interface can be seen in Figure 5.22, which is the same prediction as was obtained in [Qiu 1992, Tzou 2002]. The difference of electron and lattice temperatures in Figures 5.21 and 5.22 gives a strong flavor of non-equilibrium heating during the picosecond transient. It also can be seen from Figure 5.21 and Figure 5.22 that the electron temperature rises to its maximum at the beginning and then decreases while the lattice temperature rises gradually with time.

Figure 5.23 shows the displacement \( w \) of the thin film along \( z \) at \((x_{center}, y_{center})\) at different times (a) \( t = 5 \) ps, (b) \( t = 10 \) ps, (c) \( t = 15 \) ps, and (d) \( t = 20 \) ps with a mesh of \( 20 \times 20 \times 80 \) and three different laser fluences \( (J=500 \text{ J/m}^2, 1000 \text{ J/m}^2 \text{ and } 2000 \text{ J/m}^2) \). It can been seen that the displacement \( w \), particularly at \( t = 20 \) ps, changes from negative to positive for each layer along the thickness direction. The negative value indicates that the displacement moves in the negative \( z \) direction, while the positive value implies that it moves in the positive \( z \) direction. From this figure, one may see that the film is expanding. At \( t = 10 \) ps and \( 20 \) ps, the displacement shows a clear alteration across the interface, implying that both layers push each other and the bond between these two layers could be damaged under high intensity laser irradiation.
Figure 5.23 Displacement ($w$) profiles along $z$ at ($x_{\text{center}}, y_{\text{center}}$) at different times (a) $t = 5$ ps, (b) $t = 10$ ps, (c) $t = 15$ ps, and (d) $t = 20$ ps with a mesh of $20 \times 20 \times 80$ and three different laser fluences ($J$) of 500 J/m$^2$, 1000 J/m$^2$ and 2000 J/m$^2$. 
Figure 5.24 shows the normal stress $\sigma_z$ along $z$ at $(x_{\text{center}}, y_{\text{center}})$ at different times (a) $t = 5$ ps, (b) $t = 10$ ps, (c) $t = 15$ ps, and (d) $t = 20$ ps with a mesh of $20 \times 20 \times 80$ and three different laser fluences ($J = 500$ J/m$^2$, 1000 J/m$^2$, and 2000 J/m$^2$). In our experience, the conventional finite difference method produces local oscillations in the normal stress $\sigma_z$ (see Figure 5.2). Figure 5.24 indicates that the curve of $\sigma_z$ is smooth and does not appear to have local oscillations, implying that our method prevents the appearance of non-physical oscillations in the solution.

Figures 5.25-5.29 were plotted based on the results obtained with a mesh of $20 \times 20 \times 80$ and with a laser fluence of $J = 1000$ J/m$^2$. Figures 5.25 and 5.26 show contours of the electron temperature distribution and the lattice temperature distribution in the cross section of $y = y_{\text{center}}$ at different times (a) $t = 0.25$ ps, (b) $t = 0.5$ ps, (c) $t = 1$ ps, (d) $t = 10$ ps, and (e) $t = 20$ ps, respectively.

Both figures reveal that the heat is mainly transferred along the $z$ direction. This result confirms the fact that the femtosecond lasers are an ideal candidate for precise thermal processing of functional nanophase materials. Figure 5.26 also shows that there is a clear difference between the lattice temperatures in these two layers, because of the different conductivities.

Figures 5.27-5.29 show contours of displacements $(u, v, w)$ in the cross section of $y = y_{\text{center}}$ at different times (a) $t = 5$ ps, (b) $t = 10$ ps, (c) $t = 15$ ps, and (d) $t = 20$ ps, respectively. According to Figures 5.27-5.29, the central part of the film is expanding because displacements change from negative to positive along the center line in the $z$ direction, and along the $x$ and $y$ directions, respectively.
Figure 5.24 Normal stress ($\sigma_z$) profiles along $z$ at ($x_{center}$, $y_{center}$) at different times (a) $t = 5$ ps, (b) $t = 10$ ps, (c) $t = 15$ ps, and (d) $t = 20$ ps with a mesh of $20 \times 20 \times 80$ and three different laser fluences ($J$) of 500 J/m$^2$, 1000 J/m$^2$ and 2000 J/m$^2$. 
Figure 5.25 Contours of electron temperature distributions in the cross section of $y = 50 \mu m$ at different times (a) $t = 0.25$ ps, (b) $t = 0.5$ ps, (c) $t = 1$ ps, (d) $t = 10$ ps, and (e) $t = 20$ ps with a mesh of $20 \times 20 \times 80$ and a laser fluence ($J$) of 1000 J/m$^2$.
Figure 5.26 Contours of lattice temperature distributions in the cross section of $y = 50$ μm at different times (a) $t = 0.25$ ps, (b) $t = 0.5$ ps, (c) $t = 1$ ps, (d) $t = 10$ ps, and (e) $t = 20$ ps with a mesh of $20 \times 20 \times 80$ and a laser fluence ($J$) of $1000 \text{ J/m}^2$. 
Figure 5.27 Contours of displacement ($w$) distributions in the cross section of $y = 50 \, \mu m$ at different times (a) $t = 5$ ps, (b) $t = 10$ ps, (c) $t = 15$ ps, and (d) $t = 20$ ps with a mesh of $20 \times 20 \times 80$ and a laser fluence ($J$) of $1000 \, J/m^2$. 
Figure 5.28 Contours of displacement ($u$) distributions in the cross section of $y = 50 \, \mu m$ at different times (a) $t = 5$ ps, (b) $t = 10$ ps, (c) $t = 15$ ps, and (d) $t = 20$ ps with a mesh of $20 \times 20 \times 80$ and a laser fluence ($J$) of $1000 \, J/m^2$. 
Figure 5.29 Contours of displacement ($v$) distributions in the cross section of $x = 50$ $\mu$m at different times (a) $t = 5$ ps, (b) $t = 10$ ps, (c) $t = 15$ ps, and (d) $t = 20$ ps with a mesh of $20 \times 20 \times 80$ and a laser fluence ($J$) of 1000 $J/m^2$. 
CHAPTER 6

CONCLUSION AND FUTURE WORK

In this dissertation, we consider a 3D single-layered thin film and a 3D double-layered thin film exposed to ultrashort-pulsed lasers. A mathematical model for describing the thermal deformation in 3D thin films is proposed. This model, based on the parabolic two-step heat transport equations, accounts for the coupling effect between lattice temperature and strain rate, as well as for the hot-electron blast effect in momentum transfer. For the 3D double-layered thin film case, perfect thermal contract at the interface was considered. Furthermore, stress free and thermal insulated conditions were considered for the boundary. Because solving the mathematical model analytically is very difficult due to nonlinearity and three dimensions, we developed a new finite difference scheme for studying thermal deformation in 3D thin films exposed to ultrashort-pulsed lasers, based on the proposed mathematical model. To avoid numerical oscillations, we first introduced velocity components into the dynamic equation of motion and replaced the displacement components. We then designed a triggered grid where the unknown variables are placed at different locations so that the checker-board solution could be eliminated. We further employed a fourth-order compact scheme to calculate stresses derivatives in the dynamic equations of motion so that the third-order derivatives
of stresses did not appear in the truncation error. To test the applicability of the developed numerical method, we considered a 3D single-layered gold thin film and a 3D double-layered gold and chromium thin film exposed to ultrashort-pulsed lasers, respectively. The numerical method was tested to be grid independent, implying that the solution is convergent. The normal stress along the $z$-direction was checked, and result showed that there were no numerical oscillations. Further, electron and lattice temperatures, displacements, stresses, and strains were obtained based on the developed finite difference schemes. Results indicate that the central part of thin films expands during heating. In the 3D double-layered case, numerical results also show that the displacements at the thickness direction at the interface move against each other, implying that the bond between two layers of gold and chromium could be damaged under high intensity laser irradiation.

Further research will focus on 3D double-layered case where the interface could be imperfect thermal contact. This condition will introduce an extra nonlinear behavior at the interface. Furthermore, pulse duration that is smaller than the electron relaxation time may be considered. For this case, the parabolic two-step model may not be appropriate and the hyperbolic two-step model needs to be considered.
APPENDIX A

SOURCE CODE FOR 3D
SINGLE-LAYERED CASE
C Main program
C Set all variable
implicit double precision (a-h,l-o-z)
dimension t(4001),x(51),y(51),z(221)
dimension TLo(41,41,101),TL0old(41,41,101),
$ TEm(4001),TLm(4001),
$ ulm(4001),u2m(4001),u3m(4001),
$ v1m(4001),v2m(4001),v3m(4001),
C Ex, Ey, Ez normal strain and shear strain
$ epx(41,41,101),epy(41,41,101),epz(41,41,101),
$ epxy(41,41,101),epxz(41,41,101),epyz(41,41,101),
$ xsao(41,41,101),ysao(41,41,101),zsao(41,41,101),
$ ssaoxy(41,41,101),ssaoxz(41,41,101),
$ ssaoyz(41,41,101),
$ epz(41,41,101),epz(41,41,101),
C Normal stress and shear stress
$ sax(41,41,101),say(41,41,101),saz(41,41,101),
$ saxy(41,41,101),saxz(41,41,101),sayz(41,41,101),
$ saxn(41,41,101),sayn(41,41,101),sazn(41,41,101),
$ saxn(41,41,101),sayn(41,41,101),sazn(41,41,101),
C Velocity displacement
$ vxo(41,41,101),vyo(41,41,101),vzo(41,41,101),
$ vx(41,41,101),vyn(41,41,101),vzn(41,41,101),
$ uxo(41,41,101),uyo(41,41,101),uzo(41,41,101),
$ uxn(41,41,101),uy(41,41,101),uzn(41,41,101),
C Stress derivatives
$ d(41,41,101),b(221),c(221),a(221),beta(221),
$ difx(41,41,101),dify(41,41,101),difz(41,41,101),
$ difxy(41,41,101),difyx(41,41,101),
$ difxz(41,41,101),difzy(41,41,101),
$ difzz(41,41,101),
C Additional set
$ u(41,41,101),u(41,41,101),u(41,41,101),
$ u(41,41,101),
$ u(41,41,101),u(41,41,101),u(41,41,101),
$ u(41,41,101),
C Data
C Lame constant
clem=199.0d+9
C Shear modulus
cmiu=27.0d+9
C Thermal expansion coefficient
alpha=14.2d-6
open(unit=8, file='etm.txt')
open(unit=7, file='um.txt')
C Dimension
lx=1.0D-4
ly=1.0D-4
lz=1.0D-7
diffyzz(i,j,k)=0.0
do i=1,nx
  do j=1,ny+1
    do k=1,nz+l
      uxo(i,j,k)=0.0
    enddo
  enddo
enddo

uxo(i,j,k)=0.0
vxo(i,j,k)=0.0
do i=1,nx+l
  doj=1,ny+l
    do k=1,nz+l
      vxo(i,j,k)=0.0
    enddo
  enddo
enddo

uxo(i,j,k)=0.0
vxo(i,j,k)=0.0
enddo
do i=1,nx
  do j=1,ny
    do k=2,nz
      epxyo(i,j,k)=0.0
    enddo
  enddo
enddo
dol=n=nt
t(n)=n*dt
tl(n)=(n-1)*dt+dt/2.0

C Guess normal and shear strain Ex, Ey, Ez, Exy, Eyz, Exz Values

di=1,nx+1
doj=1,ny+1
dok=1,nz+1
uyo(i,j,k)=0.0
vyo(i,j,k)=0.0
enddo
do i=1,nx+1
  doj=1,ny+1
    do k=1,nz
      uzo(i,j,k)=0.0
    enddo
  enddo
enddo

uxo(i,j,k)=0.0
vxo(i,j,k)=0.0
enddo
do i=1,nx+1
  doj=1,ny+1
    do k=1,nz
      vzo(i,j,k)=0.0
    enddo
  enddo
enddo

di=1,nx
  do j=2,ny
    do k=1,nz
      epxyo(i,j,k)=0.0
    enddo
  enddo
enddo
di=1,nx
  do j=2,ny
    do k=1,nz
      epzo(i,j,k)=0.0
    enddo
  enddo
enddo
do i=2,nx
  do j=1,ny
    do k=1,nz
      epzo(i,j,k)=0.0
    enddo
  enddo
enddo
do i=2,nx
  do j=1,ny
    do k=1,nz
      epzo(i,j,k)=0.0
    enddo
  enddo
enddo

n=1
TEm(n)=0.0

C Iteration

tol=1d-16
detuvmax=tol+1d-5
do while (detuvmax.gt.tol)
detuvmax=0.0
det1max=0.0
det2max=0.0
det3max=0.0
det4max=0.0
det5max=0.0
det6max=0.0

call subroutine calculate temperature

Call temp(nx,ny,nz,dx,dy,dz,x,y,z,tl(n),dt,$ TLo,TLold,TEo,TEold,$ epox,pyxo,pyzo,epyzo,epyzo)

C Compute normal stress

do j=1,ny+1
  do k=1,nz+1
    saxn(i,j,k)=0.0
  endo
enddo
doj=1,ny
  do k=1,nz
    saxn(i,j,k)=0.0
  endo
enddo
do i=1,nx
  do j=1,ny
    do k=1,nz
      gko(i,j,k)=0.0
    endo
  enddo
enddo

TEm(n)=0.0
do i=1,nx+1
  saxzn(ij,1)=0.0
  saxzn(ij,nz+1)=0.0
end do
end do

do i=2,nx
  do j=2,ny
    do k=2,nz
      saxzn(ij,k)=(clemta+2.0*cmiu)*epxn(ij,k)
      $+clemta*epyn(ij,k)
      $+clemta*epzn(ij,k)(3.0*clemta+2.0*cmiu)*alpha
      *(TLold(ij,k)-300.0)
    end do
  end do
end do

sayzn(ij,k)=clemta*epxn(ij,k)f(clemta+2.0*cmiu)
$*epyn(ij,k)
$+clemta*epzn(ij,k)(3.0*clemta+2.0*cmiu)
$*alpha*(TLold(ij,k)-300.0)

sazn(ij,k)=clemta*epxn(ij,k)+clemta*epzn(ij,k)+(clemta+2.0*cmiu)
$=epzn(ij,k)(3.0*clemta+2.0*cmiu)
$*alpha*(TLold(ij,k)-300.0)
end do
end do
end do

C Calculate shear stress

do j=1,ny
  do k=2,nz
    saxyn(1,j,k)=0.0
    saxyn(nx,j,k)=0.0
  end do
end do

saxzn(1,j,1)=0.0
saxzn(nx,j,1)=0.0
end do
end do

saxzn(ij,1)=0.0
saxzn(nx,j,nz)=0.0
end do
end do

sayzn(ij,1)=0.0
sayzn(nx,j,nz)=0.0
end do
end do

C Calculate derivative of stress difx

do k=2,nz
  do j=2,ny
    do i=2,nx
      saxyn(i,j,k)=cmiu*epxyn(i,j,k)
    end do
  end do
end do

b(2)=0.0
a(2)=11.0/12.0
c(2)=-1.0/24.0
end do
end do

b(2)=0.0
a(2)=11.0/12.0
c(2)=-1.0/24.0
end do
end do

b(i)=1.0/24.0
c(i)=1.0/24.0
end do
end do

b(i)=1.0/24.0
c(i)=1.0/24.0
end do
end do

b(i)=1.0/24.0
c(i)=1.0/24.0
end do
end do

b(i)=1.0/24.0
c(i)=1.0/24.0
end do
end do
b(nx-1)=-1.0/24.0
a(nx-1)=11.0/12.0
c(nx-1)=0.0

do k=2,nz
do i=2,nx
dif(i,1,k)=(sayn(i,2,k)-sayn(i,1,k))/dy
dif(i,ny,k)=(sayn(i,ny+1,k)-sayn(i,ny,k))/dy
do end do
end do

$ -1.0/24.0*dify(i,j,k)$

beta(nx)=0.0
do k=2,nz
do i=2,nx
df(i,j)==(sayn(i,j,k)-sayn(i,j+1,k))/dy
do end do
end do

do m=2,nx-l
i=nx-m+1
beta(i)=b(i)/(a(i)-c(i)*beta(i+1))
do j=2,ny
do k=2,nz
gama(i,j,k)=(d(i,j,k)+c(i)*gama(i+1,j,k))/(a(i)

$ -c(i)*beta(i+1))$

do end do
end do
end do

Do j=2,ny
Do k=2,nz
ul(i,j,k)=0.0
end do
end do
end do

do i=2,nx
i=j=nx-m+1
beta(i)=b(i)/(a(i)-c(i)*beta(i+1))
do j=2,ny
Do k=2,nz
gama(i,j,k)=(d(i,j,k)+c(i)*gama(i+1,j,k))/(a(i)

$ -c(i)*beta(i+1))$

do end do
end do
end do
end do

Do m=2,nx-l
i=nx-m+1
beta(i)=b(i)/(a(i)-c(i)*beta(i+1))
do j=2,ny
Do k=2,nz
gama(i,j,k)=(d(i,j,k)+c(i)*gama(i+1,j,k))/(a(i)

$ -c(i)*beta(i+1))$

do end do
end do
end do
end do

C Calculate derivative of stress dify

Do k=2,nz
do j=2,ny

$ -1.0/24.0*dify(i,j,k)$

do end do
end do

C Calculate derivative of stress dify

Do k=2,nz
do i=2,nx
diff(i,1,k)=(sayn(i,2,k)-sayn(i,1,k))/dy
diff(i,ny,k)=(sayn(i,ny+1,k)-sayn(i,ny,k))/dy
do end do
end do

$ -1.0/24.0*dify(i,j,k)$

do end do
end do

C Calculate derivative of stress dify
end do

do j=2,ny-1
do i=2,nx
do k=2,nz
u2(i,j,k)=beta(j)*u2(i,j-1,k)+gama(i,j,k)
dify(i,j,k)=u2(i,j,k)
end do
end do
end do

do i=1,nx
a(i)=0
b(i)=0
c(i)=0
beta(i)=0
do j=1,ny
do k=1,nz
gama(i,j,k)=0.0
d(i,j,k)=0.0
end do
end do
end do

C Calculate derivative of stress difz

end do
do i=2,nx
do j=2,ny
difz(i,j,1)=(sazn(i,j,2)-sazn(i,j,1))/dz
difz(i,j,nz)=(sazn(i,j,nz+1)-sazn(i,j,nz))/dz
end do
end do

b(2)=0.0
a(2)=11.0/12.0
c(2)=1.0/24.0

end do
do i=2,nx
do j=2,ny
d(i,j,2)=(sazn(i,j,3)-sazn(i,j,2))/dz-1.0/24.0*difz(i,j,1)
end do
end do

b(3)=0.0
a(3)=11.0/12.0
c(3)=-1.0/24.0

C Calculate derivative of stress difxyx

end do
do i=2,nx
do j=2,ny
d(i,j,nz-1)=(sazn(i,j,nz)-sazn(i,j,nz-1))/dz
end do
end do

b(nz-1)=-1.0/24.0
a(nz-1)=11.0/12.0
c(nz-1)=0.0

end do
do i=2,nx
do j=2,ny
d(i,j,nz-1)=(sazn(i,j,nz)-sazn(i,j,nz-1))/dz
end do
end do

b(3)=0.0
a(3)=11.0/12.0
c(3)=-1.0/24.0
do k=2, nz
  do j=1, ny
    \( d(3j,k) = (saxyn(3j,k) - saxyn(2j,k)) / dx \)
  end do
end do

$ -1.0/24.0 \cdot \text{difxyx}(2j,k)$

C Calculate derivative of stress difxyy

$ -1.0/24.0 \cdot \text{difxyy}(i,2,k)$
do m=3,ny-1
j=ny-m+2
beta(j)=b(j)/(a(j)-c(j)*beta(j+1))
do i=1,nx
do k=2,nz
gama(i,j,k)=(d(i,j,k)+c(j)*gama(i,j+1,k))/(a(j)-c(j)*beta(j+1))
end do
end do

\$ -c(j)\beta(j+1)
end do
end do

C Calculate derivative of stress difxnx

do k=1,nz
i=nx-m+2
beta(i)=b(i)/(a(i)-c(i)*beta(i+1))
do j=2,ny
do k=1,nz
gama(i,j,k)=0.0
d(i,j,k)=0.0
end do
end do
end do

b(3)=0.0
a(3)=11.0/12.0
c(3)=-1.0/24.0

\$ -1.0/24.0*\text{difxnx}(2,j,k)
end do
end do

\$ -1.0/24.0*\text{difxnx}(2,j,k)
end do
end do
C Calculate derivative of stress $d_{i,j,zz}$

```fortran
! do i=1,nx
! do j=2,ny
  d(i,j,2)=(saxzn(i,j,2)-saxzn(i,j,1))/dz  
  d(i,j,nz)=(saxzn(i,j,nz)-saxzn(i,j,nz-1))/dz
! end do
! end do

b(3)=0.0
a(3)=11.0/12.0
c(3)=-1.0/24.0

! do i=1,nx
! do j=2,ny
  d(i,j,3)=(saxzn(i,j,3)-saxzn(i,j,2))/dz
  $-1.0/24.0*d_{i,j,2}$
! end do
! end do

! do k=4,nz-2
  b(k)=-1.0/24.0
  a(k)=11.0/12.0
  c(k)=-1.0/24.0
  do j=2,ny
    do i=1,nx
      d(i,j,k)=(saxzn(i,j,k)-saxzn(i,j,k-1))/dz 
      $-1.0/24.0*d_{i,j,2}$
    end do
  end do
end do

! do i=1,nx
! do j=2,ny
  d(i,j,nz-1)=(saxzn(i,j,nz-1)-saxzn(i,j,nz-2))/dz
  $-1.0/24.0*d_{i,j,2}$
! end do
! end do

beta(nz)=0.0
a(nz)=11.0/12.0
c(nz)=0.0

! do i=1,nx
! do j=2,ny
  d(i,j,nz-1)=(saxzn(i,j,nz-1)-saxzn(i,j,nz-2))/dz
  $-1.0/24.0*d_{i,j,2}$
! end do
! end do

beta(nz)=0.0
a(nz)=11.0/12.0
c(nz)=0.0

! do m=3,nz-1
! k=nz-m+2
  beta(k)=b(k)/(a(k)-c(k)*beta(k+1))
  do j=2,ny
    do i=1,nx
      gama(i,j,k)=(d(i,j,k)+c(k)*gama(i,j,k+1))/(a(k)-c(k)*beta(k+1))
      $-c(k)*beta(k+1)$
    end do
  end do
  do i=1,nx
! do j=2,ny
  do k=3,nz-1
    u7(i,j,k)=beta(k)*u7(i,j,k-1)+gama(i,j,k)
    d(i,j,k)=u7(i,j,k)
! end do
! end do
! end do

C Calculate derivative of stress $d_{i,j,zy}$

```
do i=2,nx
    d(i,ny-1,k)=(sayzn(i,ny-1,k)-sayzn(i,ny-2,k))/dy
    $ -1.0/24.0\cdot \text{difzyz}(i,ny,k)$
end do
end do

beta(ny)=0
do k=1,nz
    do i=2,nx
        beta(i,k)=b(j)/(a(j)-c(j)*beta(j+1))
        do j=1,ny
            do k=1,nz
                gama(i,j,k)=(d(i,j,k)+c(j)*gama(i,j,k+1))/(a(j)-c(j)*beta(j+1))
            end do
        end do
    end do
end do

C Calculate derivative of stress \text{difyz}

$ \text{C}(\text{s}\text{i}\text{f}\text{y}\text{z})$

do i=2,nx
    do j=1,ny
        do k=1,nz
            u8(i,j,k)=0.0
        end do
    end do
end do

do m=3,ny-1
    j=ny-m+2
    beta(j)=b(j)/(a(j)-c(j)*beta(j+1))
    do i=2,nx
        do k=1,nz
            gama(i,j,k)=(d(i,j,k)+c(j)*gama(i,j,k+1))/(a(j)-c(j)*beta(j+1))
        end do
    end do
end do

b(3)=0.0
a(3)=11.0/12.0
c(3)=-1.0/24.0
do i=1,nx
  a(i)=0
  b(i)=0
  c(i)=0
  beta(i)=0
end do

do j=1,ny
  do k=1,nz
    gamma(i,j,k)=0.0
    d(i,j,k)=0.0
  end do
end do
end do

C Calculate velocity
Call velocity(nx,ny,nz,dx,dy,dz,dt,TEo,TEold,saxo,$sayo,sazo,saxyo,sayyo,$saxn,sayn,sazn,saxyn,sayzn,vxo,$vyo,vzo,vxn,vyn,vzn,$uxo,uyo,uzo,uxn,uyu,uzn,difx,dify,$difz,difxyy,difxz2,difyx,x,difzyz,difzyx)
C End the current time step
C

do k=l,nz+l
  do j=l,ny+l
    do i=l,nx+l
      TEo(i,j,k)=TEold(i,j,k)
      TLo(i,j,k)=TLold(i,j,k)
      epxo(i,j,k)=epxn(i,j,k)
      epyo(i,j,k)=epyn(i,j,k)
      epzo(i,j,k)=epzn(i,j,k)
      epxyo(i,j,k)=epxyn(i,j,k)
      epxzo(i,j,k)=epxzn(i,j,k)
      epyzo(i,j,k)=epyzn(i,j,k)
      saxo(i,j,k)=saxn(i,j,k)
      sayo(i,j,k)=sayn(i,j,k)
      sazo(i,j,k)=sazn(i,j,k)
      saxyo(i,j,k)=saxyn(i,j,k)
      saxzo(i,j,k)=saxzn(i,j,k)
      sayzo(i,j,k)=sayzn(i,j,k)
      vxo(i,j,k)=vxn(i,j,k)
      vyo(i,j,k)=vyn(i,j,k)
      vzo(i,j,k)=vzn(i,j,k)
      uxo(i,j,k)=uxn(i,j,k)
     uyo(i,j,k)=uyn(i,j,k)
      uzo(i,j,k)=uzn(i,j,k)
    end do
  end do
end do

if (big.lt.(TEold(l,1,1,l)-300.0)) then
  big=TEold(l,1,1,l)-300.0
end if

TEm(n)=TEold(l,1,1,l)
TLm(n)=TLold(l,1,1,l)
u1m(n)=vxn(l,1,2)
u2m(n)=vyn(l,1,2)
u3m(n)=vzn(l,1,1)
v1m(n)=vxn(l,1,2)
v2m(n)=vyn(l,1,2)
v3m(n)=vzn(l,1,1)
icounter=icounter+1
write(*,*) icounter
write(8,1020) t(n),TEm(n),TLm(n)
write(7,1020) t(n),u1m(n),u2m(n),u3m(n)

C Output intermediate result

if (n.eq.50) then
  C The result at time t=0.25ps
  C Electron temp
  open(unit=11,file="te025ps.txt")
do k=1,nz+l
  write(11,1020) TEold(l,1,1,k)
enddo
  C Lattice temp
  open(unit=12,file="ctlxz025ps.txt")
do k=1,nz+l
  write(12,1010) (TLold(i,1,1,k),i=1,nx+l)
enddo
  C The result at time t=0.5ps
  C Electron temp
  open(unit=13,file="te025ps.txt")
do k=1,nz+l
  write(13,1020) TLold(l,1,1,k)
enddo
  C Lattice temp
  open(unit=14,file="ctlxz025ps.txt")
do k=1,nz+l
  write(14,1010) TEold(l,1,1,k)
enddo

if (n.eq.200) then
  C The result at time t=1ps
  C Electron temp
  open(unit=15,file="te1ps.txt")
do k=1,nz+l
  write(15,1020) TEold(l,1,1,k)
enddo
  C Lattice temp
  open(unit=16,file="ctlxz1ps.txt")
do k=1,nz+l
  write(16,1010) TLold(l,1,1,k)
enddo
  C Stress
  open(unit=17,file="saz1ps.txt")
do k=1,nz+l
  write(17,1020) sazn(l,1,1,k)
enddo

C Output

write(8,1020) t(n),TEm(n),TLm(n)
write(7,1020) t(n),u1m(n),u2m(n),u3m(n)

C Output intermediate result

if (n.eq.50) then
  C The result at time t=0.25ps
  C Electron temp
  open(unit=18,file="te025ps.txt")
do k=1,nz+l
  write(18,1010) (TEold(i,1,1,k),i=1,nx+l)
enddo
  C Lattice temp
  open(unit=19,file="ctlxz025ps.txt")
do k=1,nz+l
  write(19,1020) TLold(l,1,1,k)
enddo
  C Stress
  open(unit=20,file="saz1ps.txt")
do k=1,nz+l
  write(20,1020) sazn(l,1,1,k)
enddo

if (n.eq.200) then
  C The result at time t=1ps
  C Electron temp
  open(unit=21,file="te1ps.txt")
do k=1,nz+l
  write(21,1020) TEold(l,1,1,k)
enddo
  C Lattice temp
  open(unit=22,file="ctlxz1ps.txt")
do k=1,nz+l
  write(22,1020) TLold(l,1,1,k)
enddo
if (n.eq.1000) then
C The result at time t=5ps
C Displacement un
open(unit=23,file=’uxnxz5ps.txt’)
do k=2,nz
write(23,1010) (uxn(i,11,k),i=1,nx)
enddo
open(unit=24,file=’uznxz5ps.txt’)
do k=1,nz
write(24,1010) (uzn(i,11,k),i=1,nx)
enddo
open(unit=25,file=’uynyz5ps.txt’)
do k=2,nz
write(25,1010) (uyn(11,j,k),j=1,ny)
enddo
C Stress
open(unit=26,file=’saxxz5ps.txt’)
do k=2,nz
write(26,1010) (saxn(i,11,k),i=1,nx+1)
enddo
open(unit=27,file=’sazxz5ps.txt’)
do k=1,nz+1
write(27,1010) (sazn(i,11,k),i=1,nx+1)
enddo
open(unit=28,file=’sayy5ps.txt’)
do k=2,nz
write(28,1010) (sayn(11,j,k),j=1,ny+1)
enddo
open(unit=29,file=’saz5ps.txt’)
do k=1,nz+1
write(29,1020) sazn(11,11,k)
enddo
end if

if (n.eq.2000) then
C The result at time t=10ps
C Displacement un
open(unit=30,file=’uxnxz10ps.txt’)
do k=1,nz+1
write(30,1010) (uxn(i,11,k),i=1,nx)
enddo
open(unit=31,file=’uznxz10ps.txt’)
do k=1,nz+1
write(31,1010) (uzn(i,11,k),i=1,nx)
enddo
open(unit=32,file=’uynyz10ps.txt’)
do k=2,nz
write(32,1010) (uyn(11,j,k),j=1,ny)
enddo
C Stress
open(unit=33,file=’saxxz10ps.txt’)
do k=2,nz
write(33,1010) (saxn(i,11,k),i=1,nx+1)
enddo
open(unit=34,file=’saxn(xz10ps.txt’)
do k=1,nz+1
write(34,1010) (sayn(11,j,k),j=1,ny+1)
enddo
end if

if (n.eq.3000) then
C The result at time t=15ps
C Displacement un
open(unit=40,file=’uxnxz15ps.txt’)
do k=1,nz+1
write(40,1010) (uxn(i,11,k),i=1,nx)
enddo
open(unit=41,file=’uznxz15ps.txt’)
do k=1,nz+1
write(41,1010) (uzn(i,11,k),i=1,nx)
enddo
open(unit=42,file=’uynyz15ps.txt’)
do k=2,nz
write(42,1010) (uyn(11,j,k),j=1,ny)
enddo
C Stress
open(unit=43,file=’saxxz15ps.txt’)
do k=2,nz
write(43,1010) (saxn(i,11,k),i=1,nx+1)
enddo
open(unit=44,file=’sazxz15ps.txt’)
do k=1,nz+1
write(44,1010) (sazn(i,11,k),i=1,nx+1)
enddo
end if
end if
if (n.eq.3400) then
C The result at time t=17ps
open(unit=58,file='sazl7ps.txt')
do k=1,nz+l
write(58,1020)sazn(ll,ll,k)
enddo
end if
if (n.eq.4000) then
C The result at time t=20ps
open(unit=47,file='ctexz20ps.txt')
dok=l,nz+l
write(47,1010) (TEold(i,l l,k),i=l,nx+l)
enddo
open(unit=48,file='te20ps.txt')
dok=l,nz+l
write(48,1020) TEold(l 1,1 l,k)
enddo
C Lattice temp
open(unit=49,file='ctlxz20ps.txt')
do k=l,nz+l
write(49,1010) (TLold(i,l l,k),i=l,nx+l)
enddo
open(unit=50,file='tl20ps.txt')
dok=l,nz+l
write(50,1020) TLold(l 1,1 l,k)
enddo
C Displacement un
open(unit=51,file='uxnxz20ps.txt')
do k=2,nz
write(51,1010) (uxn(i,ll,k),i=l,nx)
enddo
open(unit=52,file='uxnz20ps.txt')
do k=1,nz
write(52,1010) (uzn(i,l l,k),i=l,nx)
enddo
open(unit=53,file='uynyz20ps.txt')
do k=2,nz
write(53,1010) (uyn(l 1,j,k)j=l,ny)
enddo
C Stress
open(unit=54,file='saxxz20ps.txt')
do k=2,nz
write(54,1010) (saxn(i,l l,k),i=1,nx+1)
enddo
open(unit=55,file='sazxz20ps.txt')
do k=1,nz+1
write(55,1010) (sazn(i,ll,k),i=1,nx+1)
enddo
open(unit=56,file='sayyz20ps.txt')
do k=2,nz
write(56,1010) (sayn(l 1,j,k)j=1,ny+1)
enddo
open(unit=57,file='saz20ps.txt')
do k=1,nz+1
write(57,1020) sazn(11,11,k)
enddo
end if
C Complete the whole period
C The result at time t=17ps
open(unit=58,file='sazl7ps.txt')
do k=1,nz+l
write(58,1020)sazn(ll,ll,k)
enddo
end if
C The result at time t=20ps
open(unit=47,file='ctexz20ps.txt')
dok=l,nz+l
write(47,1010) (TEold(i,l l,k),i=1,nx+1)
enddo
open(unit=48,file='te20ps.txt')
dok=l,nz+l
write(48,1020) TEold(l 1,1 l,k)
enddo
C Lattice temp
open(unit=49,file='ctlxz20ps.txt')
do k=l,nz+l
write(49,1010) (TLold(i,l l,k),i=l,nx+l)
enddo
open(unit=50,file='tl20ps.txt')
dok=l,nz+l
write(50,1020) TLold(l 1,1 l,k)
enddo
C Displacement un
open(unit=51,file='uxnxz20ps.txt')
do k=2,nz
write(51,1010) (uxn(i,ll,k),i=l,nx)
enddo
open(unit=52,file='uxnz20ps.txt')
do k=1,nz
write(52,1010) (uzn(i,l l,k),i=l,nx)
enddo
open(unit=53,file='uynyz20ps.txt')
do k=2,nz
write(53,1010) (uyn(l 1,j,k)j=1,ny)
enddo
C Stress
open(unit=54,file='saxxz20ps.txt')
do k=2,nz
write(54,1010) (saxn(i,l l,k),i=1,nx+1)
enddo
open(unit=55,file='sazxz20ps.txt')
do k=1,nz+1
write(55,1010) (sazn(i,ll,k),i=1,nx+1)
enddo
open(unit=56,file='sayyz20ps.txt')
do k=2,nz
write(56,1010) (sayn(l 1,j,k)j=1,ny+1)
enddo
open(unit=57,file='saz20ps.txt')
do k=1,nz+1
write(57,1020) sazn(11,11,k)
enddo
end if
C Complete the whole period
1 end do
print *, big
open(unit=59,file='Te10(x,y=0).dat')
do k=1,nz+1
write(59,1010) (z(k)*1.0D+6), TEold(11,11,k)
enddo
open(unit=60,file='Tl10(x,y=0).dat')
do k=1,nz+1
write(60,1010) (z(k)*1.0D+6), TLold(11,11,k)
enddo
open(unit=61,file='Tem224.dat')
do n=1,nt
write(61,1020) ((TEm(n)*1.0D+12),((TEm(n)*300.0)/big)
enddo
open(unit=62,file='um224.dat')
do n=1,nt
write(62,1020) ((TEm(n)*1.0D+12),(u3m(n)*1.0D+9)
enddo
print *, "zonezsel"
do k=1,nz+1
print *, (z(k)*1.0D+6), (sazn(11,11,k)*1.0D-9)
enddo
1010 format(401el5.6)
1020 format(15.6,3e15.6)
end
C End main program
C Subroutines
C Calculate temperature
Subroutine
temp(nx,ny,nz,dx,dy,dz,x,y,z,t,dt,TLo,TLold,TEo,TEold,
$ epxn,epyn,epzn,epxo,epyo,epzo)
implicit double precision (a-h,l-o-z)
dimension x(51),y(51),z(221)
dimension TLo(41,41,101),TElo(41,41,101),
$ TL(41,41,101),TLold(41,41,101),
$ TEmew(41,41,101),TEmnew(41,41,101),
$ epxn(41,41,101),epyn(41,41,101),epzn(41,41,101),
$ epzo(41,41,101),epyo(41,41,101),epzo(41,41,101),
$ dTE(41,41,101),dTL(41,41,101)
integer iteration,flagE,flagL
C Data
C Lame constant
clemta=199.0d+9
C Shear modulus
cmiu=27.0d+9
C Thermal expansion coefficient
alpha=14.2d-6
C Electron heat capacity
ceo=2.1d+4
C Lattice heat capacity
cl=2.5d+6
C Electron - lattice coupling factor
g=2.6d+16
C Electron thermal conductivity
cke0=3.15d+2
C Laser fluence
flu=500.0
C Laser pulse duration
tp=0.1d-12
C Optical penetration depth
delta=15.3d-9
C Surface reflectivity
sur=0.93
C Spatial profile parameters
zs=1.0d-6
iteration=0

rx=dt/(4.0*dx*dx)
ry=dt/(4.0*dy*dy)
rz=dt/(4.0*d2*dz)
d0=g*dt/(2.0*cl)
deterror=1.0d-3
ee=(3.0*clemta+2.0*cmiu)*alpha*300.0/cl

C Iteration starts
C flagE and flagL indicate whether TE and TL are
precise enough
C keep on iterating as long as flagE or flagL equals to 1
2 do k=2,nz
   doj=2,ny
   do i=2,nx
   C Heat source
   aa=-z(k)/delta-(x(i)-10.0*dx)*(x(i)-10.0*dx)
   $ +y(j)-10.0*dy)*(y(j)-10.0*dy)/(zs*zs)
   $ -2.77*(t-2.0*tp)*(t-2.0*tp)/(tp*tp)
   q=0.94*flu*(1.0-sur)*exp(aa)/(tp*delta)
   $ -2.77*(t-2.0*tp)*(t-2.0*tp)/(tp*tp)
   a0=ce0*(TEo(ij,k)+TEold(ij,k))/(2.0*300.0)
   b1=cke0*(TEold(i+1,j,k)+TLo(i+1,j,k))
   $ +TTeold(i,j,k)*rx
   b2=cke0*(TEold(i,j,k)+TLo(i,j,k))
   $ +TTeold(i+1,j,k)*rx
   b3=cke0*(TEold(i,j+1,k)+TLo(i,j+1,k))
   $ +TTeold(i,j,k)*ry
   b4=cke0*(TEold(i,j,k)+TLo(i,j,k))
   $ +TTeold(i+1,j,k)*ry
   b5=cke0*(TEold(i,j,k)+TLo(i,j,k))
   $ +TTeold(i,j,k)*rz
   b6=cke0*(TEold(i,j,k)+TLo(i,j,k))
   $ +TTeold(i,j,k)*rz
   C Boundary Conditions
C Test for convergence

detmax=0.0

do i=2,nx
  do j=2,ny
    do k=2,nz
      det1=abs(TEnew(i,j,k)-TEold(i,j,k))
      if (det1.gt.detmax) detmax=det1
      det2=abs(TLnew(i,j,k)-TLold(i,j,k))
      if (det2.gt.detmax) detmax=det2
    end do
  end do
end do
if (detmax.le.deterror) goto 3

C Update all the TEold, TLold with TEnew and TLnew

uxn(i,j,k)=(theta*vxn(i,j,k)+(1.0-theta)*vxo(i,j,k))*dt+uxo(i,j,k)
end do
end do
end do

C Density

lou=1.93d+4
C Electron - blast coefficient
tri=70
theta=0.5
$ +\tri^*(1.0-\theta)*(T_Eo(i,j,k+1)*T_Eo(i,j,k+1)-T_Eo(i,j,k))$ 

$ *T_Eo(i,j,k)/(dz)\times dt/lou+vzo(i,j,k)$

end do

return

uzn(i,j,k)=(theta*vzn(i,j,k)+(1.0

$ -\theta)*vzo(i,j,k)$

end do

end do

end do
APPENDIX B

SOURCE CODE FOR 3D DOUBLE-LAYERED CASE
C Main program
C Set all variables

implicit double precision (a-h,l,o-z)
dimension t(4001),tl(4001),x(51),y(51),z(221)
dimension TEo(41,41,101),TEold(41,41,101),
$ TLo(41,41,101),TLold(41,41,101),
$ TEm(4001),TLm(4001),
$ ulm(4001),u2m(4001),u3m(4001),
$ vlm(4001),v2m(4001),v3m(4001),

C Ex, Ey, Ez normal strain and shear strain
$ epxo(41,41,101),epyo(41,41,101),epzo(41,41,101),
$ epxyo(41,41,101),epzxo(41,41,101),
$ ssaoxy(41,41,101),ssaoxz(41,41,101),
$ ssaoyz(41,41,101),
$ epxn(41,41,101),epyyn(41,41,101),epzn(41,41,101),
$ epxyn(41,41,101),epxzn(41,41,101),epyzn(41,41,101),

C Normal stress and shear stress
$ saxo(41,41,101),sayo(41,41,101),sazo(41,41,101),
$ saxyo(41,41,101),saxzo(41,41,101),sayzo(41,41,101),
$ saxn(41,41,101),sayn(41,41,101),sazn(41,41,101),
$ saxyn(41,41,101),saxzn(41,41,101),sayzn(41,41,101),

C Velocity and displacement
$ vxo(41,41,101),vyo(41,41,101),vzo(41,41,101),
$ vxo(41,41,101),vyn(41,41,101),vzn(41,41,101),
$ ux(41,41,101),uyo(41,41,101),uzo(41,41,101),
$ uxo(41,41,101),uy(41,41,101),uz(41,41,101),

C Stress derivative
$ d(41,41,101),c(221),gama(221),
$ d(41,41,101),
$ difx(41,41,101),dify(41,41,101),difz(41,41,101),
$ difxy(41,41,101),difxxy(41,41,101),
$ difxx(41,41,101),
$ difzz(41,41,101),difzy(41,41,101),
$ difzyz(41,41,101),

C Additional set
$ u(41,41,101),u2(41,41,101),u3(41,41,101),
$ u(41,41,101),
$ u(41,41,101),u6(41,41,101),
$ u7(41,41,101),u8(41,41,101),
$ u9(41,41,101),

C Data
C Lame constant
clement=199.0d+9
clement2=83.3d+9
C Shear modulus
cmiu1=27.0d+9
cmiu2=115.0d+9
C Thermal expansion coefficient
alphei=14.2d-6
alphei2=4.9d-6
open(unit=8, file="etm.txt")
open(unit=7, file="um.txt")
C dimension
bx=1.0D-4
ly=1.0D-4
lz=1.0D-7
difxzx(i,j,k)=0.0
difxzz(i,j,k)=0.0
difyzy(i,j,k)=0.0
difyzz(i,j,k)=0.0
enddo
enddo
enddo

doi=l,nx
doj=l,ny+l
dok=l,nz+l
uxo(ij,k)=0.0
vxo(ij,k)=0.0
enddo
enddo
enddo

doi=l,nx+l
doj=l,ny
dok=l,nz+l
uyo(ij,k)=0.0
vyo(ij,k)=0.0
enddo
enddo
enddo

doi=l,nx+l
doj=l,ny+1
dok=l,nz+1
uzo(ij,k)=0.0
vzo(ij,k)=0.0
enddo
enddo
enddo

doi=l,nx
doj=l,ny+1
dok=l,nz+1
uzo(ij,k)=0.0
vzo(ij,k)=0.0
enddo
enddo
enddo

doi=l,nx
doj=l,ny
sk(ij,k)=0.0
szy(ij,k)=0.0
enddo
enddo
enddo

doi=l,nx
doj=2,ny
sk(ij,k)=0.0
szy(ij,k)=0.0
enddo
enddo
enddo

doi=2,nx
doj=1,ny
sk(ij,k)=0.0
szy(ij,k)=0.0
enddo
enddo
enddo

n=1

TEM(n)=0.0
TLm(n)=0.0

big=0.0

write(*,*) 'start'
doi=1,nx
doj=1,ny
sk(ij,k)=0.0
szy(ij,k)=0.0
enddo
enddo
enddo

do i=1,nx

do 1 n=1,nt

t(n)=nt

t1(n)=(n-1)*dt+dt/2.0

c guess normal and shear strain Ex, Ey, Ez, Exy,
syz, Exz values

do i=1,nx+1
do j=1,ny+1
do k=1,nz+1
epxn(i,j,k)=epxo(i,j,k)
epyn(i,j,k)=epyo(i,j,k)
epzn(i,j,k)=epzo(i,j,k)
epxyo(i,j,k)=epxyo(i,j,k)
epyzo(i,j,k)=epyzo(i,j,k)
epxzo(i,j,k)=epxzo(i,j,k)
enddo
enddo
enddo

c iteration

tol=1d-17
detuvmax=tol+1d-5
do while (detuvmax.gt.tol)
detuvmax=0.0
detlmax=0.0
det2max=0.0
det3max=0.0
det4max=0.0
det5max=0.0
det6max=0.0
c call subroutine calculate temperature

c compute normal stress

do j=1,ny+1
do k=1,nz+1
sxn(i,j,k)=0.0
saxn(i,j,k)=0.0
enddo
enddo
enddo

C Call temp(nx,ny,nz,nz2,dx,dy,dz,x,y,z,tl(n),dt,TLo,
TLold,TEo,TEold,exy,exz,t1(n),dt,TLo,
$ TLold,TEo,TEold,epxo,epyo,epzo)
c compute normal stress

do j=1,ny+1
do k=1,nz+1
sxn(i,j,k)=0.0
saxn(i,j,k)=0.0
enddo
enddo
enddo

end
do j=1,ny+1
do i=1,nx+1
sazn(i,j,1)=0.0
sazn(i,j,nz+1)=0.0
end do
end do

end do
end do
do i=2,nx
do j=2,ny

C gold

clemta=clemta1
cmiu=cmiu1
alpha=alpha1

do k=2,nz2-1
saxn(i,j,k)=(clemta+2.0*cmiu)*epxn(i,j,k)
$+clemta*epyn(i,j,k)
$+clemta*epzn(i,j,k)
$-(3.0*clemta+2.0*cmiu)*alpha*(TLold(i,j,k)-300.0)
sayn(i,j,k)=clemta*epxn(i,j,k)
$+(clemta+2.0*cmiu)*epyn(i,j,k)
$+clemta*epzn(i,j,k)
$-(3.0*clemta+2.0*cmiu)*alpha*(TLold(i,j,k)-300.0)
sazn(i,j,k)=clemta*epxn(i,j,k)
$+(clemta+2.0*cmiu)*epzn(i,j,k)
$+clemta*epyn(i,j,k)
$-(3.0*clemta+2.0*cmiu)*alpha*(TLold(i,j,k)-300.0)
end do

sayn(i,j,k)=clemta*epxn(i,j,k)
$+(clemta+2.0*cmiu)*epyn(i,j,k)
$+clemta*epzn(i,j,k)
$-(3.0*clemta+2.0*cmiu)*alpha*(TLold(i,j,k)-300.0)
sayn(i,j,k)=clemta*epzn(i,j,k)
$+clemta*epyn(i,j,k)
$-(3.0*clemta+2.0*cmiu)*alpha*(TLold(i,j,k)-300.0)
end do

C Chromium

clemta=clemta2
cmiu=cmiu2
alpha=alpha2

do k=2,nz2+1,nz
saxn(i,j,k)=(clemta+2.0*cmiu)*epxn(i,j,k)
$+clemta*epyn(i,j,k)
$+clemta*epzn(i,j,k)
$-(3.0*clemta+2.0*cmiu)*alpha*(TLold(i,j,k)-300.0)
sayn(i,j,k)=clemta*epxn(i,j,k)
$+(clemta+2.0*cmiu)*epyn(i,j,k)
$+clemta*epzn(i,j,k)
$-(3.0*clemta+2.0*cmiu)*alpha*(TLold(i,j,k)-300.0)
sazn(i,j,k)=clemta*epxn(i,j,k)
$+(clemta+2.0*cmiu)*epzn(i,j,k)
$+clemta*epyn(i,j,k)
$-(3.0*clemta+2.0*cmiu)*alpha*(TLold(i,j,k)-300.0)
end do

end do
end do

k=nz2
saxn(i,j,k)=(saxn(i,j,k+1)+saxn(i,j,k-1))/2
sayn(i,j,k)=(sayn(i,j,k+1)+sayn(i,j,k-1))/2
sazn(i,j,k)=(sazn(i,j,k+1)+sazn(i,j,k-1))/2
end do
end do

C Calculate shear stress

saxzn(i,j,1)=0.0
saxzn(i,j,nz)=0.0
end do
end do

end do
end do
do i=1,ny
do k=2,nz
saxyn(i,j,k)=0.0
saxyn(nx,j,k)=0.0
end do
end do
do i=1,ny
do k=2,nz
saxyn(i,1,k)=0.0
saxyn(i,ny,k)=0.0
end do
end do

end do
end do
end do
do i=2,nx
do k=1,nz
sayzn(i,1,k)=0.0
sayzn(i,ny,k)=0.0
end do
end do
end do
do i=2,nx
do j=1,ny
sayzn(i,j,1)=0.0
sayzn(i,j,nz)=0.0
end do
end do
do j=2,ny-1
do i=2,nx
clenta=clenta1
cmiu=cmiu1
alpha=alpha1
do k=2,nz-1
sayzn(i,j,k)=cmiu*sayzn(i,j,k)
end do
cleta=cleta2
cmiu=cmiu2
alpha=alpha2
do k=iesz,2-1
sayzn(i,j,k)=cmiu*sayzn(i,j,k)
end do
end do
end do
end do
end do
end do
end do

C Calculate derivative of stress difx

do k=2,nz
do j=2,ny
difx(1,j,k)=(saxn(2,j,k)-saxn(1,j,k))/dx
difx(nx,j,k)=(saxn(nx+1,j,k)-saxn(nx,j,k))/dx
end do
end do

b(2)=0.0
a(2)=1.0/12.0
c(2)=1.0/24.0

end do
end do
end do
do j=2,ny-1
do i=2,nx-1

d(2,j,k)=(saxn(3,j,k)-saxn(2,j,k))/dx
$\ -1.0/24.0*\text{difx}(1,j,k)$
end do
end do
end do

C Calculate derivative of stress dify

do i=1,nx
a(i)=0
b(i)=0
c(i)=0
end do
end do
end do
do j=1,ny
d(i+1,j,k)=beta(i)*ul(i,j,k)+gama(i,j,k)
difx(i,j,k)=ul(i,j,k)
$\ -1.0/24.0*\text{difx}(1,j,k)$
end do
end do
end do
do k=2,nz
do i=2,nx
dify(i,1,k)=(sayn(i,2,k)-sayn(i,1,k))/dy
dify(i,ny,k)=(sayn(i,ny+1,k)-sayn(i,ny,k))/dy
end do
do i=2,nx
end do
end do

b(2)=0.0
a(2)=11.0/12.0
c(2)=1.0/24.0

do k=2,nz
do i=2,nx
d(i,2,k)=(sayn(i,3,k)-sayn(i,2,k))/dy
$ -1.0/24.0*dify(i,1,k)
end do
do i=2,nx
end do
end do

b(nx)=0.0
a(nx)=11.0/12.0
c(nx)=0.0

do i=2,nx
do j=1,ny-2
b(j)=-1.0/24.0
a(j)=11.0/12.0
c(j)=-1.0/24.0
do i=2,nx
do j=1,ny-2
d(k)=b(k)
do i=2,nx
end do

C Calculate derivative of stress difz

do i=2,nx
do j=2,ny
diffz(i,j,1)=(sayn(i,j,2)-sayn(i,j,1))/dz
diffz(i,j,nz)=(sayn(i,j,nz+1)-sayn(i,j,nz))/dz
end do
end do

b(2)=0.0
a(2)=11.0/12.0
c(2)=1.0/24.0

do i=2,nx
do j=2,ny
d(i,j,2)=(sayn(i,j,3)-sayn(i,j,2))/dz
$ -1.0/24.0*dify(i,j,1)
end do
do i=2,nx
end do
end do

beta(ny)=0.0
do i=2,nx
end do
end do

beta(1)=0.0

do j=2,ny

d(i,j,1)=(sayn(i,j+1)-sayn(i,j))/dz
$ -1.0/24.0*dify(i,j,1)
end do
do i=2,nx
end do
end do

b(3,ny-2)=0.0
a(3)=11.0/12.0
c(3)=-1.0/24.0

do i=2,nx
do j=2,ny
d(i,j,2)=(sayn(i,j,3)-sayn(i,j,2))/dz
$ -1.0/24.0*dify(i,j,1)
end do
do i=2,nx
end do
end do

beta(m)=0.0

do j=2,ny
end do
end do

beta(2)=0.0

do i=2,ny
end do
end do

C Calculate derivative of stress difz

b(nz)=1.0/24.0
a(nz)=11.0/12.0
c(nz)=0.0

do i=2,ny
do j=2,ny

d(i,j,2)=(sayn(i,j,3)-sayn(i,j,2))/dz
$ -1.0/24.0*dify(i,j,1)
end do
do i=2,ny
end do
end do

b(nz-1)=1.0/24.0
a(nz-1)=11.0/12.0
c(nz-1)=0.0

do i=2,ny
do j=2,ny
end do
end do
end do

b(2)=0.0
a(2)=11.0/12.0
c(2)=-1.0/24.0

do i=2,ny
do j=2,ny
d(i,j,nz-1)=(sayn(i,j,nz)-sayn(i,j,nz-1))/dz
end do
end do
end do

b(1)=0.0
a(1)=11.0/12.0
c(1)=0.0

do i=2,ny
do j=2,ny

d(i,j,1)=(sayn(i,j,2)-sayn(i,j,1))/dz
$ -1.0/24.0*dify(i,j,1)
end do
do i=2,ny
end do
end do
end do

b(2)=0.0
a(2)=11.0/12.0
c(2)=-1.0/24.0

do i=2,ny
do j=2,ny
end do
end do
end do

b(1)=0.0
a(1)=11.0/12.0
c(1)=0.0

do i=2,ny
do j=2,ny
end do
end do
end do
DO k=2,nz
DO j=1,ny
DIFZ(i,j,nz)=DIFZ(i,j,nz) - 1.0/24.0*GAMA(i,j,nz)
END DO
END DO

BETA(nz)=0.0
DO i=2,nx
DO j=2,ny
GAMA(i,j,nz)=0.0
END DO
END DO

DO m=2,nz-1
K=nz-m+1
BETA(K)=(1.0/24.0)*DIFZ(i,j,nz)
DO j=2,ny
DO i=2,nx
GAMA(i,j,k)=(DIFZ(i,j,k)+C(K)*GAMA(i,j,k+1))/A(K)
END DO
END DO

DO i=2,nx
DO j=2,ny
U3(i,j,1)=0.0
END DO
END DO

DO i=2,nx
DO j=2,ny
DO k=2,nz-1
U3(i,j,k)=BETA(K)*U3(i,j,k-1)+GAMA(i,j,k)
END DO
END DO

DO i=1,nx
A(I)=0
B(I)=0
C(I)=0
BETA(I)=0
DO J=1,ny
DO K=2,nz
GAMA(I,J,K)=0.0
D(I,J,K)=0.0
END DO
END DO

C Calculate derivative of stress Diffyx

DO K=2, nz
DO J=1, ny
DIFXYX(2,K,J)=(SAXYN(2,K,J)-SAXYN(1,K,J))/DX
DIFXYX(NX,K,J)=(SAXYN(NX,K,J)-SAXYN(NX-1,K,J))/DX
END DO
END DO

B(3)=0.0
A(3)=11.0/12.0
C(3)=-1.0/24.0
do i=1,nx 
a(i)=0 
b(i)=0 
c(i)=0 
end do 
end do 

C Calculate derivative of stress difxyy 

do k=2,nz 
do i=1,nx 
difxyy(i,2,k)=(saxyn(i,2,k)-saxyn(i,1,k))/dy 
difxyy(i,ny,k)=(saxyn(i,ny,k)-saxyn(i,ny-1,k))/dy 
end do 
end do 

$ -1.0/24.0 \cdot \text{difxyy}(i,2,k) 
end do 
end do 

C Calculate derivative of stress difxzx 

do k=1,nz 
do j=2,ny 
difxzx(2,j,k)=(saxzn(2,j,k)-saxzn(1,j,k))/dx 
difxzx(nx,j,k)=(saxzn(nx,j,k)-saxzn(nx-1,j,k))/dx 
end do 
end do 

$ -1.0/24.0 \cdot \text{difxzx}(2,j,k) 
end do 
end do 

c(i)=-1.0/24.0
do j=2,ny
  do k=l,nz
    d(i,j,k)=(saxzn(i,j,k)-saxzn(i-1,j,k))/dx
  end do
end do
end do
end do
b(nx-1)=-1.0/24.0
a(nx-1)=11.0/12.0
c(nx-1)=0.0

do k=1,nz
  do j=2,ny
    d(nx-1,j,k)=(saxzn(nx-1,j,k)-saxzn(nx-2,j,k))/dx
  end do
end do
end do

beta(nx)=0.0

do k=1,nz
  do j=2,ny
    gama(nx,j,k)=0.0
  end do
end do

do m=3,nx-1
  i=nx-m+2
  beta(i)=b(i)/(a(i)-c(i)*beta(i+1))
  do j=2,ny
    do k=1,nz
      gama(i,j,k)=(d(i,j,k)+c(i)*gama(i+1,j,k))/(a(i)
        -c(i)*beta(i+1))
    end do
  end do
end do

beta(nz)=0.0

do k=1,nz
  do j=2,ny
    gama(nz,j,k)=0.0
  end do
end do

C Calculate derivative of stress difxzz

do i=1,nx
  do j=2,ny
    difxzz(i,j,2)=(saxzn(i,j,2)-saxzn(i,j,1))/dz
    difxzz(i,j,nz)=(saxzn(i,j,nz)-saxzn(i,j,nz-1))/dz
  end do
end do
end do

b(3)=0.0
a(3)=11.0/12.0
c(3)=-1.0/24.0

do i=1,nx
  do j=2,ny
    d(i,j,3)=(saxzn(i,j,3)-saxzn(i,j,2))/dz
  end do
end do

b(nz-1)=-1.0/24.0
a(nz-1)=11.0/12.0
c(nz-1)=0.0

do i=1,nx
  do j=2,ny
    d(i,nz-1)=(saxzn(i,nz-1)-saxzn(i,nz-2))/dz
  end do
end do

beta(nz)=0.0

do i=1,nx
  do j=2,ny
    gama(i,nz)=0.0
  end do
end do

do m=3,nz-1
  k=nz-m+2
  beta(k)=b(k)/(a(k)-c(k)*beta(k+1))
  do j=2,ny
    do i=1,nx
      gama(i,j,k)=(d(i,j,k)+c(i)*gama(i,j,k+1))/(a(k)
        -c(k)*beta(k+1))
    end do
  end do
end do
do i=1,nx
do j=2,ny
u7(ij,2)=0.0
end do
end do
do i=1,nx
do j=2,ny
do k=3,nz-1
u7(ij,k)=beta(k)*u7(ij,k-1)+gama(ij,k)
difyxz(i,j,k)=u7(ij,j,k)
end do
end do
end do

do i=1,nx
a(i)=0
b(i)=0
c(i)=0
beta(i)=0
do j=1,ny
do k=1,nz
gama(ij,k)=0.0
d(i,j,k)=0.0
end do
end do
end do

C Calculate derivative of stress difyzy

do k=1,nz
do i=1,ny
difyzy(i,j,2)=(sayzn(i,j,2)-sayzn(i,j,1))/dy
difyzy(i,j,ny)=(sayzn(i,j,ny)-sayzn(i,j,ny-1))/dy
end do
end do

b(3)=0.0
a(3)=11.0/12.0
c(3)=-1.0/24.0

end do

C Calculate derivative of stress difyzz

do k=1,nz
do i=2,ny
difyzz(i,j,2)=(sayzn(i,j,2)-sayzn(i,j,1))/dz
difyzz(i,j,nz)=(sayzn(i,j,nz)-sayzn(i,j,nz-1))/dz
end do
end do
\( b(3) = 0.0 \)
\( a(3) = 11.0/12.0 \)
\( c(3) = -1.0/24.0 \)

\[
\begin{align*}
&\text{do } i=2, nx \\text{ end do} \\
&\text{do } j=1, ny \\text{ end do} \\
&d(i,j,3) = (\text{sayzn}(i,j,3) - \text{sayzn}(i,j,2))/dz \\
&\text{end do} \\
\end{align*}
\]

\[
\begin{align*}
&\text{do } i=2, nx \\
&\text{do } j=1, ny \\
&d(i,j,3) = \frac{\text{sayzn}(i,j,3) - \text{sayzn}(i,j,2)}{dz} - \frac{1.0}{24.0}\text{difyzz}(i,j,2) \\
&\text{end do} \\
&\text{end do} \\
\end{align*}
\]

\[
\begin{align*}
&\text{do } k=4, nz-2 \\
&b(k) = -1.0/24.0 \\
&a(k) = 11.0/12.0 \\
&c(k) = -1.0/24.0 \\
&\text{do } j=1, ny \\
&\text{do } i=2, nx \\
&d(i,j,k) = (\text{sayzn}(i,j,k) - \text{sayzn}(i,j,k-1))/dz \\
&\text{end do} \\
&\text{end do} \\
\end{align*}
\]

\[
\begin{align*}
&\text{do } i=2, nx \\
&\text{do } j=1, ny \\
&d(i,j,nz-1) = (\text{sayzn}(i,j,nz-1) - \text{sayzn}(i,j,nz-2))/dz \\
&\text{end do} \\
&\text{end do} \\
\end{align*}
\]

\[
\begin{align*}
&b(nz-1) = -1.0/24.0 \\
&a(nz-1) = 11.0/12.0 \\
&c(nz-1) = 0.0 \\
&\text{do } i=2, nx \\
&\text{do } j=1, ny \\
&d(i,j,nz-1) = \frac{\text{sayzn}(i,j,nz-1) - \text{sayzn}(i,j,nz-2)}{dz} \\
&\text{end do} \\
&\text{end do} \\
\end{align*}
\]

\[
\begin{align*}
&\text{do } k=4, nz-2 \\
&b(k) = -1.0/24.0 \\
&a(k) = 11.0/12.0 \\
&c(k) = -1.0/24.0 \\
&\text{do } j=1, ny \\
&\text{do } i=2, nx \\
&d(i,j,k) = \frac{\text{sayzn}(i,j,k) - \text{sayzn}(i,j,k-1)}{dz} - \frac{1.0}{24.0}\text{difyzz}(i,j,k) \\
&\text{end do} \\
&\text{end do} \\
\end{align*}
\]

\[
\begin{align*}
&\beta(nz) = 0.0 \\
&\text{do } i=2, nx \\
&\text{do } j=1, ny \\
&\text{do } k=3, nz-1 \\
&\text{do } k=k, nz \\
&\text{do } j=2, ny \\
&\text{do } i=2, nx \\
&u9(i,j,2) = 0.0 \\
&\text{end do} \\
&\text{end do} \\
\end{align*}
\]

\[
\begin{align*}
&\text{C Calculate velocity} \\
&\text{call velocity(nx, ny, nz2, dx, dy, dz, dt, TEo, TEold,} \\
&\text{saxo, sayo, saxyo, saxzo, sayzo,} \\
&\text{sazn, sayzn, saxzn, saxz, sayzn, vxo,} \\
&\text{vyo, vzo, vyn, vzn,} \\
&\text{uxo, uyo, uzoo, uxn, uyn, uzn, difx, dify,} \\
&\text{diz, difxxy, difxzz, difxyx, difyyx, difxzy, difzyx)} \\
\end{align*}
\]

\[
\begin{align*}
&\text{C Calculate strain} \\
&\text{do } k=2, nz \\
&\text{do } j=2, ny \\
&\text{do } i=2, nx \\
&\text{epxn}(i,j,k) = (\text{theta}*(\text{vxn}(i,j,k)-\text{vxn}(i-1,j,k)) \\
&\text{+} (1.0-\text{theta})*(\text{vxo}(i,j,k)-\text{vxo}(i-1,j,k))) \text{dt/dx} \\
&\text{+} \text{epxo}(i,j,k) \\
&\text{end do} \\
&\text{end do} \\
\end{align*}
\]

\[
\begin{align*}
&\text{do } k=2, nz \\
&\text{do } j=2, ny \\
&\text{do } i=2, nx \\
&\text{epyn}(i,j,k) = (\text{theta}*(\text{vyn}(i,j,k)-\text{vyn}(i,j-1,k)) \\
&\text{+} (1.0-\text{theta})*(\text{vyo}(i,j,k)-\text{vyo}(i,j-1,k))) \text{dt/dy} \\
&\text{+} \text{epyo}(i,j,k) \\
&\text{end do} \\
&\text{end do} \\
\end{align*}
\]

\[
\begin{align*}
&\text{do } k=2, nz \\
&\text{do } j=2, ny-1 \\
&\text{do } i=2, nx-1 \\
&\text{epxn}(i,j,k) = (\text{theta}*(\text{vxn}(i,j+1,k)-\text{vxn}(i,j,k)) \\
&\text{+} (1.0-\text{theta})*(\text{vxo}(i,j+1,k)-\text{vxo}(i,j,k))) \text{dt/dx} \\
&\text{+} (\text{theta}*(\text{vxo}(i+1,j,k)-\text{vxo}(i,j,k)) \\
&\text{+} (1.0-\text{theta})*(\text{vxo}(i+1,j,k)-\text{vxo}(i,j,k))) \text{dt/dz} \\
&\text{+} \text{epzo}(i,j,k) \\
&\text{end do} \\
&\text{end do} \\
\end{align*}
\]

\[
\begin{align*}
&\text{C Calculate Shear strain} \\
&\text{do } k=2, nz \\
&\text{do } j=2, ny-1 \\
&\text{do } i=2, nx-1 \\
&\text{epxn}(i,j,k) = (\text{theta}*(\text{vxn}(i,j+1,k)-\text{vxn}(i,j,k)) \\
&\text{+} (1.0-\text{theta})*(\text{vxo}(i,j+1,k)-\text{vxo}(i,j,k))) \text{dt/dx} \\
&\text{+} (\text{theta}*(\text{vxo}(i+1,j,k)-\text{vxo}(i,j,k)) \\
&\text{+} (1.0-\text{theta})*(\text{vxo}(i+1,j,k)-\text{vxo}(i,j,k))) \text{dt/dz} \\
&\text{+} \text{epzo}(i,j,k) \\
&\text{end do} \\
&\text{end do} \\
\end{align*}
\]

\[
\begin{align*}
&\text{do } k=2, nz \\
&\text{do } j=2, ny \\
&\text{do } i=2, nx \\
&\text{epxn}(i,j,k) = (\text{theta}*(\text{vxn}(i,j+1,k)-\text{vxn}(i,j,k)) \\
&\text{+} (1.0-\text{theta})*(\text{vxo}(i,j+1,k)-\text{vxo}(i,j,k))) \text{dt/dx} \\
&\text{+} (\text{theta}*(\text{vxo}(i+1,j,k)-\text{vxo}(i,j,k)) \\
&\text{+} (1.0-\text{theta})*(\text{vxo}(i+1,j,k)-\text{vxo}(i,j,k))) \text{dt/dz} \\
&\text{+} \text{epzo}(i,j,k) \\
&\text{end do} \\
&\text{end do} \\
\end{align*}
\]
epxzn(i,j,k)=(theta*(vxn(i,j,k+1)-vxn(i,j,k))
$ + (1.0-theta)*(vxo(i,j,k+1)-vxo(i,j,k)))*dt/dz
$ +(theta*(vzn(i+lj,k)-vzn(i,j,k))
$ +(1.0-theta)*(vzo(i+lj,k)-vzo(i,j,k)))*dt/dx
$ +epxzo(i,j,k)
end do
end do
end do
dok=2,nz-l
doj=2,ny-l
doi=2,nx
epyzn(ij,k)=(theta*(vyn(ij,k+1)-vyn(ij,k))
$ + (1.0-theta)*(vyo(ij,k+1)-vyo(ij,k)))*dt/dz
$ +(theta*(vzn(ij+l,k)-vzn(ij,k))
$ +(1.0-theta)*(vzo(ij+l,k)-vzo(ij,k)))*dt/dy
$ +epyzo(ij,k)
end do
end do
end do
C Check convergence
do k=1,nz+1
do j=1,ny+1
do i=1,nx+1
det1=epxn(ij,k)+xsao0(ij,k)
det2=epyn(ij,k)+ysao0(ij,k)
det3=epzn(ij,k)+zaso0(ij,k)
det4=epxyo(ij,k)+saxooy(ij,k)
det5=epxzo(ij,k)+saxoz(ij,k)
det6=epyzn(ij,k)+saxozy(ij,k)
det=max(abs(det1),abs(det2),abs(det3),abs(det4),abs(det5),abs(det6))
if (abs(det).lt.(abs(det)+detmax)) then
endif
C End the current time step
C--
enddo
do=1,nx+1
do=1,ny+1
do=1,nx+1
xsao0(ij,k)=epxn(ij,k)
yso0(ij,k)=epyn(ij,k)
zso0(ij,k)=epzn(ij,k)
ssoxy0(ij,k)=epxyo(ij,k)
ssoxz0(ij,k)=saxoxx(ij,k)
ssoy0(ij,k)=epyzn(ij,k)
write(*,*) 'detmax=', detmax
C End do with detmax
C Output
write(8,1020) t(n),TEM(n),TLm(n)
write(7,1020) t(n),u1m(n),u2m(n),u3m(n)
C Output intermediate result
if(n.eq.50)then
C The result at time t=0.25ps
C Electron temp
open(unit=10,file='ctexz025ps.txt')
C Read input
if (big.gt.(big+TEold(11,11,1)-300.0)) then
endif

write(10,1010) (TEold(i,11,k),i=1,nx+1)
enddo
open(unit=11,file='te025ps.txt')
do k=1,nz+1
write(11,1020) TEold(11,11,k)
enddo
C Lattice temp
open(unit=12,file='ctlxz025ps.txt')
do k=1,nz+1
write(12,1010) (TLold(i,11,k),i=1,nx+1)
enddo
open(unit=13,file='tl025ps.txt')
do k=1,nz+1
write(13,1020) TLold(11,11,k)
enddo
if(n.eq.100) then
C The result at time t=0.5ps
C Electron temp
open(unit=14,file='ctexz05ps.txt')
do k=1,nz+1
write(14,1010) (TEold(i,11,k),i=1,nx+1)
enddo
open(unit=15,file='te05ps.txt')
do k=1,nz+1
write(15,1020) TEold(11,11,k)
enddo
C Lattice temp
open(unit=16,file='ctlxz05ps.txt')
do k=1,nz+1
write(16,1010) (TLold(i,11,k),i=1,nx+1)
enddo
open(unit=17,file='tl05ps.txt')
do k=1,nz+1
write(17,1020) TLold(11,11,k)
enddo
if(n.eq.200) then
C The result at time t=1ps
C Electron temp
open(unit=18,file='ctexz1ps.txt')
do k=1,nz+1
write(18,1010) (TEold(i,11,k),i=1,nx+1)
enddo
open(unit=19,file='te1ps.txt')
do k=1,nz+1
write(19,1020) TEold(11,11,k)
enddo
C Lattice temp
open(unit=20,file='ctlxz1ps.txt')
do k=1,nz+1
write(20,1010) (TLold(i,11,k),i=1,nx+1)
enddo
open(unit=21,file='tl1ps.txt')
do k=1,nz+1
write(21,1020) TLold(11,11,k)
enddo
C Stress
open(unit=22,file='saxz1ps.txt')
do k=1,nz+1
write(22,1020) sazn(11,11,k)
enddo
open(unit=76,file='sax1ps.txt')
do k=2,nz
write(76,1020) saxn(11,11,k)
enddo
open(unit=77,file='say1ps.txt')
do k=2,nz
write(77,1020) sayn(11,11,k)
enddo
if(n.eq.1000) then
C The result at time t=5ps
C Displacement un
open(unit=23,file='uxnxz5ps.txt')
do k=2,nz
write(23,1010) (uxn(i,11,k),i=1,nx)
enddo
open(unit=24,file='uznxz5ps.txt')
do k=1,nz
write(24,1010) (uzn(i,11,k),i=1,nx)
enddo
open(unit=25,file='uynyz5ps.txt')
do k=1,nz
write(25,1010) (uyn(j,11,k),j=1,ny)
enddo
open(unit=64,file='uxn5ps.txt')
do k=2,nz
write(64,1020) uxn(11,11,k)
enddo
open(unit=65,file='uyn5ps.txt')
do k=2,nz
write(65,1020) uyn(11,11,k)
enddo
open(unit=66,file='uzn5ps.txt')
do k=1,nz
write(66,1020) uzn(11,11,k)
enddo
C Stress
open(unit=26,file='saxxz5ps.txt')
do k=2,nz
write(26,1010) (saxn(i,11,k),i=1,nx+1)
enddo
open(unit=27,file='sazxz5ps.txt')
do k=1,nz+1
write(27,1010) (sazn(i,11,k),i=1,nx+1)
enddo
open(unit=28,file='sayy5ps.txt')
do k=2,nz
write(28,1010) (sayn(j,11,k),j=1,ny+1)
enddo
open(unit=78,file='sax5ps.txt')
do k=2,nz
write(78,1020) saxn(11,11,k)
enddo
open(unit=79,file='say5ps.txt')
do k=2,nz
write(79,1020)sayn(11,11,k)
enddo
end if

if (n.eq.2000) then
C The result at time t=10ps
open(unit=30, file='ctexz10ps.txt')
dok=l,nz+l
write(30,1010)(TEold(i,11,k),i=l,nx+1)
enddo
open(unit=31, file='te10ps.txt')
dok=l,nz+l
write(31,1020) TEold(l1,11,k)
enddo
open(unit=32, file='ctlx10ps.txt')
dok=l,nz+l
write(32,1010) (TLold(i,11,k),i=l,nx+1)
enddo
open(unit=63, file='tll10ps.txt')
dok=l,nz+l
write(63,1020) TLold(l1,11,k)
enddo
open(unit=67, file='uxn10ps.txt')
dok=l,nz
write(67,1020)uxn(ll,11,k)
enddo
open(unit=68, file='uyn10ps.txt')
dok=l,nz
write(68,1020)uyn(ll,11,k)
enddo
open(unit=69, file='uzn10ps.txt')
dok=l,nz
write(69,1020)uzn(ll,11,k)
enddo
open(unit=36, file='saxx10ps.txt')
dok=l,nz
write(36,1020)saxn(ll,11,k)
enddo
open(unit=37, file='sazx10ps.txt')
dok=l,nz+l
write(37,1020)sazn(ll,11,k)
enddo
open(unit=38, file='sayy10ps.txt')
dok=l,nz+l
write(38,1020)sayn(11,j,k),j=1,ny+1)
enddo
end if

if (n.eq.3000) then
C The result at time t=15ps
C Displacement un
open(unit=40, file='uxnx15ps.txt')
dok=l,nz+1
write(40,1010) (uxn(i,11,k),i=1,nx)
enddo
open(unit=41, file='uznx15ps.txt')
dok=l,nz+1
write(41,1010) (uzn(i,11,k),i=1,nx)
enddo
open(unit=42, file='uyn15ps.txt')
dok=l,nz+1
write(42,1010) (uyn(l1,j,k),j=1,ny)
enddo
open(unit=70, file='uxn15ps.txt')
dok=l,nz+1
write(70,1020)uxn(ll,11,k)
enddo
open(unit=71, file='uyn15ps.txt')
dok=l,nz+1
write(71,1020)uyn(ll,11,k)
enddo
open(unit=72, file='uzn15ps.txt')
dok=l,nz+1
write(72,1020)uzn(ll,11,k)
enddo
C Stress
open(unit=43, file='saxx15ps.txt')
dok=l,nz+1
write(43,1020) (saxn(i,11,k),i=1,nx+1)
enddo
open(unit=44, file='sazx15ps.txt')
dok=l,nz+1
write(44,1020) (sazn(i,11,k),i=1,nx+1)
enddo
open(unit=45, file='sayy15ps.txt')
dok=l,nz+1
write(45,1020) (sayn(i,j,k),j=1,ny+1)
enddo
end if
122

open(unit=83,file='say15ps.txt')
do k=2,nz
write(83,1020) sayn(11,11,k)
enddo
end if
if (n.eq.3400) then
C The result at time t=17ps
open(unit=58,file='sazl7ps.txt')
do k=1,nz+1
write(58,1020) sazn(11,11,k)
enddo
end if
if (n.eq.4000) then
C The result at time t=20ps
open(unit=47,file='ctexz20ps.txt')
do k=1,nz+1
write(47,1010) (TEold(i,11,k),i=1,nx+1)
enddo
open(unit=48,file='te20ps.txt')
do k=1,nz+1
write(48,1020) TEold(11,11,k)
enddo
open(unit=49,file='ctlxz20ps.txt')
do k=1,nz+1
write(49,1010) (TLold(i,1,1,k),i=1,nx+1)
enddo
open(unit=50,file='tl20ps.txt')
do k=1,nz+1
write(50,1020) TLold(11,11,k)
enddo
C Lattice temp
open(unit=51, file='uxnxz20ps.txt')
do k=2,nz
write(51,1010) (uxn(i,1,1,k),i=1,nx)
enddo
open(unit=52, file='uznxz20ps.txt')
do k=2,nz
write(52,1010) (uzn(i,1,1,k),i=1,nx)
enddo
open(unit=53, file='uynyz20ps.txt')
do k=2,nz
write(53,1010) (uyn(1,j,1,k),j=1,ny)
enddo
open(unit=54, file='saxxz20ps.txt')
do k=2,nz
write(54,1010) (saxn(i,1,1,k),i=1,nx+1)
enddo
open(unit=55, file='saxxz20ps.txt')
do k=1,nz+1
write(55,1010) (sazn(i,1,1,k),i=1,nx+1)
enddo
open(unit=56, file='sayy20ps.txt')
do k=2,nz
write(56,1010) (sayn(1,j,1,k),j=1,ny+1)
enddo
open(unit=57, file='saz20ps.txt')
do k=1,nz+1
write(57,1020) sazn(11,11,k)
enddo
open(unit=58, file='say20ps.txt')
do k=2,nz
write(58,1020) sayn(11,11,k)
enddo
C Complete the whole period
1 end do
print *, big
open(unit=59, file='Te10(x,y=0).dat')
do k=1,nz+1
write(59,1010) (z(k)*1.0D+6), TEold(11,11,k)
enddo
open(unit=60, file='Tl10(x,y=0).dat')
do k=1,nz+1
write(60,1010) (z(k)*1.0D+6), TLold(11,11,k)
enddo
open(unit=61, file='Tem224.dat')
do n=1,nt
write(61,1020) (t(n)*1.0D+12),((TEm(n)-300.0)/big)
enddo
open(unit=62, file='um224.dat')
do n=1,nt
write(62,1020) (t(n)*1.0D+12),(u3m(n)*1.0D+9)
enddo
open(unit=63, file='sigmazl0(x,y=0).dat')
print *, "zonezsel"
do k=1,nz+1
print *, (z(k)*1.0D+6), (sazn(11,11,k)*1.0D-9)
enddo
C End main program
C Subroutines
C Calculate temperature

subroutine temp(nx,ny,nz,nz2,$
  dx,dy,dz,x,y,z,t,dt,TLo,TLold,TEo,TEold,$
  epxn,epyn,epzn,epxo,epyo,epzo)

  implicit double precision (a-h,l,o-z)
  dimension x(51),y(51),z(221)
  dimension TEo(41,41,101),TEold(41,41,101),$
  TLo(41,41,101),TLold(41,41,101),$
  epxn(41,41,101),epyn(41,41,101),epzn(41,41,101),$
  epxo(41,41,101),epyo(41,41,101),$
  TLo(41,41,101),TLold(41,41,101),$
  dTE(41,41,101),dTL(41,41,101)

  integer iteration,flagE,flagL

C data

C Lame constant
  clemta=199.0d+9
  clemta2=83.3d+9
C Shear modulus
  cmiu=27.0d+9
  cmiu2=115.0d+9
C Thermal expansion coefficient
  alphal=14.2d-6
  alpha2=4.9d-6
C Electron heat capacity
  ce01=2.1d+4
  ce02=5.8d+4
C Lattic heat capacity
  cl=2.5d+6
  cl2=3.3d+6
C Electron - lattic coupling factor
  gl=2.6d+16
  g2=42.0d+16
C Electron thermal conductivity
  cke01=315.0
  cke02=94.0
C Laser fluence
  flu=1000.0
C Laser pulse duration
  tp=0.1d-12
C Optical penetration depth
  delta=15.3d-9
C Surface reflectivity
  sur=0.93
C Spatial profile parameters
  zs=1.0d-6

  iteration=0
  rx=dt/(4.0*dx*dx)
  ry=dt/(4.0*dy*dy)
  rz=dt/(4.0*dz*dz)
  d0=g*dt/(2.0*cl)
deterorr=1.0d-3

C flagE and flagL indicate whether TE and TL are
C keep on iterating as long as flagE or flagL equals to 1

2 do j=2,ny
  do i=2,nx
    clemta=clemta1
    cmiu=cmiu1
    alpha=alpha1
    ce0=ce01
    cl=cl1
    g=g1
    cke0=cke01
    d0=g*dt/(2.0*cl)
    ce=(3.0*clemta+2.0*cmiu)*alpha*300.0/cl
    do k=2,nz2-1

C Heat source

  aa=-z(k)/delta-((x(i)-10.0*dx)*(x(i>10.0*dx)$
    +(yG)-10.0*dy)*(y(j)-10.0*dy))/(zs*zs)$
    -2.77*(t-2.0*tp)*(t-2.0*tp)/(tp*tp)$
  q=0.94*flu*(1.0-sur)*exp(aa)/(tp*delta)$

  a0=ce0*(TEo(ij,k)+TEold(ij,k))/(2.0*300.0)$
  b1=cke0*(TEo(i+1j,k)+TEold(i+1j,k))$
  b2=cke0*(TEold(i-1j,k)+TLold(i-1j,k))$
  b3=cke0*(TEold(i+1j,k)+TLold(i+1j,k))$
  b4=cke0*(TEold(i-1j,k)+TLold(i-1j,k))$
  b5=cke0*(TEold(ij,k+1)+TLold(ij,k+1))$
  b6=cke0*(TEold(ij,k-1)+TLold(ij,k-1))$
  b=cke0*(TEold(ij,k)+TLold(ij,k))$
  c1=cke0*(TEo(i+lj,k)+TLo(i+lj,k))$
  c2=cke0*(TEo(ij,k)+TLo(ij,k))$
  c3=cke0*(TEo(ij+1,k)+TLo(ij+1,k))$
  c4=cke0*(TEo(ij,k)+TLo(ij,k))$
  c5=cke0*(TEo(ij,k+1)+TLo(ij,k+1))$
  c6=cke0*(TEo(ij,k-1)+TLo(ij,k-1))$

  dd=a0+bl+b2+b3+b4+b5+b6+g*dt/(2.0*(1.0+d0))$

C Iteration starts

  TLo(41,41,101),TLold(41,41,101),$
  epxn(41,41,101),epyn(41,41,101),epzn(41,41,101),$
  epxo(41,41,101),epyo(41,41,101),$...
[124] C Boundary Conditions
[124] do k=2,nz
[124] do j=2,ny
[124] TEnew(i,j,1,k)=TEnew(i,j,2,k)
[124] TEnew(i,j,nz+1,k)=TEnew(i,j,nz,k)
[124] TLnew(i,j,1,k)=TLnew(i,j,2,k)
[124] TLnew(i,j,nz+1,k)=TLnew(i,j,nz,k)
[124] end do
[124] end do
[124] C Test for convergence
[124] detmax=0.0
[124] do i=2,nx
[124] do j=2,ny
[124] detmax=max(detmax,abs(TEnew(i,j,k)-TEold(i,j,k))
[124] end do
[124] end do
[124] C

$ +c_3^*(\text{TEo}(i,j+1,k)-\text{TEo}(i,j,k))$
$ -c_4^*(\text{TEo}(i,j,k)-\text{TEo}(i,j-1,k))$
$ +c_5^*(\text{TEo}(i,j,k+1)-\text{TEo}(i,j,k))$
$ -c_6^*(\text{TEo}(i,j,k)-\text{TEo}(i,j,k-1))$
$ +q*dt)/dd$

$ TL_{\text{new}}(i,j,k)=d_0^*\text{TE}_{\text{new}}(i,j,k)/(1.0+d_0)$
$ +d_0^*(\text{TEo}(i,j,k)-\text{TLo}(i,j,k))/(1.0+d_0)$
$ -c_0^*(\text{TEo}(i,j,k))$
$ +\text{epxo}(i,j,k)+\text{epyo}(i,j,k)+\text{epzo}(i,j,k))$
$ -\text{epxo}(i,j,k)$
$ +\text{epyo}(i,j,k)+\text{epzo}(i,j,k))$/

$ d_0=g*dt/(2.0*(1.0+d_0))$
$ EE=(3.0*\text{clemta}+2.0*\text{cmiu})*\alpha*300.0/cl$
$ do k=nz2+1,nz$

$ +q*dd$

$ TL_{\text{new}}(i,j,k)=d_0^*\text{TE}_{\text{new}}(i,j,k)/(1.0+d_0)$
$ +d_0^*(\text{TEo}(i,j,k)-\text{TLo}(i,j,k))/(1.0+d_0)$
$ +\text{TLo}(i,j,k)/(1.0+d_0)$
$ -\text{ee}/(1.0+d_0)$
$ +\text{epxo}(i,j,k)$
$ +\text{epxo}(i,j,k)$

$ =c_{01}^*\text{TE}_{\text{new}}(i,j,k+1)$
$ =cke_{01}^*\text{TE}_{\text{new}}(i,j,k-1)$
$ /(cke_{01}^++cke_{02}^*)$
$ =cke_{02}^*\text{TL}_{\text{new}}(i,j,k+1)$
$ =cke_{01}^*\text{TL}_{\text{new}}(i,j,k-1)$
$ /(cke_{01}^++cke_{02}^*)$

$ q=0.94*\text{flu}*(1.0+*\text{exp}(aa)/(tp*\text{delta})$
$ ae=ae0^*\text{TE}_{\text{old}}(i,j,k)/\text{TL}_{\text{old}}(i,j,k)$
$ c_{\text{bl}}=cke_{01}^*\text{TE}_{\text{old}}(i,j,k)$
$ c_{\text{b2}}=cke_{01}^*\text{TE}_{\text{old}}(i,j,k)$
$ c_{\text{b3}}=cke_{01}^*\text{TE}_{\text{old}}(i,j,k)$
$ c_{\text{b4}}=cke_{01}^*\text{TE}_{\text{old}}(i,j,k)$
$ c_{\text{b5}}=cke_{01}^*\text{TE}_{\text{old}}(i,j,k)$
$ c_{\text{b6}}=cke_{01}^*\text{TE}_{\text{old}}(i,j,k)$
$ c_{\text{bl}}=cke_{01}^*\text{TL}_{\text{old}}(i,j,k)$
$ c_{\text{b2}}=cke_{01}^*\text{TL}_{\text{old}}(i,j,k)$
$ c_{\text{b3}}=cke_{01}^*\text{TL}_{\text{old}}(i,j,k)$
$ c_{\text{b4}}=cke_{01}^*\text{TL}_{\text{old}}(i,j,k)$
$ c_{\text{b5}}=cke_{01}^*\text{TL}_{\text{old}}(i,j,k)$
$ c_{\text{b6}}=cke_{01}^*\text{TL}_{\text{old}}(i,j,k)$
$ c_{\text{bl}}=cke_{01}^*\text{TL}_{\text{old}}(i,j,k)$
$ c_{\text{b2}}=cke_{01}^*\text{TL}_{\text{old}}(i,j,k)$
$ c_{\text{b3}}=cke_{01}^*\text{TL}_{\text{old}}(i,j,k)$
$ c_{\text{b4}}=cke_{01}^*\text{TL}_{\text{old}}(i,j,k)$
$ c_{\text{b5}}=cke_{01}^*\text{TL}_{\text{old}}(i,j,k)$
$ c_{\text{b6}}=cke_{01}^*\text{TL}_{\text{old}}(i,j,k)$
$ c_{\text{bl}}=cke_{01}^*\text{TL}_{\text{old}}(i,j,k)$
$ c_{\text{b2}}=cke_{01}^*\text{TL}_{\text{old}}(i,j,k)$
$ c_{\text{b3}}=cke_{01}^*\text{TL}_{\text{old}}(i,j,k)$
$ c_{\text{b4}}=cke_{01}^*\text{TL}_{\text{old}}(i,j,k)$
$ c_{\text{b5}}=cke_{01}^*\text{TL}_{\text{old}}(i,j,k)$
$ c_{\text{b6}}=cke_{01}^*\text{TL}_{\text{old}}(i,j,k)$
$ c_{\text{bl}}=cke_{01}^*\text{TL}_{\text{old}}(i,j,k)$
$ c_{\text{b2}}=cke_{01}^*\text{TL}_{\text{old}}(i,j,k)$
$ c_{\text{b3}}=cke_{01}^*\text{TL}_{\text{old}}(i,j,k)$
$ c_{\text{b4}}=cke_{01}^*\text{TL}_{\text{old}}(i,j,k)$
$ c_{\text{b5}}=cke_{01}^*\text{TL}_{\text{old}}(i,j,k)$
$ c_{\text{b6}}=cke_{01}^*\text{TL}_{\text{old}}(i,j,k)$

$ dd=a_0+b_1+b_2+b_3+b_4+b_5+b_6+g*dt/(2.0*(1.0+d_0))$
$ \text{TE}_{\text{new}}(i,j,k)=(b_1*\text{TE}_{\text{old}}(i+1,j,k)+b_2*\text{TE}_{\text{old}}(i-1,j,k)$
$ +b_3*\text{TE}_{\text{old}}(i,j+1,k)+b_4*\text{TE}_{\text{old}}(i,j-1,k)$
$ +b_5*\text{TE}_{\text{old}}(i,j,k+1)+b_6*\text{TE}_{\text{old}}(i,j,k-1)$
$ -g*dt*(\text{TEo}(i,j,k)-\text{TLo}(i,j,k))/(2.0*(1.0+d_0))$

$ +c_{1}^*\text{TE}_{\text{old}}(i+1,j,k)-\text{TE}_{\text{old}}(i,j,k))$
$ -c_{2}^*\text{TE}_{\text{old}}(i-1,j,k)$
$ +c_{3}^*\text{TE}_{\text{old}}(i,j+k+1)-\text{TE}_{\text{old}}(i,j,k))$
$ -c_{4}^*\text{TE}_{\text{old}}(i,j,k-1)$
$ +c_{5}^*\text{TE}_{\text{old}}(i,j,k+1)-\text{TE}_{\text{old}}(i,j,k+1))$
$ +c_{6}^*\text{TE}_{\text{old}}(i,j,k-1)-\text{TE}_{\text{old}}(i,j,k-1))$
detl=abs(TEnew(ij,k)-TEold(ij,k))
if (detl.gt.detmax) detmax=detl

det2=abs(TLnew(ij,k)-TLold(ij,k))
if (det2.gt.detmax) detmax=det2
enddo
enddo
enddo

Density
loul=1.93d+4
lou2=7190.0

Electron - blast coefficient
tri 1=70.0
tri2=193.3

if (detmax.le.deterror) goto 3
doi=l,nx+l

doj=l,ny+l

dok=l,nz+l

TEold(ij,k)=TEnew(ij,k)
TLold(ij,k)=TLnew(ij,k)
enddo
enddo
enddo

iteration=iteration+1
goto 2

C Update all the TEold, TLold with TEnew and TLnew

3 doj=2,ny

doi=l,nx+l

dok=l,nz+l

vxn(ij,k)=(difx(ij,k)
$+difxxy(ij,k)+difxzzi(ij,k))
$+tri*theta*(TEold(i+1,j,k))
$*TEold(i+1,j,k)*TEold(i,j,k)/dx)
$+tri*(1.0-theta)*TEo(i+1,j,k))
$*/dx)*dt/lou+vxo(ij,k)

uxn(ij,k)=(theta*vxn(ij,k)
$+(1.0-theta)*vxo(ij,k))*dt+uxo(ij,k)
end do

lo=lo2
tri=tri2

do k=nz2+1,nz
vxn(ij,k)=(difx(ij,k)
uxn(ij,k)=(theta*vxn(ij,k)
$+(1.0-theta)*vxo(ij,k))*dt+uxo(ij,k)
end do

vyn(ij,k)=(vyn(ij,k+1)+vyn(ij,k-1))/2
uxn(ij,k)=(uxn(ij,k+1)+uxn(ij,k-1))/2

uxn(ij,k)=(theta*vyn(ij,k)
$+(1.0-theta)*vyn(ij,k))*dt+uxo(ij,k)
end do

implicit double precision (a-h,l-o-z)

dimension TEOd(41,41,101),TEold(41,41,101),
$ saxo(41,41,101),sayo(41,41,101),sazo(41,41,101),
$ saxly(41,41,101),sayx(41,41,101),sayz(41,41,101),
$ saxyn(41,41,101),saxzn(41,41,101),sayzn(41,41,101),
$ vxo(41,41,101),vyn(41,41,101),vzn(41,41,101),
$ uxo(41,41,101),uyo(41,41,101),uzo(41,41,101),
$ difx,difxxy,difxzzi,difyx,difyzzi,difxz,difyzy)

C End of subroutine temp()
$\begin{align*}
&+\theta\cdot v_{\text{yo}}(i,j,k)\cdot dt+\text{uyo}(i,j,k) \\
\text{end do}
\end{align*}$

lou=lou2
tri=tri2
do k=nz2+1,nz
$v_{\text{yn}}(i,j,k)=(\text{dfy}x(i,j,k)
$ +\text{dfy}y(i,j,k)+\text{dfy}z(i,j,k)
$ +\text{tri}\cdot\theta\cdot (\text{TE}_{\text{old}}(i,j,k+1,k)-\text{TE}_{\text{old}}(i,j,k)))/(\text{dy})$
$ +\text{tri}\cdot(1.0-\theta)\cdot (\text{TE}_{\text{o}}(i,j,k+1,k)-\text{TE}_{\text{o}}(i,j,k))
$ /\text{dy})\cdot dt/lou+\text{vyo}(i,j,k)

$\begin{align*}
&+\theta\cdot v_{\text{yn}}(i,j,k) \\
\text{end do}
\end{align*}$

k=nz2
$v_{\text{yn}}(i,j,k)=(v_{\text{yn}}(i,j,k+1)+v_{\text{yn}}(i,j,k-1))/2
u_{\text{yn}}(i,j,k)=(u_{\text{yn}}(i,j,k+1)+u_{\text{yn}}(i,j,k-1))/2
\text{end do}
\text{end do}$
do i=2,nx
do j=2,ny
lou=lou1
tri=tri1
do k=1,nz2-1

$v_{\text{zn}}(i,j,k)=(\text{difx}z(i,j,k)
$ +\text{dify}z(i,j,k)+\text{difz}(i,j,k)
$ +\text{tri}\cdot\theta\cdot (\text{TE}_{\text{old}}(i,j,k+1)-\text{TE}_{\text{old}}(i,j,k))$
$ +\text{tri}\cdot(1.0-\theta)\cdot (\text{TE}_{\text{o}}(i,j,k+1)-\text{TE}_{\text{o}}(i,j,k))
$ /\text{dz})\cdot dt/lou+v_{\text{zo}}(i,j,k)

$\begin{align*}
&+\theta\cdot v_{\text{zn}}(i,j,k) \\
\text{end do}
\end{align*}$

lou=lou2
tri=tri2
do k=nz2,nz
$v_{\text{zn}}(i,j,k)=(\text{difx}z(i,j,k)
$ +\text{dify}z(i,j,k)+\text{difz}(i,j,k)
$ +\text{tri}\cdot\theta\cdot (\text{TE}_{\text{old}}(i,j,k+1)-\text{TE}_{\text{old}}(i,j,k))$
$ +\text{tri}\cdot(1.0-\theta)\cdot (\text{TE}_{\text{o}}(i,j,k+1)-\text{TE}_{\text{o}}(i,j,k))
$ /\text{dz})\cdot dt/lou+v_{\text{zo}}(i,j,k)

$\begin{align*}
&+\theta\cdot v_{\text{zn}}(i,j,k) \\
\text{end do}
\end{align*}$

return
end
REFERENCES


